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TWO-PHASE FLOW PRESSURE DROP IN A TEE JUNCTION

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Ву

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تعتمد كلية الدراسات العليا هذه النسخة من الرسالية التوقيع السائل من الرسالية

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DEDICATION

To the loving memory of my grandfather, who I shall always remember. To my father and mother who gave me love and support. To my wife, daughter, sister and brothers who gave me hope and love.

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Nomenclature

cross sectional area; (m²)
diameter; (mm)
friction factor
mass flux; (kg/s.m²)
head loss; (mm Hg)
superficial liquid velocity; (m/s)
superficial gas velocity; (m/s)
single-phase loss coefficient
length; (m)
mass flow rate; (kg/s)
average pressure; (Pa)
pressure difference; (Pa)
system
velocity ratio
temperature at the inlet of the T-junction; (°C)
average velocity; (m/s)
volumetric flow rate; (ℓ/hr)
mass flow rate; (kg/s)
quality
mass flow ratio
ools
void fraction
two-phase flow loss multiplier
density; (kg/m³)
surface tension; (N/m)

subscripts

a	air
d	dynamic
E	energy
G	gas
Н	homogeneous
i	test number
ℓ	liquid
M	momentum
P	pipe
s	system
t	tap
w	water
1	inlet test leg
2	branch two
3	branch three

ABSTRACT

"Two-phase flow pressure drop in a tee-junction"

By

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In this work an experimental test loop system was designed, constructed and investigated using horizontal inlet and branches orientation, equal sided inner diameter (25.4 mm), sharp edge T-junction using PVC pipes at a system pressure of 1 bar.

The experimental set up includes: centrifugal water pump, air compressor, pipe connection, mixer, air-water separator, valves and T-junction, pressure gauge and taps and pressure holder.

Experiments were performed in fluid mechanics laboratory of the engineering faculty of Mu'tah University, by the aid of an instrumentation system consisting of pressure bourdon gauge, Rotometers, Thermocouples and digital thermometer, pitot tube and U-manometer. Experimental observations of pressure differences at various positions for single-phase flow and two-phase flow, and measurements of air and water mass flow rates and temperatures are presented.

Single-phase loss coefficient, single phase and two-phase flow pressure difference values were determined.

The results show that the two-phase flow pressure difference depends on the inlet mass flux, inlet quality, and extraction rate. Pressure difference increases with increasing these parameters. Comparisons show good agreement, with some existing models.

Several conclusions and recommendations were drawn to enhance research in two-phase flow field.

1. INTRODUCTION

1.1 Introduction

Phase is simply one of the states of matter and can be either a gas, a liquid or a solid. Two-phase flow is the simultaneous flow of two phases within the piping system and it obeys all the basic laws of fluid mechanics. The equations associated with two-phase flow are merely more complicated than those of single-phase flows, the major complications in comparing two-phase flow with single-phase flow is the variety of flow patterns that can be produced in a gas-liquid system.

Branching junctions are common features of the piping networks used for single-phase flow and two-phase flow distribution systems. These networks are essential components for many facilities in the power and process industries, such as boilers, condensers, refrigerators, air-lift pumps, natural gas pipelines and nuclear reactors. Knowledge of the phase separation and pressure drop in two-phase flow systems are of primary important to the system designer to select pumps minimize pressure losses in the system piping, since they have a significant effects on the operation and efficiency of all components downstream of the junction and it leads to safe the labors and industrial fluid dynamic systems.

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1.2 Literature Review

The phenomenon of two-phase flow is important in the fields of mechanical, chemical and aerospace engineering. Owing to the inherent complexities of the physical process of two-phase flow only simplified analytical and experimental studies have had some success. Most investigations, among that available in the literature, have been experimented in nature because of the limitations of existing theoretical models.

Seeger *et al.* (1986) and Reimann *et al.* (1986) investigated the phase separation and pressure drop of two-phase flow in a T-junction with horizontal inlet, equal pipe diameters (d=50 mm) for air-water and steamwater flow at a horizontal, vertical upward or downward branch orientations. Their experiments were performed with air-water flow (maximum pressure $p_1 < 1$ MPa), steam-water flow ($p_{max} < 10$ MPa) and an inlet mass flux range $500 < G_1 < 7000 \text{ kg/m}^2$.s.

From plotting the ratio of branch to inlet qualities (x_3/x_1) versus their mass flux ratio (G_3/G_1) , they found that their results are very close to the total separation curve for upward branch orientation. In horizontal orientation, all curves are quite close to the total separation curve for high values of G_3/G_1 as shown in figure (1.1). A maximum of vicinity at $G_3/G_1 = 0.3$ and then decreases quite rapidly with decreasing G_3/G_1 . Downward branch results showed that there is a range where only liquid enters the branch and this range increases the more stratified the flow pattern is. At constant superficial gas velocity, they found that the branch to inlet quality (x_3/x_1) increases with increasing the superficial liquid velocity. Their results were compared with other works, showed good agreement with Saba & Lahey (1982), Honan & Lahey (1981), Zotzmann (1982) and McCreery (1984) at high mass flux ratios

and lacking in the lower parameter range, at which Azzopardi & Whalley (1982) and Azzopardi and Backer (1981) proposed a model based on void fraction and phase velocity distribution in the cross section together with local momentum flux that agree well at these low values.

An improved model for two-phase flow pressure difference in a T-junction was proposed. Comparing their results with other known models such as Saba & Lahey slip and homogeneous models (1984), Chisholm model (1967) and the homogeneous model, shows good agreement with superior results in airwater data for horizontal and downward branches, but fail for the case of upward branch due to the separation effects on the branch which increases with decreasing mass flux.

They suggested further investigations to include global or local void fraction measurements, in order to obtain suitable model for upward branch. They also influenced clarifying the dependency of the results on the system pressure through increasing number of experiments at each pressure level.

Ballyk *et al.* (1988) investigated separation phenomena and pressure drop of a steam-water annular flow in equal leg flow areas, horizontal dividing T-junction covering a range of operating conditions of $400 < G_1 < 1200 \text{ kg/m}^2$.s, $0 < x_1 < 0.15$ and $0.0 < m_3/m_1 < 1$. They measured pressure and void fraction distributions as well as the total flow rates and quality along the inlet and branching legs in order to determine the phase separation and pressure characteristics. They also improved the mechanical energy based model for calculating the branch pressure drop to be able to account for the inlet dynamic head of the branching flow. They found that at low branch flow splits (m_3 / m_1), the branch quality (x_3) increases very rapidly with

increasing flow splits crossing the equal phase distribution line at close to 90°. This supports the finding of Azzopardi & Whalley (1982) and Henry (1981) and suggests that the liquid is first removed from the low velocity film flowing at the tube wall in the line with the branching port. The effect of inlet mass flux variation on the branch phase separation ratio is found to be less significant than that of inlet quality, where the point of total separation is some what independent of mass flux. And there is a strong interdependence between separation phenomenon and pressure distribution, where it was clear that as the total phase separation was approached a significant increase in the run pressure recovery was encountered. This was referred to flow regime transition from annular to plug flow in the run of the T-junction. Their axial pressure recovery data obtained in their work demonstrated similar trends as Shoukri *et al.* (1987) and poor agreement with Riemann & Seeger (1986) particularly for low split ratios. Based on their results they recommended their model for evaluating the branch pressure changes.

Rubel *et al.* (1988) investigates phase distribution of low pressure steamwater mixture in a horizontal T-junction with equal sided (37.6 mm) inner diameters, the purpose of there investigation is to develop new phase distribution data under certain operating conditions and enhancing the current understanding of the phenomenon. The range of conditions is of low inlet mass flux $15 \ G \ 50 \ kg \ m^2$.s and low pressure of $100 \ p \ 250 \ kPa$, wide range of inlet qualities ($20 \ x_1 \ 90 \ \%$) were used and extraction rates of $0.15 \ w_3 \ w_1 \ 0.8$. The operating conditions corresponding to their test groups were plotted on the flow pattern map of Mandhane *et al.* (1974) from which their visual observations of inlet flow patterns are in good agreement with Mandhane predictions. Rubel and his colleagues found that as a general rule, there is uneven phase distribution at the junction where preferential

liquid flow into the branch occurred for all the data of stratified flow, while most of wavy flow data showed preferential vapor flow into the branch. They also studied the individual effects of important independent parameters such as inlet quality, inlet mass flux and the inlet superficial liquid velocity on the phase distribution. They found that at the same high mass flux and different inlet qualities the ratio of branch to inlet qualities (x_3/x_1) decreases with x_1 increasing, while an opposite trend were obtained at the lowest mass flux. At the same inlet quality but different inlet mass flux (G_1) , G_1 doesn't appear to influence (x_3/x_1) within the same inlet flow pattern but the main influence is related to the inlet flow pattern transitions.

At fixed values of inlet superficial gas velocities (J_{Gs}) the effect of inlet superficial liquid velocity (J_{Ls}) indicates that the ratio of the branch to inlet liquid extraction rates (W_{L3}/W_{L1}) decreases continuously with increasing (J_{Ls}), and shows an important effect of the velocity ratio on phase distribution irrespective of flow pattern.

Comparisons with other models show a remarkable agreement with Seeger et al. (1986) for the data corresponding to $G_3/G_{1\geq}$ 0.4 and $x_{1\geq}40\%$ and encouraging results for other ranges. Azzopardi & whalley (1982) model is based on annular flow shows that the data is generally out side $\pm 20\%$ prediction accuracy and it becomes increasingly better as x_1 and J_{G_8}/J_{L_8} increase, which moves the flow to fully-established annular flow. Hwang et al.(1988) developed his model based on a dividing streamline concept shows that the predicted values of x_3/x_1 are within $\pm 30\%$ of the corresponding measured values for all points.

Buell et al. (1994) investigated two-phase pressure drop and phase distribution at a horizontal T-junction of low pressure (1.5 bar) air-water

mixture and equal sided (37.6mm) inner diameters. Inlet flow regimes of stratified, wavy, slug and annular with inlet liquid velocities J_{L1}< 0.2 m/s at which limited pressure drop information is available. Data consist of both phase distribution and pressure drop at different extraction rates and a wide range of inlet condition is obtained. The effect of superficial liquid velocity JLI on the phase distribution for annular flow was found that the preference of gas to exit through the branch increases as J_{L1} increases at a fixed super facial gas velocity J_{GI}. This explained using the ideas suggested by Hwang et al. (1988) and Azzopardi et al. (1988), whereby an increase in JLI results in an increase in the average axial momentum of the liquid phase as consequently, a decrease in the probability of the liquid to enter the branch. But the effect of varying J_{G1} at fixed J_{L1} failed to produce consistent trends due to flow regime transition encountered with changing J_{GI} . Good agreement for the phase distribution of $m_3/m_1 \ge 0.3$ has been demonstrated against the analytical models of Hwang et al. (1988), empirical correlation of Seeger et al. (1986) and Azzopardi & Whally (1982).

Comparisons of pressure differences ΔP_{12} (inlet-run) and $\Delta P_{1.3}$ (inlet-branch) of their data with different empirical models, developed for significantly higher values of J_{L1} , showed that best prediction of ΔP_{12} data was produced by a separated flow model, originally proposed by Fouda & Rhodes (1974) and less satisfactory predictions were obtained for ΔP_{13} . For high extraction rates $m_1/m_1 \geq 0.3$ separated flow model recommended by Saba & Lahey (1984) gave the best prediction while for $m_1/m_1 < 0.3$ the model recommended by Reimann & Seeger (1986) gave better predictions. They expected that obtained results could guide the future search for generalized methods of correlation the pressure results. The performance of the above

models in predicting the pressure drop in slug flow regime is summarized and given in table (1.1)

Table (1.1) Summary of model performance for branch pressure drop

Model	Interval	%of data predicted correctly		
Model	mervar	Slug flow regime(inlet)		
Homoganaous Flour	± 50%	0		
Homogeneous Flow Model (HFM)	± 30%	0		
Wiodei (FirWi)	± 20%	0		
Sanarated Flow Model	± 50%	62		
Separated Flow Model	± 30%	31		
(SFM)	± 20%	8		
Daimann and Casaar	± 50%	62		
Reimann and Seeger	± 30%	39		
Model (RSM)	± 20%	31		
D 11 1 M 1 1	± 50%	39		
Ballyk Model	±30% 🛘	15		
(BM)	± 20%	8		
11	± 50%	0		
Hwang and Lahey Madal (III M)	± 30%	0		
Model (HLM)	± 20%	0		

Schmidt and Loth (1994) discuss several predictive methods for two-phase flow pressure loss calculation in T-junctions with combining conduits. Based on known methods, three predictive methods for pressure loss were developed; these are "loss coefficient model", "contraction coefficient model" and "momentum coefficient model". The flow loop is shown in

annular for both test sections plus slug for the larger branch test section. The objective of their investigation is to generate data concerning the effects of branch diameter using two sets of data where the branch-to-inlet diameter ratios (D_3/D_1) were 0.206 and 0.5.

They found that data of stratified flow regimes for D_3/D_1 =0.5 shows greater tendency for the gas to enter the branch than for D_3/D_1 =1.0 as in Buell *et al*. (1994). That it is reasonable to expect that smaller fraction of liquid could be taken off into the branch as the branch diameter decreases. But the trends for D_3/D_1 =0.206 is not as consistent, because at high extraction rates the tendency for liquid preferentially to enter the branch begins to increase and the phase distribution data approach the case of $(D_3/D_1$ =1.0), this related to the fact that the branch of D_3/D_1 =0.206 is much smaller than that of D_3/D_1 =0.5. Therefore, for similar values of fraction of gas entering the branch F_{BG} , the branch gas superficial velocity of D_3/D_1 =0.206 is considerably larger than that for D_3/D_1 =0.5. Due to Bernoulli effect, it is possible that liquid entrainment into the branch is responsible for this phenomenon. These trends are consistent in the stratified wavy, wavy, semi annular and annular flow regimes.

Comparison with other phase distribution models showed that the geometrically based model by Azzopardi (1988), using the correlation of Kataoka and Ishii(1983) for entrainment, gave the best results for both diameter ratio for annular flow, and the deviation between data and any of the models is highest at two extraction rates, and this deviation decreases as the extraction rate increases. The effect of D_3/D_1 on ΔP_{12} (inlet-run) pressure drop is minimal while ΔP_{13} (inlet-branch) pressure drop increase significantly with a decrease in D_3/D_1 , these results are consistent with the single-phase flow.

Their results were compared against homogeneous flow model by Saba & Lahey (1984), the separated flow model by Fouda & Rhodes (1974), the model by Riemann & Seeger (1986) and the model by Hwang and Lahey (1988). The performance of the various ΔP_{12} , ΔP_{13} , models showed that the separated flow model (SFM) has the best over all predictive capabilities and the lowest deviations for both diameter ratios. However, in the semi-annular and annular flow regimes the models by Reimann & Seeger (1986), Hwang & Lahey (1988) and the homogeneous flow model all perform better than the separated flow model.

Despite the above studies related to two-phase flow discussed in the literature survey, which intensively covered phase distribution in a T-junction at different flow regimes, test leg diameters, inlet qualities and extraction rates, a lack in pressure differences studies were observed. Only limited pressure drop data for slug flow regime is available. All models presented were failed in obtaining a model describes this pressure difference through a T-junction generally. They referred this to the lack of pressure drop data and the number of parameters affect their values.

1.3 Objective of the present work

This study aimed at the following objectives:

- 1. to build a two-phase flow facility that participates in the research in the two-phase flow field.
- 2. to study the two-phase flow pressure differences through a T-junction of air and water mixture in slug flow regime with equal sided diameters, at different inlet qualities and extraction rates.

3. to provide new pressure differences experimental data to the researchers, helping them in their efforts to obtain a general model describes the two-phase flow pressure differences through the T-junction.

1.4 Out line of the Report

Following this introduction, a theory is presented in chapte two. Chapter three is a description of experimental setup and procedure. The results and discussion of the pressure difference in a T-junction are presented in chapter four.

Finally conclusions and recommendations are summarized in chapter five.

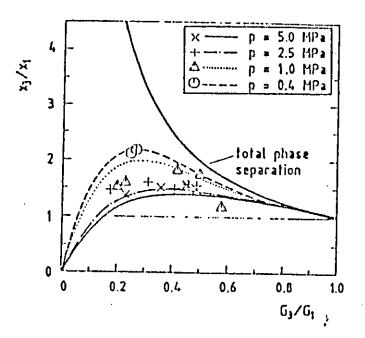


Figure (1.1) Phase separation for horizontal branch (Seeger, 1986).

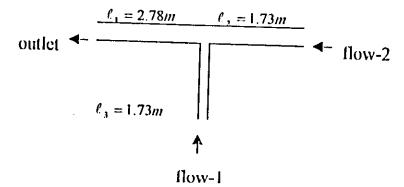


Figure (1.2) T-junction with combining conduits.(Schmidt, 1994).

2. THEORY

2.1 Introduction

A phase is simply one of the states of matter and can be either a gas, a liquid or a solid. Multiphase flow is the simultaneous flow of several phases.

The term two component is sometimes used to describe flows in which the phases do not consist of the same chemical substance. For example, steamwater flows are two-phase, while air-water are two component. Since the mathematics which describe two-phase or two-component flows are identical, it doesn't really matter which definitions are chosen. The two expressions will therefore be treated as synonyms. (Wallis, 1969)

Two-phase flow obey all the basic laws of fluid mechanics, the equations associated with two-phase flow are merely more complicated than those of single-phase flow. The major complications in comparing two-phase flow to single-phase flow is the variety of flow patterns, that can be produced in a gas-liquid system. In an attempt to simplify these complexities researchers have conducted empirical testing in an effort to develop correlations that are easy for the average design engineer to utilize in solving practical problems. Other than correlation method, researchers have developed additional techniques to predict flow patterns and quantify pressure drops.

2.2 Flow Regimes

There are various flow regimes or "patterns" that are common in any twophase flow system, each having different characteristics and associated pressure drops. The flow patterns also vary between horizontal and vertical piping runs.

In horizontal pipe, the flow regimes are (in order of lowest gas velocities to highest) bubble flow, plug flow, stratified flow, wavy flow, slug flow, annular flow and mist flow. In vertical pipe there are bubble, slug, churn, ripple, annular and mist flow.

2.2.1 Horizontal flow regimes

For two-phase flow in horizontal pipes, the flow patterns shown in figure (2.1) classified as bubble, plug, stratified, wavy, slug, annular and mist flow these patterns can be clarified as follow:

Bubble flow: - This flow is prevalent at high ratios of liquid to gas flow rates. The gas is dispersed as bubbles that flow along the upper surface of the pipe at approximately the same velocity as the liquid. At high liquid rates, bubbles may be dispersed throughout the liquid, a pattern referred to as froth flow.

Plug flow: - Plug flow describes a pattern in which alternates plugs of gas and liquid move along the upper part of the pipe. This pattern develops when the gas bubbles coalesce due to gas flow rate increases.

Stratified flow: - This flow pattern is very common in typical multi-phase systems. In stratified flow, the liquid flows along the bottom of the pipe and

the gas flows at the top of the pipe over a smooth liquid/gas interface. This pattern occurs at lower liquid flow rates than bubble or plug flow.

wave flow: - This flow pattern is also very common in typical multi-phase systems. Similar to stratified flow, the liquid flows along the bottom of the pipe. However, increased gas velocities above the liquid produce waves on the liquid/gas interface. Increasing gas flow rates produce waves of increasing amplitude.

Slug flow: - This flow pattern is also common in typical multi-phase systems. When the wavy flow wave crests become sufficiently high to bridge the pipe, they form frothy slugs which move much faster than the average liquid velocity. Slug flow is also formed from plug flow as the gas flow rate is increased at a constant liquid flow rate. Slug flow can cause severe and sometimes dangerous vibrations in piping and equipment because of the impact of the high velocity slugs against bends and other fittings.

Annular flow: - In annular flow, the liquid flows as a thin film along the pipe wall and gas flows in the core. The gas moves at high velocity in the core and carries with it some of the liquid as a spray.

Mist flow: - This flow pattern occurs at very high gas velocities at which nearly all of the liquid in the pipe is entrained as small droplets. More and more liquid is carried in the gaseous core at the expense of the annular film, until nearly all of the liquid is entrained in the gas. (Peramaki, m. p., 2001)

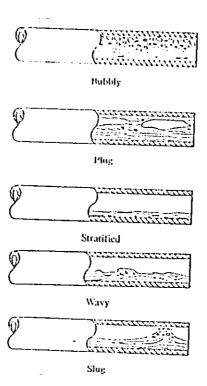


Figure (2.1) Flow patterns for horizontal flow of a liquid and a gas.(Mills, 1995)

2.2.2 Vertical flow regimes

Visual definitions of flow patterns in vertical flow appear to cause more difficulty than do those in horizontal flow. Figure (2.2) illustrates the vertical flow patterns and they can be described as follow:

Bubble flow: - In bubble flow pattern the gas phase is nearly uniformly distributed in the form of bubbles in a continuous liquid phase with some tendency to concentrate toward the center of the pipe. As the gas flow rate increases the bubbles steadily increase in size and numbers. This flow pattern has a high relative pressure drop due to the hydrostatic losses resulting from very low air to water ratios.

Slug flow: - In this pattern gas bubbles coalesce to form larger bullet shaped bubbles. Most of the gas is contained in large bullet shaped bubbles which

have a diameter nearly equal to the pipe diameter. These bubbles are commonly called "Taylor Bubbles". Taylor bubbles move uniformly upward and are separated by slugs of liquid which bridge the pipe and contain small gas bubbles. Between the Taylor bubbles and the pipe wall, liquid flows downward in the form of a thin falling film. Taylor bubbles increase in length and diameter and their upward velocity increases with an increasing gas flow rate.

Churn flow: - Churn flow is similar to slug flow, although much more chaotic and disorderly. The Taylor bubbles narrow and become distorted. The continuity of the liquid in the slug between bubbles is repeatedly destroyed by a high local gas concentration within the slug. As the liquid slug is destroyed, it falls. This liquid accumulates, forms a bridge and is again lifted by the gas. The elements of this structure are in a continual process of collapse and reformation. This activity results in a pattern-less turbulent mixture. Churn flow is characterized by strong intermittent and intense mixing, with neither phase easily described as continuous or dispersed.

Churn flow at low air to water ratios has the highest relative pressure drops common to multi-phase systems. The interaction between the gas and liquid phases increases the pressure drop compared to other flow patterns. For this reason this flow pattern should be avoided. Additionally, the slugging produces strong vibrations in piping.

Ripple flow: - Ripple flow is characterized by an upwards-moving, wavy or ripply layer of liquid on the piping wall, far less chaotic that of churn flow. The zone of ripple flow is narrow and actually forms a transition region between churn and annular flow. The pressure drop associated with this

transition is a relative low and if possible, designers should attempt to produce this flow pattern in vertical pipe runs and drop tubes.

Annular (film) flow: - The annular flow pattern consists of an annular liquid layer moving on the pipe wall with a gas core moving at much higher velocities. This flow pattern results from the smoothing out of the waves and ripples of Ripple flow by the attrition of higher gas flow rates. Some liquid is entrained as droplets within the gas core. Gravitational (hydrostatic) forces become less important than interphase forces, and the liquid is mainly carried as a thin film a long the pipe wall.

Mist flow: - Annular flow becomes mist flow when the gas flow rate becomes high enough to capture the water droplets from the gas fiquid interface, removing the annular liquid layer. Mist flow occurs when all the liquid is carried as fine drops within the gas phase. (Peramaki, m. p., 2001)

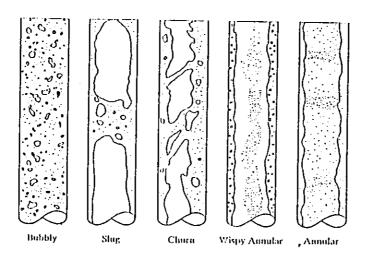


Figure (2.2) Flow patterns for vertical flow of a liquid and a gas. (Mills, 1995).

2.2.3 Flow Patterns Maps

Numerous authors have presented flow-regime maps in which various areas are indicated on the graph for which there are two independent coordinates.

The two-phase flow pattern map such as Mandhane shown in figure (2.3) serve a very useful purpose in allowing designers to predict the nature of the flow pattern that might be expected in a given case. This can assist in selecting the most suitable correlation for calculating the pressure drops. Otherwise, this information can be used to avoid slug flow in the horizontal piping and early churn flow in the vertical piping.

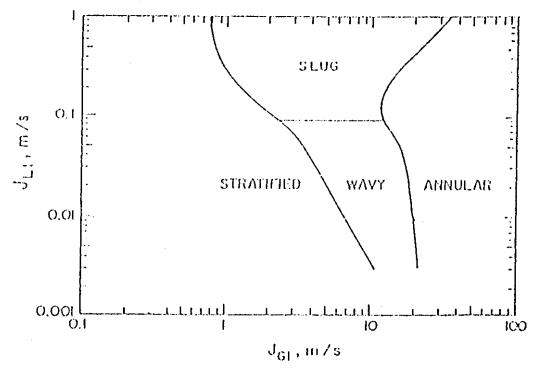


Figure (2.3) flow regime map showing inlet conditions. (Mandhare, 1974)

2.3 Method Of Analysis

Two phase flow obey all the basic laws of fluid mechanics, the equations are merely more complicated or more numerous than those of single-phase flow.

Several characteristics make them more complicated, such as: the difference in the fluid characteristics (especially the phase density difference) create flow patterns that can be dramatically different when only the pipe orientation changes. In addition to the fact that flow pattern description is not merely an identification of laminar, transitional or turbulent flow.

2.3.1 Simple Definitions

The two phase are distinguished by subscripts 1 and 2 in general, or by ℓ and g for a liquid-gas system or ℓ and s for a liquid-solid system.

The total mass flow rate is represented by the symbol W or m. The total flow is the sum of the component flows:

$$m_{ph} = m_p + m \tag{2.1}$$

the ratio of gas flow to total flow x,

$$\chi = m_g/m_{2ph} \tag{2.2}$$

is sometimes called the dryness fraction or the quality.

For a channel with cross sectional area A, the mass flux or mass velocity G is defined as,

$$G=m_{2ph}/A_{2ph} \tag{2.3}$$

The void fraction α is defined as the ratio of the gas-flow cross-sectional area A_g to the total cross sectional area,

$$\alpha = A_g / A_{2ph} \tag{2.4}$$

where A_{2ph} must equal the sum of the cross sectional areas occupied by the two phases :

$$\Lambda_{2ph} = A_g + A_{\ell} \tag{2.5}$$

It follows that the liquid volume fraction α_{ℓ} is given by,

$$\alpha_{\ell} = 1 - \alpha = A_{\ell} / A_{2ph}$$
 (2.6)

It is also useful to define superficial gas and liquid fluxes, J_{ℓ} and J_{G} respectively as,

$$J_{G}=Gx/\rho_{g} \tag{2.7}$$

$$J_{\ell} = G(1-x)/\rho_{\ell} \tag{2.8}$$

Although these parameters have units of velocity, they can also be thought of as the volume flux of each phase through the channel. Numerically they are equal to the velocity that each phase would have if it flowed at its specified mass flow rate through the channel alone.

Simple analytical models can be quite successful, both for organizing experimental results and for predicting design parameters. For example, homogeneous flow theory provides the simplest technique for analyzing two-phase (or multiphase) flows. Suitable average properties are determined and the mixture is treated as a pseudofluid that obeys the usual equations of single-phase flow. All of standard methods of fluid mechanics can then be applied. But separation flow theory takes the account of the fact that the two phases can have differing properties and different velocities. Separate equations are written for each phase and the interaction between the phases is also considered. Drift flux model is essentially a separated-flow model in which attention is focused on the relative motion rather than on the motion of the individual phases. (Wallis, 1969)

2.3.2 Two Phase Pressure Drop Fundamentals

Two-phase flows obey all the basic laws of fluid mechanics, the equations associated with two-phase flow are merely more complicated or more numerous than those of single-phase flows. In an attempt to simplify these complexities researchers have conducted empirical testing in an effort to

develop correlations that are easy for the average design engineer to utilize in solving practical problems. Other than the correlation method researchers have developed additional techniques to predict flow patterns and quantify pressure drops.

Pressure drop calculations for two-phase flow are more complex than for single-phase flow. The introduction of a second phase causes a sharp rise in pressure drop due to the following:

- 1) the cross-sectional area available for the flow of one phase is reduced by the introduction of the second.
- 2) the flow boundary is no longer a relatively smooth pipe was, but is a rough, irregular, mobile interface between phases.
- 3) the introduction of a second phase can cause a very unsteady flow accompanied by intense turbulence.

Powers, 1992 states that the actual pressure drop in horizontal construction dewatering pipe containing both water and air can be from 150 to 200 percent of the drop due to water flowing in the pipe a lone. (Peramaki, 2001)

2.3.2.1 Single Phase Flow

For single-phase incompressible flow in a T-junction, a correlation for the inlet-to-run pressure difference is normally derived from a momentum approach as follows:

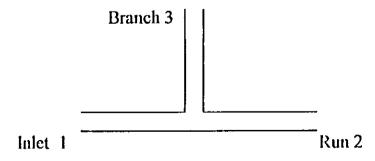


Figure (2.4) Test section shape of the T-junction considered in Theory.

$$\Delta p_{12} = p_1 - p_2 = k_{12} (G_2^2 - G_1^2) / \rho$$
 (2.9)

Where P_1 and P_2 are the average pressures at the junction from the inlet and run sides. G_1 and G_2 are the inlet and run mass fluxes, k_{12} is the momentum correction factor and ρ is the density.

An energy approach has been frequently adopted for the inlet-to-branch pressure difference, subdividing Δp_{13} into a reversible component $(\Delta P_{13})_{rev}$ and an irreversible component $(\Delta P_{13})_{irr}$, thus

$$\Delta P_{13} = P_1 - P_3 = (\Delta P_{13})_{rev} + (\Delta P_{13})_{irr}$$
 (2.10)

where,
$$(\Delta P_{13})_{rev} = (G_3^2 - G_1^2) / 2\rho$$
 (2.10.1)

$$(\Delta P_{13})_{irr} = K_{13}G_1^2 / 2 \rho \tag{2.10.2}$$

 P_3 is the pressure at the junction from the branch side, G_3 is the branch mass flux and K_{13} is a loss coefficient. (Buell, et al., 1994, Reimann and Seeger, 1986, Walters, et al. 1997)

The loss coefficient were correlated by many authors, their results gives different correlations summarized as follow

- Buell et al.(1994) correlation:

$$K_{13}=1-0.982(G_3/G_1)+1.843(G_3/G_1)^2-0.717(G_3/G_1)^3$$
 (2.11)

- Ballyk, et al.(1988) correlation:

$$K_{13} = 1.081 - 0.914 (m_3/m_1) + 1.05 (m_3/m_1)^2$$
 (2.12)

- Reimann and Seeger (1986) correlation:

$$K_{13}=1.036 - 0.9546(G_3/G_1) + 1.2123(G_3/G_1)^2$$
 (2.13)

- Saba and Lahey curve obtained from Buell, et al., 1994.

2.3.2.2 Two-Phase flow

The available correlations for ΔP_{12} and ΔP_{13} during two-phase flow are mostly modifications of the single-phase flow correlations. In general, most of these correlations can be agglomerated into these forms:

$$\Delta P_{12} = K_{12} ((G_2^2/\rho_2) - (G_1^2/\rho_1))$$
 (2.14)

and
$$\Delta P_{13} = \frac{\rho_{II3}}{2} ((G_3^2/\rho_3^2) - (G_1^2/\rho_1^2)) + K_{13} \frac{G_1^2}{2\rho_t} \phi$$
 (2.15)

Where ρ_1, ρ_2 and ρ_3 are mixture densities in the inlet, run and branch sides respectively, ρ_{II3} is the homogeneous density in the branch, ϕ is a two-phase loss multiplier and ρ_t is the liquid density. (Buell et al., 1994, Reimann and Seeger, 1986, Walters et al. 1997)

Different models were computed, the particular specifications some of these models are as follow:

- Homogeneous flow model (HFM): the two-phase loss multiplier, as suggested by saba & Lahey (1984), was correlated by

$$_{\phi}=\rho_{l}/\rho_{ll1} \tag{2.16}$$

 K_{12} and K_{13} estimated from single phase flow experiments, and the homogeneous densities were determined from:

$$\rho_{III} = \left[\frac{(1 - x_i)}{\rho_t} + \frac{x_i}{\rho_x} \right]^{-1} \tag{2.17}$$

Where ρ_g is the gas density and i = 1, 2 and 3 for the inlet, run and branch respectively.

- Separated flow model (SFM): the momentum-weighted densities were used for ρ_1 and ρ_2 in equation (2.14), as proposed by Fouda & Rhodes (1974), where

$$\rho_{Mi} = \left[\frac{(1 - x_i)^2}{\rho_t (1 - \alpha_i)} + \frac{x_i^2}{\alpha_i \rho_s} \right]^{-1} \quad \mathbf{i} = 1, 2, 3$$
 (2.18)

where ρ_M is the momentum density and α is the void fraction evaluated from the correlation of Rouhani (1969), $\phi = \frac{\rho_t}{\rho_{M1}}$, and the energy weighted densities were used for ρ_1 and ρ_3 in equation (2.15), as proposed by Saba & Lahey (1984), where

$$\rho_{Ni} = \left[\frac{(1 - x_i)^3}{\rho^2 t (1 - \alpha_i)^2} + \frac{x_i^3}{\alpha_i^2 \rho_g^2} \right]^{-0.5} \quad \mathbf{i} = 1, 2, 3$$
 (2.19)

where ρ_E is the energy density and α is the void fraction calculated from the Rouhani's (1969) correlations, as

$$\alpha_i = \frac{x_i}{x_i + (S_i/R)(1-x_i)}$$
 (2.20)

where, S_i is the velocity ratio and $R = \rho_i/\rho_g$.

$$S_{i} = \frac{\rho_{l}}{(1-x_{i})} \left(\frac{(1+0.12(1-x_{i}))}{\rho_{H}} + \frac{(\frac{1.18}{\sqrt{\rho_{l}}}(g \times \sigma \times (\rho_{l} - \rho_{G}))^{\frac{1}{4}}}{G_{i}} - \frac{x_{i}}{\rho_{G}})(2.21) \right)$$

as g is the acceleration due to gravity and σ is the surface tension.

- Hwang & Lahey (1988) model (IILM): in equation (2.14), this model employs $\rho_1 = \rho_{M1}$, $\rho_2 = \rho_{H2}$. The correlation of Zuber & Frindlay (1965) was used in calculating the void fraction, as recommended by Hwang & Lahey. In equation (2.15), the HLM specifies $\rho_1 = \rho_{K1}$, $\rho_3 = \rho_{H3}$ and $\phi = \rho_L \rho_{H3} / \rho_{H1}^2$.
- Ballyk et al., (1988) model (BM): in equation (2.14) this model specifies $\rho_1 = \rho_{M1}$, $\rho_3 = \rho_{H3}$ and $K_{12} = 1$. In equation (2.15), the BM specifies $\rho_3 = \rho_{K3}$, an equivalent density for ρ_1 and an empirical formulation for ϕ as a function of extraction rate.

- Reimann & Seeger (1986)model (RSM): for ΔP_{13} , this model uses equation (2.15) with $\rho_1 = \rho_{II1}$, $\rho_3 = \rho_{II3}$ and $\phi = \rho_I \rho_{II3} / \rho_{II}^2$. A different method was proposed for ΔP_{12} by splitting it into a reversible pressure difference between the inlet and the throat of a venacontracta in the run and a second pressure difference between the venacontracta and a position down stream in the run.

3. EXPERIMENTAL SET UP AND PROCEDURE

3.1 Introduction

The two-phase flow test loop has a lot of connections using different type of fittings (tee, bend, elbow, valve...) at different branch orientations (horizontal, vertical upward and down ward), and different flow entrance positions or may be more than one inlet. For the purpose of the present study, a two-phase flow test loop was designed and built using a T-junction oriented horizontally, this loop is shown in figure (3.1).

This chapter is divided into five sections. Section (3.2) describes the experimental set up. Section (3.3) describes the building steps of the test loop. The instruments used in the measurements was described In section (3.4). Lastly, section (3.5) presents the experimental procedure. Two tests were carried out, these include: single-phase flow and two-phase flow.

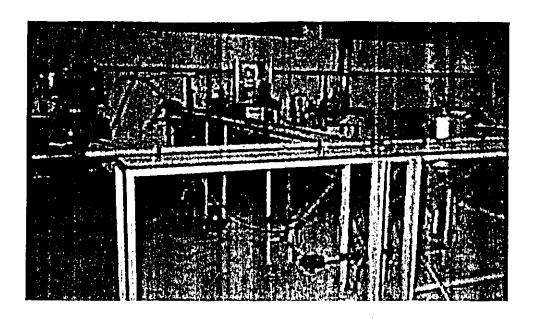


Figure (3.1) The experimental test system.

3.2 Experimental set up

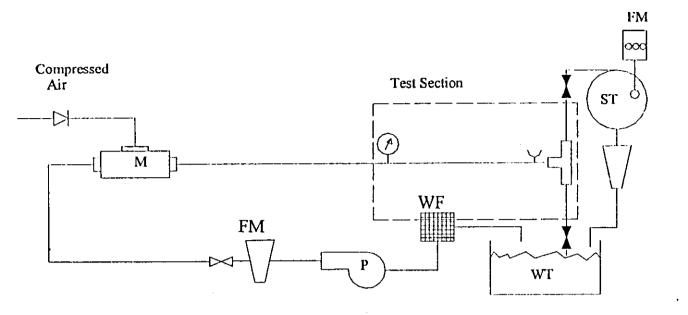
The test loop shown in figure (3.2) was designed and constructed in order to be used in the present experimental program involving a T-junction with a horizontal inlet and branches orientation. The test loop consists of water filter, centrifugal pump, mixer, air compressor, way valves, PVC pipes and T-junction, pressure taps and one pressure gauge, air-water separator and water tank.

The following sections will describe the components of the test loop and summarize the manufacturing processes that were applied to build them.

3.2.1 Water pump

A centrifugal pump (type 2CPm, 2900 rpm) with the following specifications $10\text{-}80~\ell/\text{min}$ flow rate, 40-15~m head, $H_{\text{max}}\text{=}41~\text{m}$, $Q_{\text{max}}\text{=}80~\ell/\text{min}$ has been used to supply water to the test loop. The inlet duct conducts the fluid into the center of the impeller, due to impeller rotation, the fluid is centrifuged outwards, towards the outside diameter of the impeller.

The high velocity of the fluid at the exit from the impeller is converted into pressure energy by velocity reduction in the volute, which acts basically as a fluid collection channel, the high pressure fluid that passes through the pump discharged to the test loop.



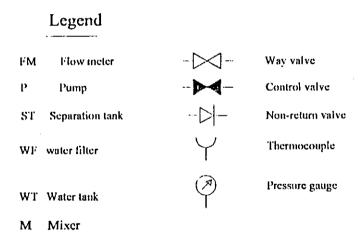


Figure (3.2) Experimental test facility.

3.2.2 Air compressor

"Gilkes tutor reciprocating air compressor" used in this study which is basically a simple mechanical device, that supplies air to the test loop. It can be operated at a maximum pressure of 10.3 bar nominal, 425 rpm minimum speed and 850 rpm as a maximum speed.

shown in figure (3.3), a photograph for the mixer is also shown in figure (3.4).

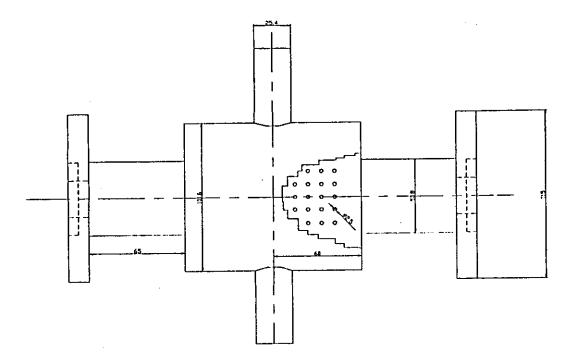


Figure (3.3) A schematic plot of the Mixer.

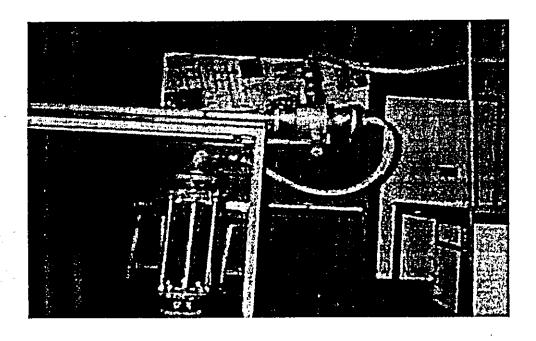


Figure (3.4) Photograph of the mixer.

3.2.5 Air-Water Separator

An air-water separator was designed and manufactured from 1.25mm thickness galvanized metal sheet as shown in figure (3.5). It is an essential element that enables us to separate air from water at outlets. It has simply a sink at the lower end and a vent at the upper one. The sink was connected to the rotometer, where the outlet water mass flow rate was measured, while the air velocity was measured by a pitot tube placed at the vent. Separation phenomena depends on a free vortex principle. Photogrph of the air-water separator is shown in figure (3.6)

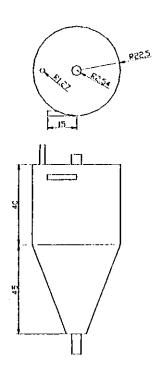


Figure (3.5) A schematic plot of air-water separator.

3.2.6 Valves and T-Junction:

Two valves and a sharp edge T-junction that made from PVC 32-1", UNE 53112 and PN-1.6 were used to control the flow rates and the system

pressure. The T-junction is the vital element at which the pressure drop within is studied. A non-return valve was set in the compressor supply line to avoid water entering to the compressor.

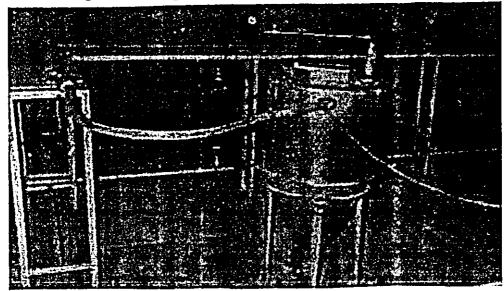


Figure (3.6) Photograph of the air-water separator.

3.2.7 Pressure Gauge and Taps

A 1 bar scale bourdon gauge, figure (3.10), is used to justify the system pressure, placed at a distance of (60×diameter) from the center of the T-junction. Thirty two pressure taps were manufactured from copper, with 1.5mm inner diameter distributed regularly at (2d, 5d, 10d, 20d and 40d) from the center of the T-junction, at the inlet and the branches as shown in figure (3.7). A schematic plot of the pressure tap is also shown in also shown.

3.2.8 Pressure taps holder:

This device was designed and manufactured in order to measure the pressure drop through the 15 taps. A pressure holder as shown in fig(3.8), was manufactured from a 60cm long PVC pipe closed from one end and have a valve from the other. 15 drills (6mm diameter) were arranged on the upper

face, connected to the pressure taps at the test loop and one on the other face connected to the U-manometer

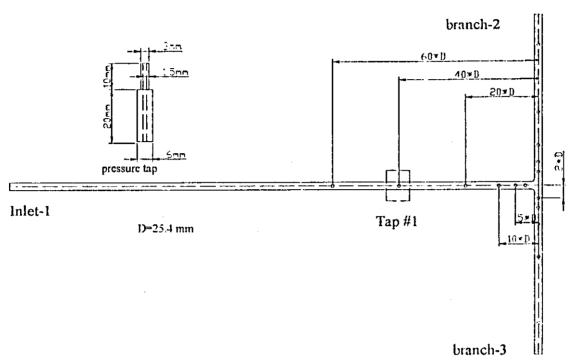


Figure (3.7) Layout of pressure taps.

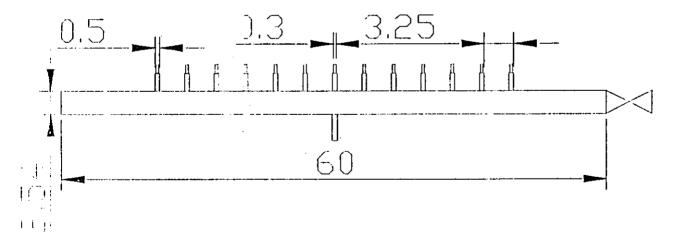


Figure (3.8) Layout of the pressure taps holder.

3.3 Building the test loop

The following steps were followed to build the test section shown in figure (3.1), used in this study:

- 1. three stands (420×30×120,120×30×120,120×30×120) length, width and height, were manufactured and fixed to the ground.
- 2. thirty two pressure taps were manufactured from copper and placed at regular displacements at the PVC pipes and holder.
- 3. Two rotometers were placed at the stand legs.
- 4. PVC pipes were connected and glued with the T-junction and placed at the stands.
- 5. two way valves were connected to the ends of the branches to control the mass flow rates.
- 6. The pressure taps holder was fixed at the branch stand legs.
- 7. Silicon pipes with 1.5 mm inner diameter were connected to the pressure taps at the PVC pipes with the pressure taps holder, as shown in figure (3.9).
- 8. a 1bar pressure gauge was placed at the stand and connected with the system pressure tap through a tee, while the other side of the tee were connected with the U-manometer.

- 9. the mixer was placed at the inlet to the test section.
- 10. a centrifugal pump was placed to supply the system with water from the water tank.
- a rotometer was set at the stand legs after the pump to measure the inlet mass flow rate.
- 12. a way valve was placed at the rotometer outlet to control the inlet mass flow rate and connected with the mixer.
- 13. the air compressor was connected with the mixer by hose of one inch diameter having non-return valve at its end to avoid water entering to the compressor.
- 14. one of the branches returns water to the tank directly, while the other connected to the air-water separator.
- 15. air-water separator was manufactured from 1.25mm thickness, galvanized metal sheet, and connected to another rotometer to measure the outlet water mass flow rate. While the outlet of the rotometer returns water to the tank directly.
- 16. Pitot tube connected with the manometer was set at the air-water separator vent to measure air mass flow rate.
- 17. a digital thermometer was used to measure the temperature at the inlet to the T-junction.

18. a $5\mu m$ water filter was connected to the pump inlet, to assure pure water entering to the loop.

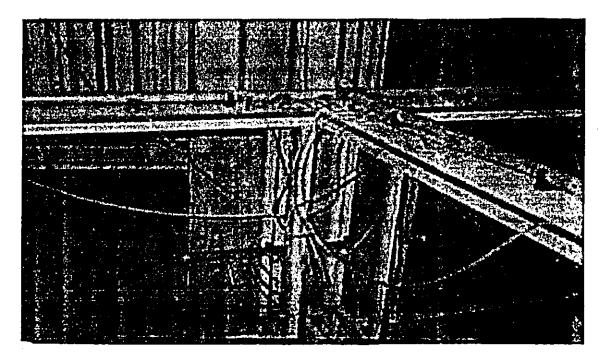


Figure (3.9) Photograph of the test section.

3.4 Measuring instrumentation

The instrumentation used during the test are introduced briefly in the following:

1. Pressure bourdon gauge: Bourdon gauge shown in figure (3.10) is typical of the devices used for measuring gauge pressures. The pressure element is a hollow, curved, flat metallic tube, closed at one end with the other end connected to the pressure to be measured. When the internal pressure is increased, the tube tends to straighten, pulling on a linkage to which is attached a pointer and causing the pointer to move. The dial reads zero when inside and outside of the tube are at the same pressure, regardless of its particular value. A 1bar scale gauge was used to measure the system

pressure, set at the inlet pipe at $(60 \times d)$ from the centerline of the T-junction, the minimum reading of the gauge is 10 Pa.

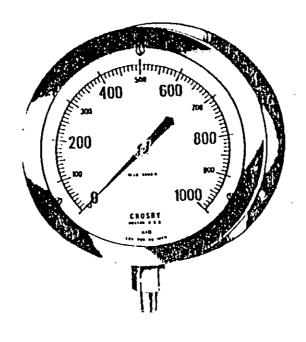


Figure (3.10) Bourdon gauge

2. Rotometers: Two rotometers were used, the first measures the inlet mass flow rate of water to the mixer, with a range of (900-9000L/hr), with minimum reading of 100 L/hr. And the other measures the water flow out from the separator with a range of (180-1800L/hr) with minimum reading of 20 L/hr. The principle of rotometers as shown in figure (3.11), is to maintain the float in equilibrium and as it is of constant weight, the pressure difference is independent of discharge. The cause of the pressure difference is the head loss associated with the high velocity of water around the float periphery. Since this head loss is constant then the peripheral velocity is constant. To maintain a constant velocity with a varying discharge rate, the cross-sectional area through which this high velocity occurs must vary. This variation of cross sectional area will increase as the float moves up and down the tapered rotometer tube.

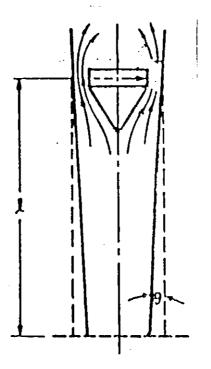


Figure (3.11) Principle of the rotometer

- 3. Thermocouple and digital thermometer: One copper constantan thermocouple wire type was used to measure the temperature at the inlet to the T-junction. It was connected to a digital thermometer (305) type with 0.1° high resolution, maximum record and offset adjest.
- 4. Orifice: It is used to measure the inlet mass flow rate according to BS 1042 with d_1 =12.7mm and D=25.4mm:

$$m = 6.574 \times 10^{-3} \sqrt{\frac{\Delta P. P_3}{T_3}}$$
 Kg/s [3.1]

where ΔP = orifice differential head mm wg

 P_3 = orifice plate down stream pressure bar abs

$$=P_3'+P_0$$
 bar abs

where P' = pressure gauge reading bar abs

 $P_o = atmospheric pressure bar abs$

And
$$T_3 = T'_3 + 273.15$$
 K

5. Pitot tube: The outlet air mass flow rate at the separator was calculated from the air velocity measured by means of a pitot tube. If h is the liquid dynamic head in mm then the velocity is given by:

$$V = \sqrt{\frac{2.\rho_{\ell}.g.h}{\rho_{a}}} \qquad \text{m/s}$$
 (3.2)

then the mass flow rate, m, is given by

$$m = P_a. V. A_p \tag{3.3}$$

where, A_p is the air-water separator vent cross sectional area.

6. U-manometer: U-tube containing mercury was used to measure the pressure loss across the test section points. Considering fig.(3.12) since the pressure difference was taken at the same elevation and pipe diameter then from Bernoulli's equation:

$$\frac{P_1 - P_2}{\rho_{H_2O}g} = h_L \tag{3.4}$$

pressure in both limbs of the U-tube are equal at level 00. Therefore equating pressures at 00:

$$P_2 - \rho_{H_2O}g(X+Y) + \rho_{Hg}gX = P_1 - \rho_{H_2O}(gY_1)$$
 (3.5)

arrange above equations taking the specific gravity of mercury as 13.6 gives:

$$h_L = 126X \tag{3.6}$$

$$\Delta P_{1x} = 12.6g\rho_{H_2O}X\tag{3.7}$$

where X = Head loss at the inlet or branches taps

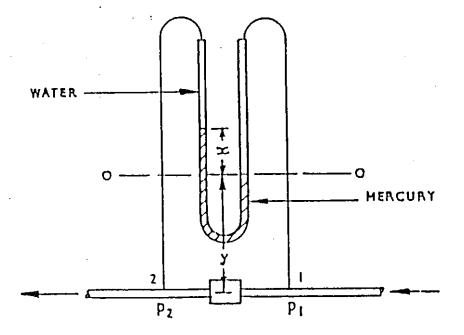


Figure (3.12) mercury U-tube manometer used to measure the pressure loss.

3.5 Experimental tests and procedure

The experimental test facility used in this study was placed in the Fluid Mechanics Laboratory of Mechanical Engineering department of Mu'tah University. The test facility was allowed to operate at different air and water mass flow rates. The data collected are:

- 1. Inlet and outlet water mass flow rates.
- 2. Inlet and outlet air mass flow rates.
- 3. System pressure (P_s).
- 4. Temperature at the inlet of the T-junction (T_1) .
- 5. Pressure difference values at the fifteen pressure tap locations.

3.5.1 Single-Phase Flow Test

The objective of this test is to obtain the loss coefficient (K) for the material (PVC) used in this study.

Single-phase (liquid) pressure drop data were obtained in this investigation using seven test runs, corresponding to a mass flux of 709.8 Kg/s.m² and $0.0 \le G_2/G_1 \le 1.0$. Magnitudes of the loss coefficient (K) were calculated from the measured values of Δ_{12}^P .

Obtained results are tabulated in table (B₁-B₃), at which pressure differences at different extraction rates were taken from the U-manometer. The inlet and outlet water mass fluxes were taken from rotometers. Details of procedure is the same as in section (3.5.2) except that there is no air flow.

3.5.2 Two-Phase Flow Test

The objective of this investigation is to find the pressure difference across the T-junction.

The test system shown in figure (3.2) consists of a 5 μ m water filter that filters the water enters to the system. A rotometer was set after the pump to measure the water flow rates (1000, 1100, 1200, 1300, 1400 and 1500)L/hr entering the mixer. Air compressor supplies air to the system at which an orifice is used to measure the air mass flow rates. A non-return valve was set on the compressor supply line to avoid water entering to the compressor.

Mixer was used to mix air with water, and supply regular stream lines of the mixture (air & water) to the inlet of the test loop. This flow is divided into the two branches, during the flow the pressure difference was measured by using pressure taps at 15 points along the inlet and branches with respect to the system pressure, which was measured by a pressure gauge. The 15 pressure taps were connected to the U-manometer through a pressure holder. In order to read the pressure difference directly from the U-manometer a line from the system pressure was connected to the other edge of the U-manometer. The outlet air and water mass flow rates were measured by a means of a pitot tube and a rotometer at the outlets of the air-water separator, The data collected are tabulated in tables (A.2.1-A.3).

4. RESULTS AND DISCUSSION

4.1 Introduction

In this chapter, the pressure differences are studied with detailed analysis. This study is based on the experimental results, obtained from the performed tests.

Results and sample calculation are presented in section (4.2), and the final part deals with the discussion and analysis of the results obtained.

4.2 Calculations and Results

The calculations and curves presented in this study were performed by using Excel package.

The data collected during all tests are tabulated in appendix A. All measured pressure difference values (four to six readings within one minute) at each tap location were averaged to ensure low errors due to fluctuations in head measurements. The calculations are based on the equations presented previously in chapters two and three. The results and sample calculation of all tests are presented in the following sub-sections.

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4.2.1 Single-Phase Flow Test

This test was performed at inlet mass flux of 709.8 kg/s.m² and extraction rates of $0.0 \le G_2/G_1 \le 1.0$. The data collected during the test are presented in tables (A.1) in appendix A. From obtained data the pressure difference (ΔP_{12}) and the loss coefficient (K) were obtained and listed in tables (B.1-B.3) in appendix B.

At each extraction rate (G_2/G_1) , the U-manometer heads of the taps at the inlet and at branch-2 (H_1, H_2) in mmHg and the inlet temperature to the T-junction (T_1) were recorded. Each head measurement is a group of five readings for the taps distributed along each test leg. The pressure differences $(\Delta^i_{M})_i$ was estimated from equation (3.7), where subscripts s, t and i are representing system pressure, tap pressure and test leg number, respectively. For the purpose of simplifying calculations in order to be consistent with others data, the pressure differences were transformed to be $(\Delta^i_M)_i$, which is the pressure difference between each tap against tap#1 as indicated in the sample calculation. The values of $(\Delta^i_M)_i$ were plotted against tap position as shown in figure (4.1), the average pressures and then the pressure differences across the T-junction can be taken by extrapolating the fully developed values to the centerline of the T-junction.

From obtained results the loss coefficient values (K) can be calculated from equation (2.10) and plotted against the extraction rates (G_2/G_1) as shown in figure (4.2). The equation that represents the loss coefficient as a function of extraction rates can be obtained from the available Excel polynomial curve fitting as:

$$K=1.0799(G_2/G_1)^2-0.6686(G_2/G_1)+1.1321$$
(4.1)

This relation was compared with previous results as shown in figure (4.3). The values of water density and the acceleration due to gravity were assumed to be constant at given temperatures as (996 kg/m³, 9.81 m/s²) Mills, 1995, respectively, since the variation of temperatures is relatively small. The inlet volumetric flow rates (ℓ/hr) were taken in term of mass flux directly in the calculations according to the following relation:

$$G = V(L/hr) * \frac{1hr}{3600 \sec} * \frac{1m^3}{1000L} * \rho_{n_2o} (\frac{kg}{m^3}) * \frac{1}{A} (m^2)$$
 (4.2)

$$= 0.546 * \dot{V}$$
 (4.3)

Sample calculation

Referring to table (A.1)the measured values of average pressure heads (II₁, H_2) at each pressure location for the extraction rate $G_2/G_1=0.3846$ are tabulated in table (4.1)

Table (4.1) average pressure head for extraction rate G₂/G₁=0.3846

G ₂	G ₂ /G ₁	Distance at	I-I ₁	Distance at	H ₂
(kg/s.m ²)		inlet	(mm Hg)	branch-2	(mm Hg)
		(m)		(m)	
		-1.016	1.5	0.0508	4
		-0.508	2.75	0.127	4.25
273	0.3846	-0.254	3.5	0.254	4.5
		-0.127	3.75	0.508	4.75
		-0.0508	3.5	1.016	5

The pressure difference values (ΔP_{st}) can be calculated from equation (3.7):

$$(\Delta P_{s1})_1 = 126g\rho_{H_2O}X$$
, where $X = H_1$ or H_2
= 12.6 * 9.81 * 996 * 0.0015
= 184.6674 Pa
 $(\Delta P_{s2})_1 = 12.6g\rho_{H_2O}X$
= 12.6 * 9.81 * 996 * 0.00275
= 338.5568 Pa

This value is used to find the pressure difference with respect to tap #1 as follows:

$$(\Delta I_{1t}^{2})_{1} = (\Delta I_{st}^{2})_{1} - (\Delta I_{st}^{2})_{2}$$

$$(\Delta I_{12}^{2})_{1} = 184.6674 - 338.5568$$

$$= -153.887 \text{ Pa}$$
(4.4)

Results are presented in table (4.2)

Table (4.2) pressure differences for $G_2/G_1=0.3846$

Distance	$(\Delta P_{st})_{1}$	$(\Delta P_{1t})_1$	Distance at	$(\Delta l_{st}^2)_2$	$(\Delta l_{1l}^2)_2$
at inlet	bar	bar	branch-2	bar	bar
(m)			(m)		
-1.016	184.6674	0.0	0.0508	492.4463	-307.776
-0.508	338.5568	-153.887	0.127	523.2242	-338.554
-0.254	430.8905	-246.221	0.254	554.0021	-369.332
-0.127	461.6684	-276.998	0.508	584.78	-400.11
-0.0508	430.8905	-246.221	1.016	615.5579	-430.888

The values of pressure difference with respect to tap #1 (Δ_{1t}^2) were plotted against taps positions, as shown in figure (4.1) and the pressure difference across the T-junction can be obtained directly from the plot at this extraction rate to be

$$\Delta P_{12} = 55 \text{ Pa}$$

Rearranging equation (2.10) gives the loss coefficient value as:

$$K = \frac{2\rho_{u_2o}\Delta P}{G_1^2} - (\frac{G_2}{G_1})^2 + 1$$

$$= \frac{2*996*55}{709.8^2} - 0.3846^2 + 1$$

$$= 1.069553$$
(4.5)

The loss coefficient values (K) were computed for each extraction rate and plotted against it, as shown in figure (4.2). For the purpose of comparison, the loss coefficient values (K) from previous works are also estimated using equations (2.11-2.13) and plotted against extraction rates as shown in figure (4.3). Results obtained are listed in table (B.1-B.3).

4.2.2 Two-phase Flow Tests

These tests were performed at different inlet water mass fluxes of (547, 601, 656, 710, 820) $kg/s.m^2$ and air mass flow rates of (0.000452, 0.000586, 0.000641, 0.000642, 0.000786)kg/s, respectively.

The data collected during the tests are shown in table (A.2, A.3) in appendix A. The average pressures (P_1 , P_2 and P_3) and the pressure difference values across the T-junction (ΔP_{12} , ΔP_{13}) were obtained. These results are shown in table (B.7)in appendix B.

Five tests were carried out with constant air mass flow rate and different water mass fluxes for each test. During each test, the U-manometer heads of the taps at the inlet and branches (H_1 , H_2 and H_3), the inlet temperature to the T-junction (T_1), the orifice differential head ΔP , the orifice plate downstream

pressure, the atmospheric pressure, the dynamic pressure at the air-water separator vent and the inlet and outlet water volume flow rates were recorded.

The values of the pressure heads were substituted in equation (3.7) to obtain the pressure differences $(\Delta l_{st}^2)_i$ and then the pressure differences against tap#1 $(\Delta l_{1t}^2)_i$ from equations (4.4). Results were plotted against measurement positions as shown in figures (4.4-4.32). The average pressures (P₁, P₂ and P₃) and the pressure differences $(\Delta l_{12}^2, \Delta l_{13}^2)$ were obtained from these figures for all tests.

The total (air and water) mass fluxes $(G_1, G_2 \text{ and } G_3)$ and the qualities $(x_1, x_2 \text{ and } x_3)$ were estimated from equations (2.2, 2.3). The pressure differences $(\Delta P_{12}, \Delta P_{13})$ were plotted against inlet mass flux (G_1) and extraction rates $(G_2/G_1, G_3/G_1)$ at constant inlet qualities, and against inlet quality (x_1) at constant inlet mass flux, as shown in figures (4.33-4.38). Here, due to the lack of repeatability of the same values of inlet mass flux and inlet quality, values within $\pm 5\%$ error were taken the same. The superficial gas and liquid velocities are also estimated and used to determine the flow regime caused due to two-phase flow according to Mandhane, 1974.

For the purpose of comparison with other models, the reproductive accuracy (Schmidt, 1997) of obtained results is found on the basis of dimensionless statistical number: standard deviation S_{ln} of the logarithmic ratios of measured (subscript meas.) and predicted (subscript SFM or RSM) pressure differences and ploted on the parity plot as shown in figure (4.39), according to the following definitions

The logarithmic ratio between measured and predicted pressure difference $(X_{ln,i})$

$$X_{ln,i} = \ln\left(\frac{\Delta P_{i,meas.}}{\Delta P_{i,mead.}}\right) \tag{4.6}$$

The standard deviation of logarithmic ratios (S_{In})

$$S_{ln} = \exp\left\{\sqrt{\frac{\sum_{i=1}^{n} X_{ln,i}^{2}}{(n-f-1)}} - 1\right\}$$
(4.7)

Where n = number of experimental values, f = number of variables

Sample calculation

Referring to table (A.2.1-A.2.5), the measured values of average pressure heads (H₁, H₂ and H₃) for each pressure tap on the test leg, at inlet water volume flow rate of 1000 ℓ/lr and inlet air mass flow rate of 0.000641 kg/s at system pressure P_s of 1 bar are listed in table (4.3).

Table (4.3) Average pressure heads for $V_1 = 1000 \ \ell/hr$, $m_a = 0.000641 \text{ kg/s}$

\dot{V}_1	V ₂	Measureme	H_1	Measure	H ₂	I I ₃
(ℓ/hr)	(ℓ/hr)	nt positions	(mm Hg)	ment	(mm Hg)	(mm Hg)
		(m)		positions		
=				(m)		
1000	370	-1.016	3.83	0.0508	7.33	9
		-0.508	6	0.127	8	10.67
		-0.254	6.67	0.254	8.67	11.33
		-0.127	7	0.508	9	11.67
		-0.0508	6.67	1.016	9.33	12.33

The pressure difference values $(\Delta l_{st})_i$ can be calculated from equation (3.7) as:

$$(\Delta P_{st})_i = 12.6g\rho_{II,O}X$$

where,
$$X=H_1$$
, H_2 and H_3

$$(\Delta l_{sl}^{2})_{1} = 12.6 * 9.81 * 996 * 0.00383$$

= 471.5173 Pa

$$(\Delta l_{s2}^{3})_{1} = 12.6 * 9.81 * 996 * 0.006$$

= 738.6692 Pa

These values are used to find out the pressure difference with respect to tap #I from equation (4.4) as follows:

$$(\Delta_{1t}^{2})_{1} = (\Delta_{s1}^{2})_{1} - (\Delta_{s2}^{2})_{1}$$

 $(\Delta_{12}^{2})_{1} = 471.5173 - 738.6692$
= -267.1521 Pa

Results are shown in table (4.4)

Table (4.4) Pressure differences for $\dot{V}_1 = 1000 \ell/hr$,

 $m_a = 0.00064 \, \text{kg/s}$

$(\Delta P_{st})_1$	$(\Delta P_{1t})_1$	$(\Delta P_{st})_2$	$(\Delta l_{1t}^{2})_{2}$	$(\Delta l_{st}^{\gamma})_3$	$(\Delta l_{1t}^2)_3$
bar	bar	bar	bar	bar	bar
471.5173	0.0	902.4079	-430.8905	1108.004	-636.4869
738.6695	-267.1521	984.8926	-513.3753	1313.601	-842.0832
821.1542	-349.6369	1067.377	-595.86	1394.854	-923,3369
861.781	-390.2637	1108.004	-636.4868	1436.712	-965.1948
821.1542	-349,6369	1148.631	-677.1137	1517.966	-1046.4484

The values of pressure difference with respect to tap #1 (Δ_H^2) were plotted against the tap positions, as shown in figures(4.4-4.32). The pressure difference across the T-junction can be obtained directly from the plot as:

$$\Delta P_{12} = 95$$
 Pa

$$\Delta P_{13} = 390 \, \text{Pa}$$

Results are tabulated in table (B.7)

For the purpose of comparison with other models, number of parameters (homogeneous, momentum, energy densities and two-phase loss multiplier) must be estimated first. The parameters are calculated in the followings. From table (A.2, A.3), the inlet and outlets air mass flow rates can be calculated using equations (3.1, 3.2, 3.3), given in table (B.4.1-B.4.5). From equation (3.1) the inlet air mass flow rate is:

$$P_3 = P_0 + P_3'$$

= 0.942 + 0.058
= 1 bar

$$m_{a1} = 6.574 \times 10^{-3} \sqrt{\frac{\Delta P. P_3}{T_3}}$$

$$=6.574\times10^{-3}\sqrt{\frac{3\times1}{315.15}}$$

= 0.000641 kg/s

The air mass flow rate of branch-2 is found from equations(3.2, 3.3)

$$V = \sqrt{\frac{2.\rho_{\ell}.g.h}{\rho_{a}}} \quad \text{m/s}$$

$$= \sqrt{\frac{2 \times 787 \times 9.81 \times 1 \times 10^{-3}}{1.177}}$$

$$= 3.622 \quad \text{m/s}$$

then the mass flow rate, m_{a2} , is given by

 $m_{a2} = P_a$. V. A_p , where, A_p is the vent pipe cross sectional area.

=1.177
$$\times 3.622 \times \frac{\pi}{4} \times (8 \times 10^{-3})^2$$

= 0.000214 kg/s

$$m_{a3} = m_{a1} - m_{a2}$$

= 0.000641 - 0.000214
= 0.000427 kg/s

The water mass flow rate can be found from the following equation

$$m_{wl} = V(L/hr) * \frac{1hr}{3600 \sec} * \frac{1m^3}{1000L} * \rho_{H_2O}(\frac{kg}{m^3})$$

where,
$$\rho_{H_2O} = 996 kg / m^3$$

$$m_{\rm wl} = 0.000276666*V_1$$

$$= 0.000276666*1000$$

$$= 0.276666 \text{ kg/s}$$

Equation (2.3) is used to find out the total mass flux as:

$$G_i = G_{iw} + G_{ia} \qquad i = 1,2,...$$

where subscript w represents for water

$$= 0.546 \times \dot{V}_i + \frac{m_{ia}}{A} \qquad \text{for } i = 1$$

$$G_1=0.546\times 1000 + \frac{0.000214}{(\frac{\pi}{4}\times 8\times 10^{-3})^2}$$

$$= 547.2658 \text{ kg/s.m}^2$$

The same is done for G_2 , G_3 at the outlets, results are shown in table (B.5.1-B.5.5).

The inlet and outlet qualities can be obtained from equation (2.2) as follow

$$x = \frac{m_a}{m_a + m_w}$$

$$x_1 = \frac{0.00641}{0.00641 + 0.276666}$$

$$= 0.002313$$

and the same procedure is followed for the remain qualities. Results are listed in table(B.7).

The homogeneous densities $(\rho_{H_1}, \rho_{H_2}, \rho_{H_3})$ can be estimated from equation (2.17)

$$\rho_{III} = \left[\frac{(1 - x_i)}{\rho_\ell} + \frac{x_i}{\rho_g} \right]^{-1}$$

$$= \left[\frac{(1 - 0.002313)}{996} + \frac{0.002313}{1.177} \right]^{-1}$$

$$= 344.8235 \text{ kg/m}^3$$

the results are given in table (B.8).

In order to evaluate momentum density values the void fraction should be estimated first. The velocity ratio (S_i) and then the void fraction (α_i) were obtained using equations (2.20,2.21). Results substituted in equation (2.18) in order to obtain the momentum density values that shown in table (B.8).

$$S_{i} = \frac{\rho_{i}}{(1-x_{i})} \left(\frac{(1+0.12(1-x_{i}))}{\rho_{H_{1}}} + \frac{\left(\frac{1.18}{\sqrt{\rho_{l}}} (g \times \sigma \times (\rho_{l} - \rho_{G}))^{\frac{1}{4}}}{G_{i}} - \frac{x_{i}}{\rho_{G}} \right)}{G_{i}}$$

$$x_{i} = 0.002313, \rho_{H1} = 337.0609kg/m^{3}, \rho_{l} = 996, \sigma = 0.0819N/m$$
For
$$G_{1} = 547.2658kg/s.m^{2}, \rho_{G} = 1.177kg/m^{3}, g = 9.81m/s^{2}$$

$$S_t = 1.705633$$

$$\alpha_i = \frac{x_i}{x_i + (S_i/R)(1-x_i)} ,$$

$$R = \frac{\rho_l}{\rho_G} = 846.22$$

$$\alpha_1 = \frac{0.002313}{0.002313 + (\frac{1.705633}{846.22})(1 - 0.002313)}$$

$$\alpha_1 = 0.534926$$

$$\rho_{M} = \left[\frac{(1-x_i)^2}{\rho_i (1-\alpha_i)} + \frac{x_i^2}{\alpha_i \rho_r} \right]^{-1}$$

$$\rho_{M1} = \left[\frac{(1 - 0.002313)^2}{996(1 - 0.534926)} + \frac{0.002313^2}{0.534926 \times 1.177}\right]$$

$$\rho_{\rm M1} = 463.5315 \text{ kg/m}^3$$

Equation (2.19) is used to find another term called Energy density, which is also required in calculations. Results are tabulated in table (B.9).

$$\rho_{EI} = \left[\frac{(1 - x_i)^3}{\rho^2 \iota (1 - \alpha_i)^2} + \frac{x_i^3}{\alpha_i^2 \rho^2 g} \right]^{-1}$$

$$\rho_{EI} = \left[\frac{(1 - 0.002313)^3}{996^2 (1 - 0.534926)^2} + \frac{0.002313^3}{0.534926^2 \times 1.177^2} \right]^{-0.5}$$

 $=463.2663 \text{ kg/m}^3$

the two-phase loss multiplier ϕ has different correlations, results shown in table(B.9). According to the separated flow model (SFM) and the Riemann & Seeger model (RSM) the values of ϕ can be estimated as follow

$$\phi_{SFM} = \frac{\rho_{I}}{\rho_{M1}} \quad \text{this valid for the two branches (2,3)}$$

$$= \frac{996}{463.6893}$$

$$= 2.14799$$

$$(\phi_{RSM})_{I} = \frac{\rho_{I} \times \rho_{III}}{\rho_{H1}}$$

$$(\phi_{RSM})_{2} = \frac{996 \times 360.4426}{337.0609}$$

$$= 3.159939$$

$$(\phi_{RSM})_{3} = \frac{996 \times 324.6952}{337.0609}$$

$$= 2.846547$$

After these calculations, the pressure differences (ΔP_{12} , ΔP_{13}) for the above models are:

$$(\Delta P_{12})_{SFM} = \frac{\rho_{II2}}{2} \left[(G_2^2/\rho_{E2}^2) - (G_1^2/\rho_{E1}^2) \right] + K_{12} \frac{G_1^2}{2\rho_{\ell}} \phi_{SFM}$$

$$= \frac{360.4426}{2} \left[\frac{202.4423^2}{559.6064^2} - \frac{547.2658^2}{463.2663^2} \right] + 1.032545 \frac{547.2658^2}{2 \times 996} 2.14799$$

$$= 105.6608 \text{ Pa}$$

$$(\Delta P_{13})_{SFM} = \frac{\rho_{H3}}{2} \left[(G_3^2/\rho_{E3}^2) - (G_1^2/\rho_{E1}^2) \right] + K_{13} \frac{G_1^2}{2\rho_{\ell}} \phi_{SFM}$$

$$= \frac{324.6952}{2} \left[\frac{344.8235^2}{478.9547^2} - \frac{547.2658^2}{463.2663^2} \right] + 1.139553 \frac{547.2658^2}{2 \times 996} 2.14799$$

$$= 225.7379 \text{ Pa}$$

$$(\Delta P_{12})_{RSM} = \frac{\rho_{H2}}{2} \left[\left(G_2^2 / \rho_{H2}^2 \right) - \left(G_1^2 / \rho_{H1}^2 \right) \right] + K_{12} \frac{G_1^2}{2\rho_I} (\phi_{RSM})_2.$$

$$= \frac{360.4426}{2} \left[\frac{202.4423^2}{360.4426^2} - \frac{547.2658^2}{337.0609^2} \right] + 1.032545 \frac{547.2658^2}{2 \times 996} 3.159939$$
$$= 72.31312 \text{ Pa}$$

$$(\Delta P_{13})_{RSM} = \frac{\rho_{H3}}{2} ((G_3^2 / \rho_{H3}^2) - (G_1^2 / \rho_{H1}^2)) + K_{13} \frac{G_1^2}{2\rho_{\ell}} (\phi_{RSM})_3$$

$$= \frac{3246952}{2} [(\frac{2024423^2}{3246952^2} - \frac{547.2658^2}{337.0609^2}] + 1.139553 \frac{547.2658^2}{2 \times 996} 2.846547$$

$$= 242.8258 \text{ Pa}$$

Results are tabulated in table (B.10).

Determining the inlet flow regime of each test required estimating the superficial liquid and gas velocities. These superficial velocities can be calculated using equations (2.7, 2.8). Calculated values were used to determine the inlet flow regime using Mandhane map, 1974.

$$J_{G1} = G_1 x_1 / \rho_g$$

$$= \frac{547.2658 \times 0.002313}{1.177}$$

$$= 1.075448 \text{ m/s}$$

$$J_{\ell 1} = G_1 (1-x) / \rho_{\ell}$$

$$= \frac{547.2658 \times (1-0.002313)}{996}$$

$$= 0.548193 \text{ m/s}$$

The flow regime obtained at this value is slug flow regime, and the same result is obtained for the remaining calculations.

The standard deviations between measured and predicted pressure difference values can be calculated from equations (4.7), sample calculation for measured

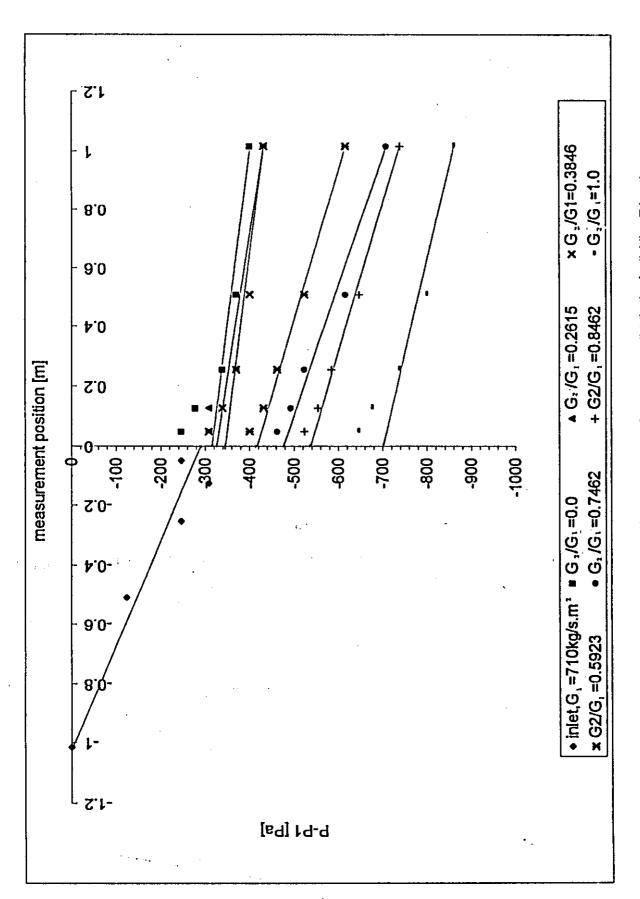


Figure (4.1) Schematic diagram of a typical single-phase flow pressure distribution in dividing T-junction

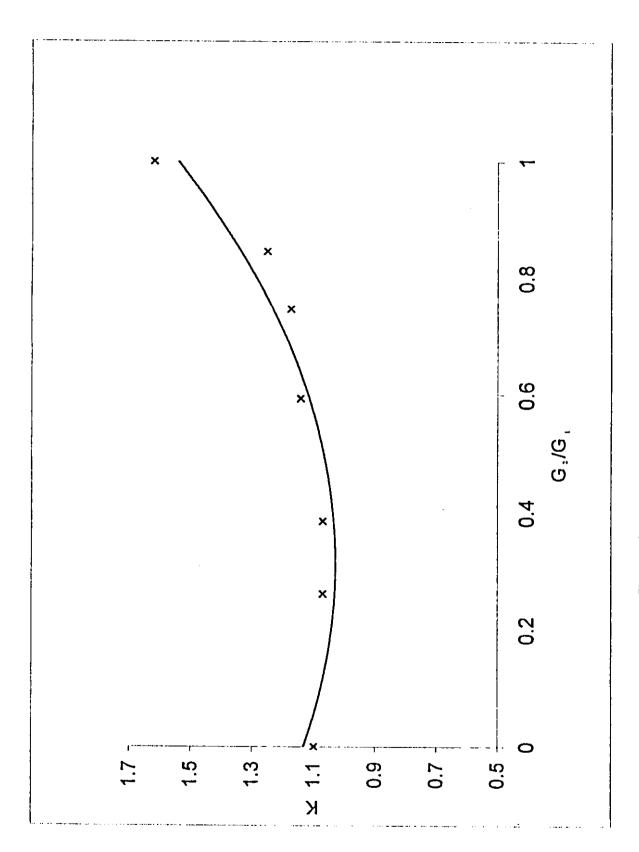


Figure (4.2) Present single-phase flow loss coefficient, k

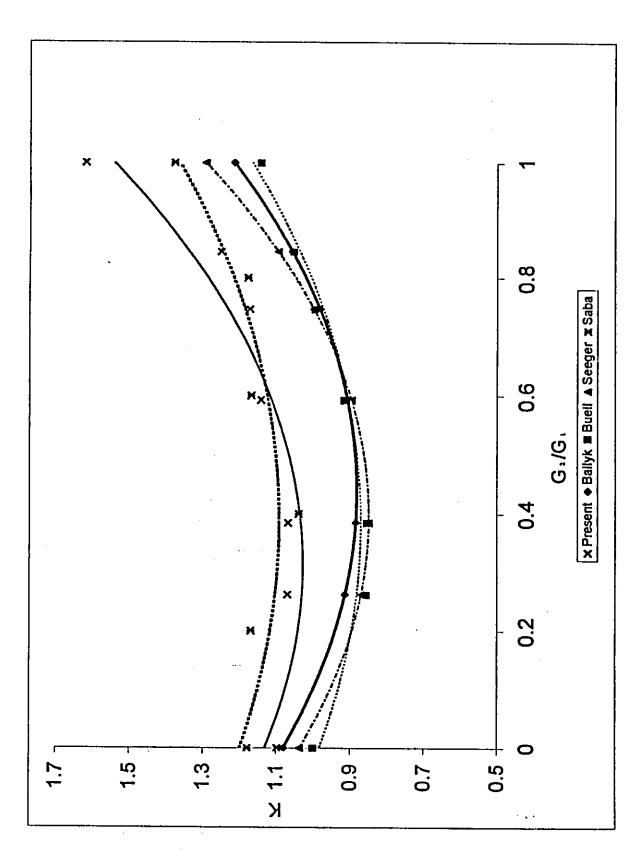


Figure (4.3) Single-phase flow loss coefficiemt, k

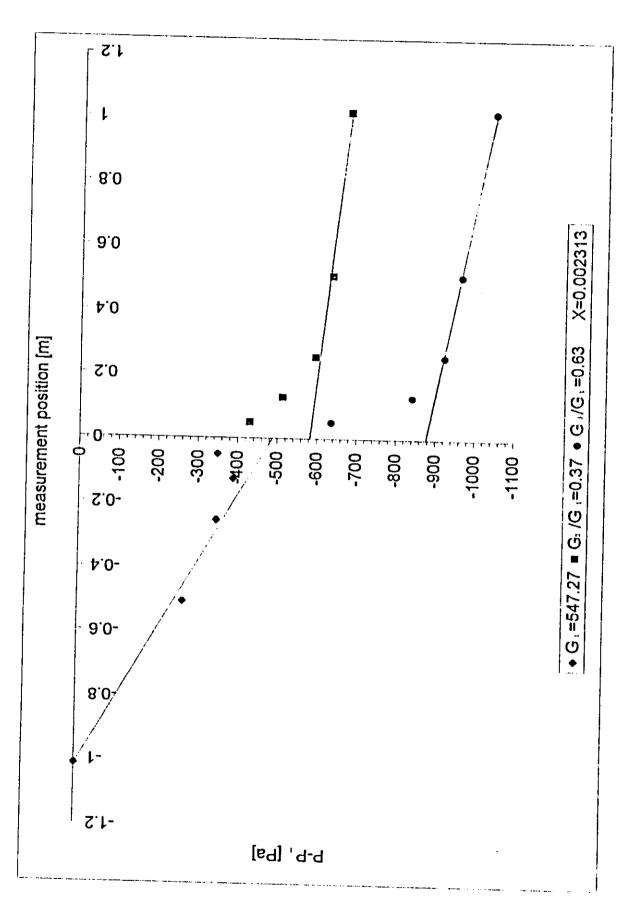


Figure (4.4) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

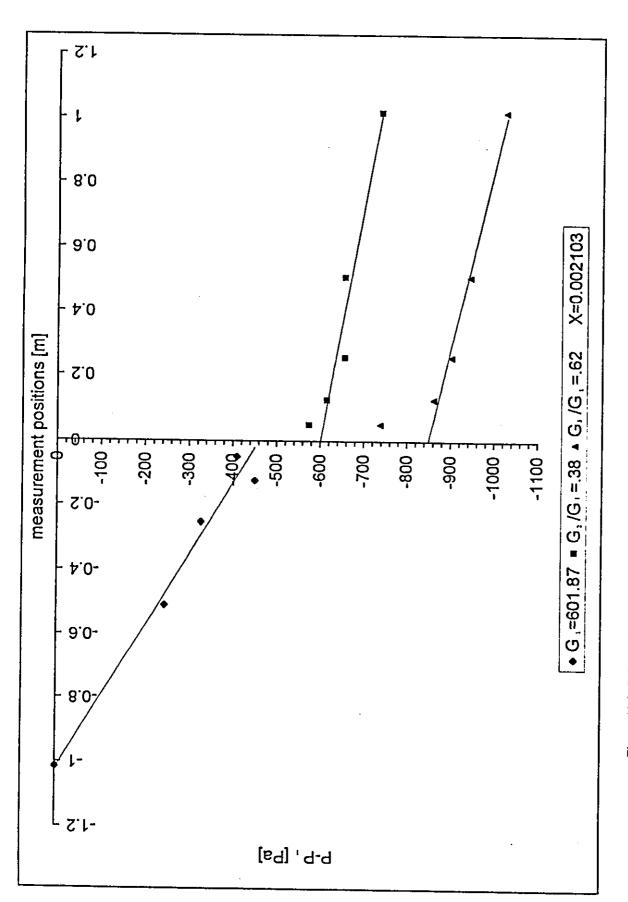


Figure (4.5) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

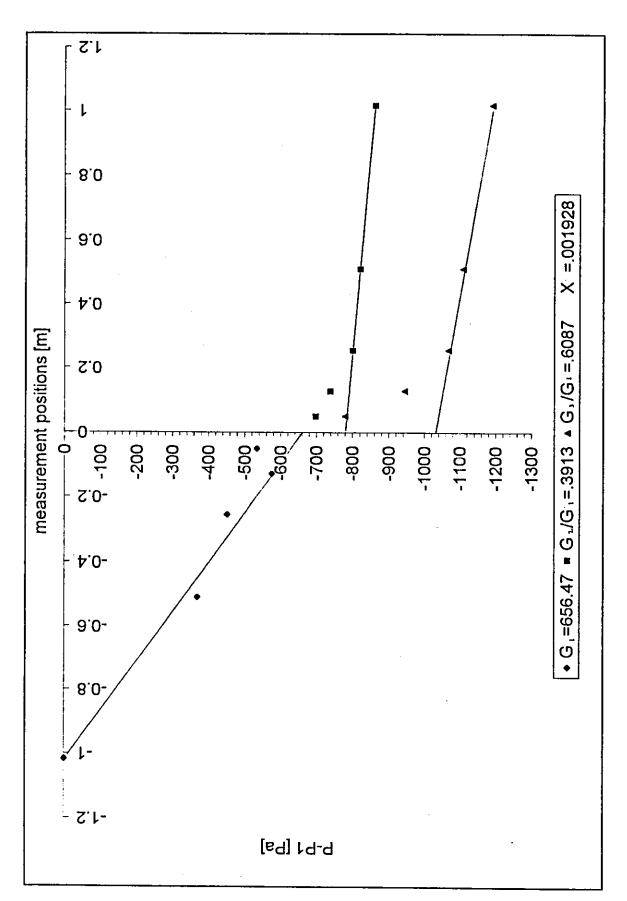


Figure (4.6) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

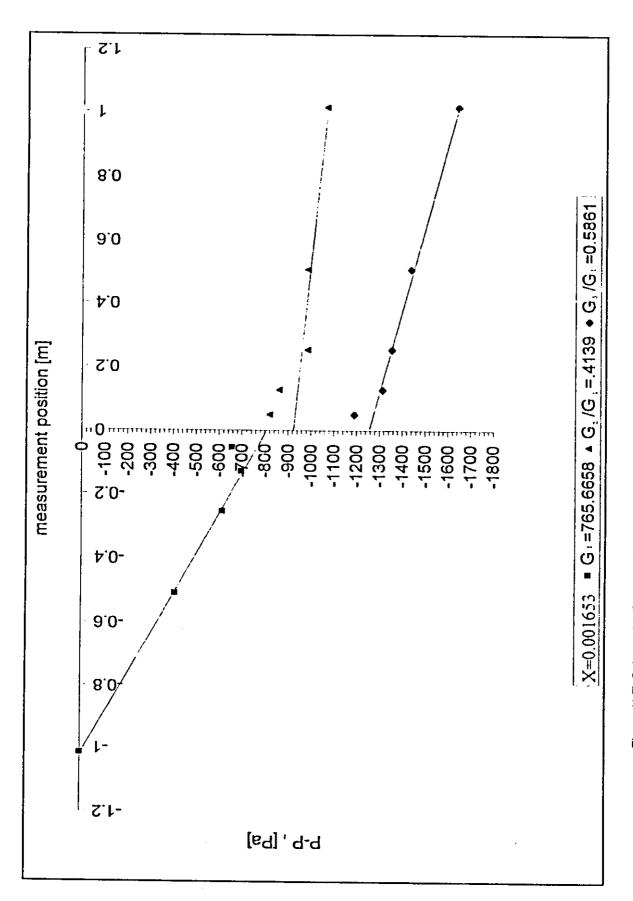


Figure (4.7) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

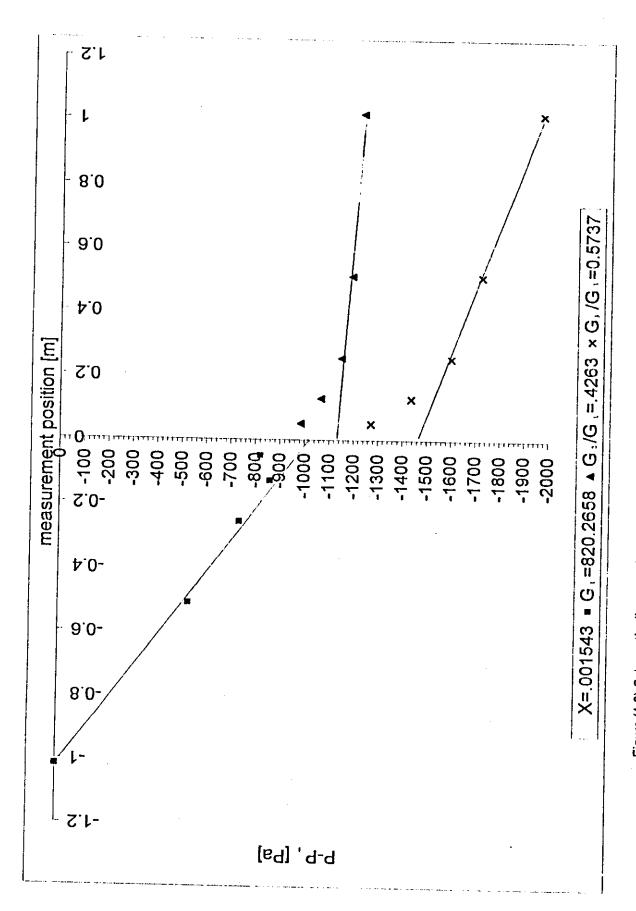


Figure (4.8) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

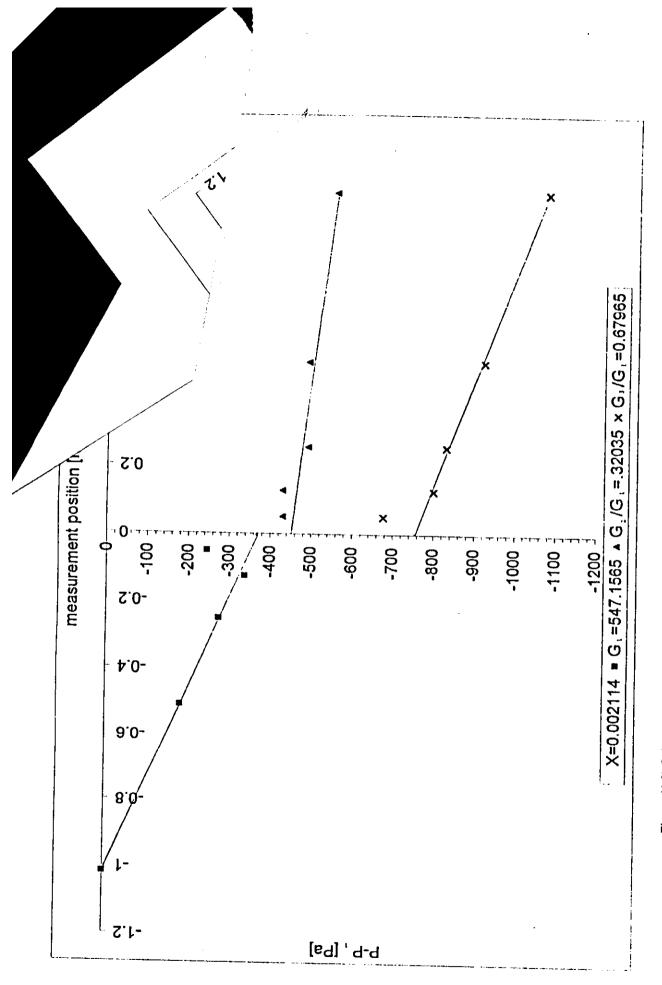


Figure (4.9) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

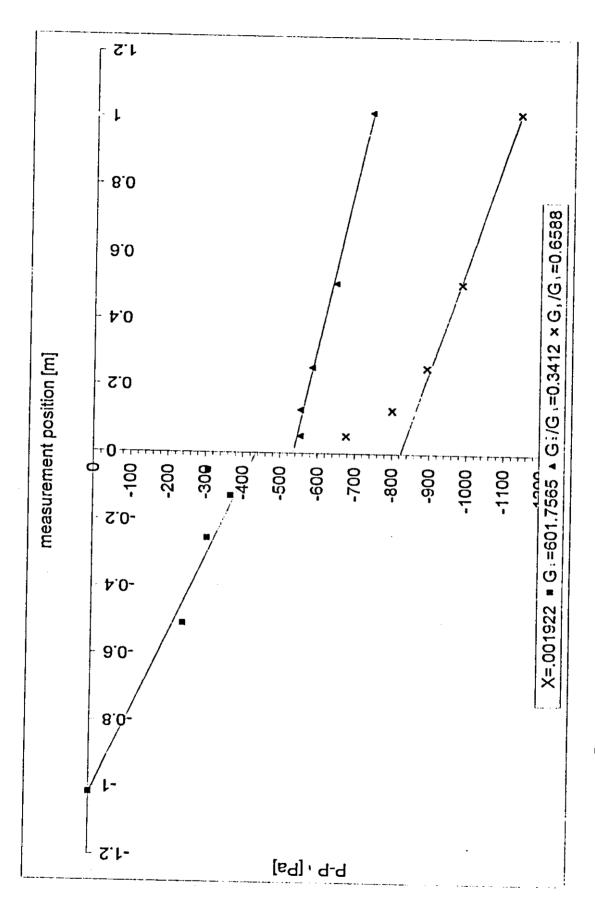


Figure (4.10) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

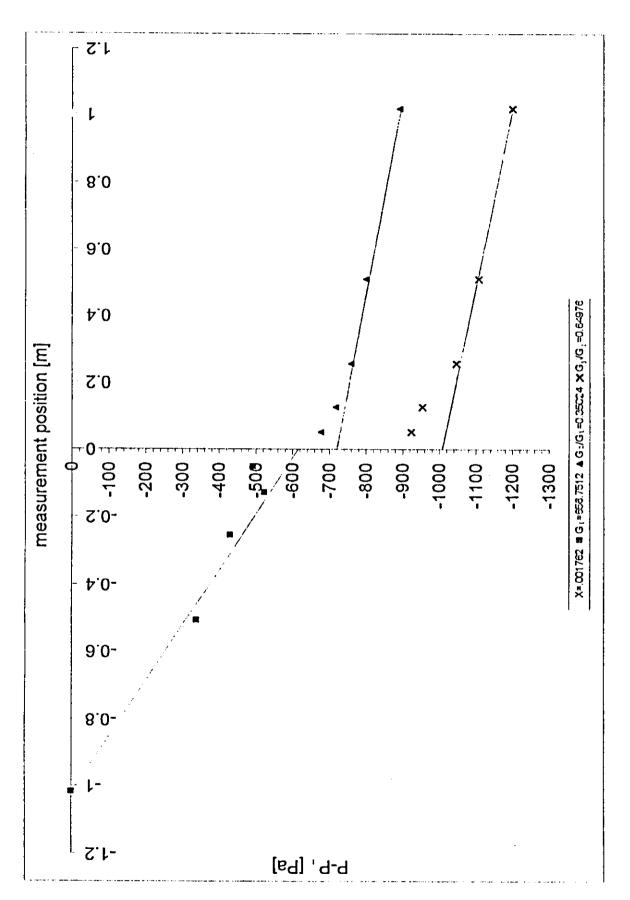


Figure (4.11) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

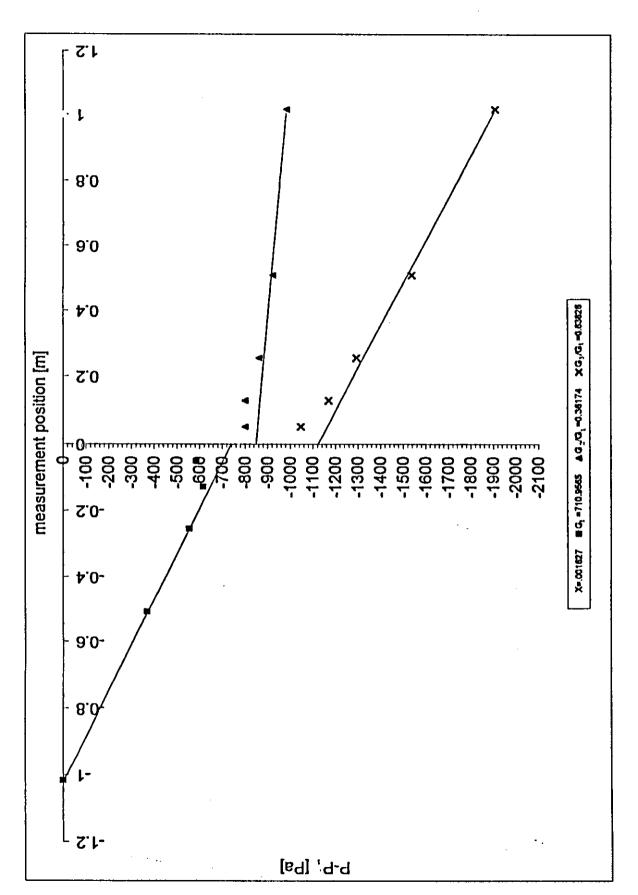


Figure (4.12) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

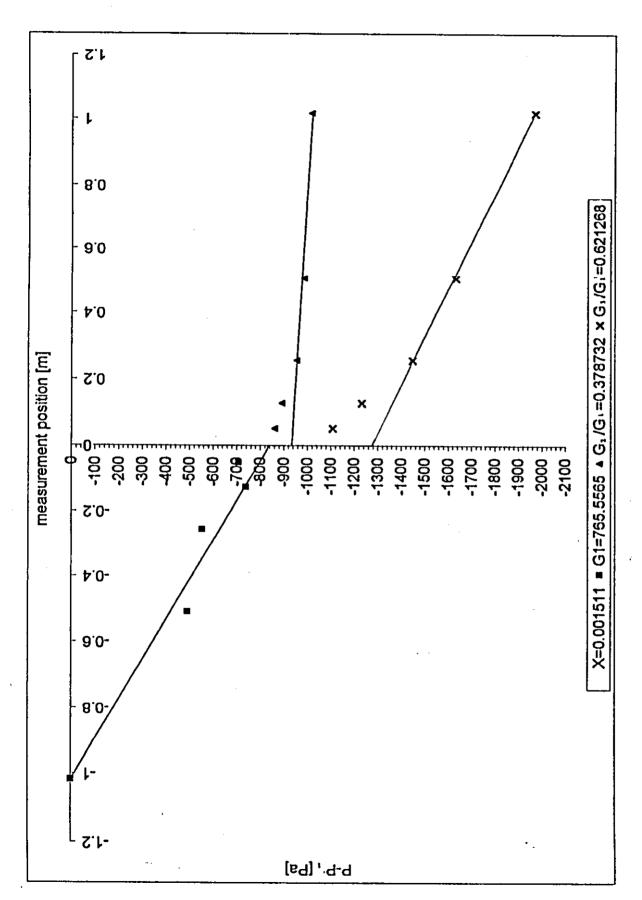


Figure (4.13) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

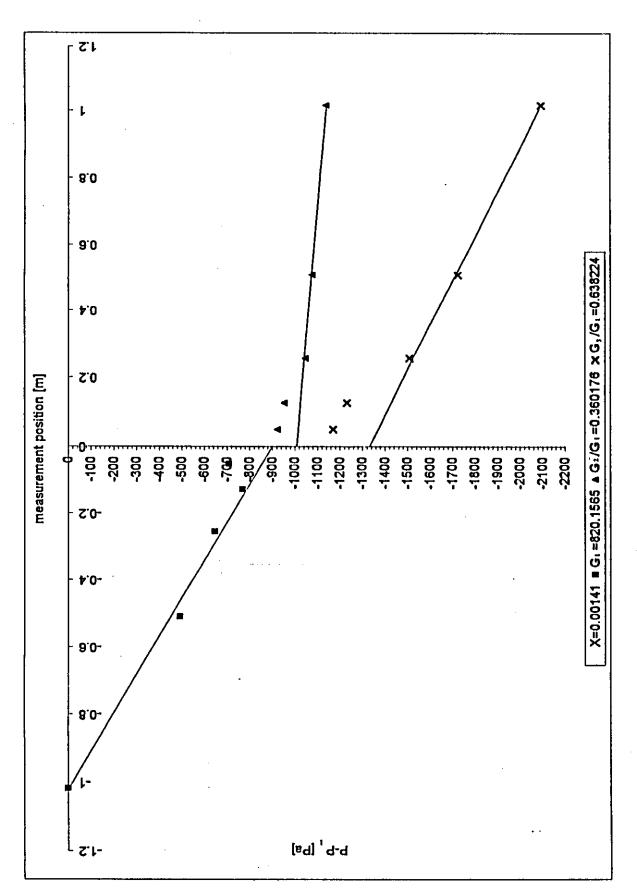


Figure (4.14) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

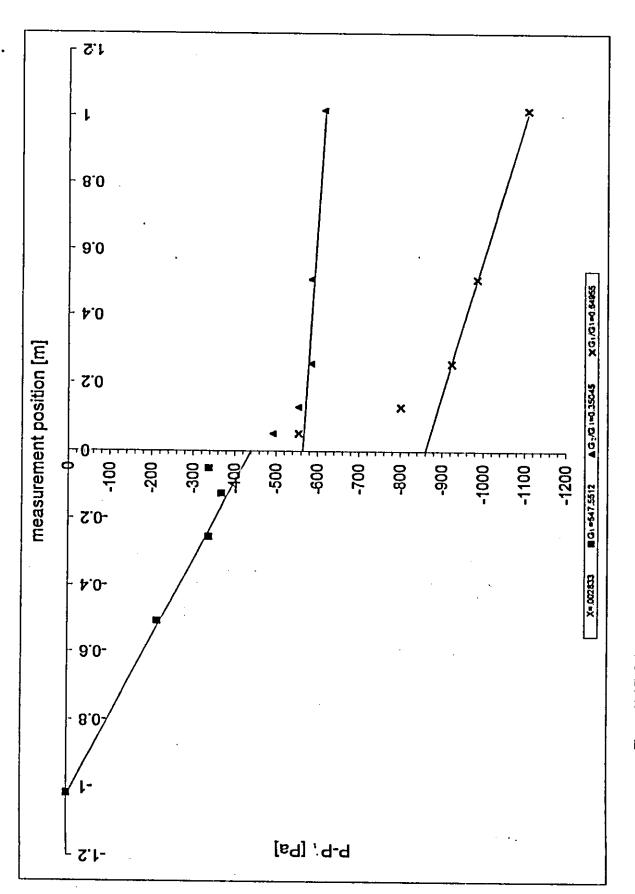


Figure (4.15) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

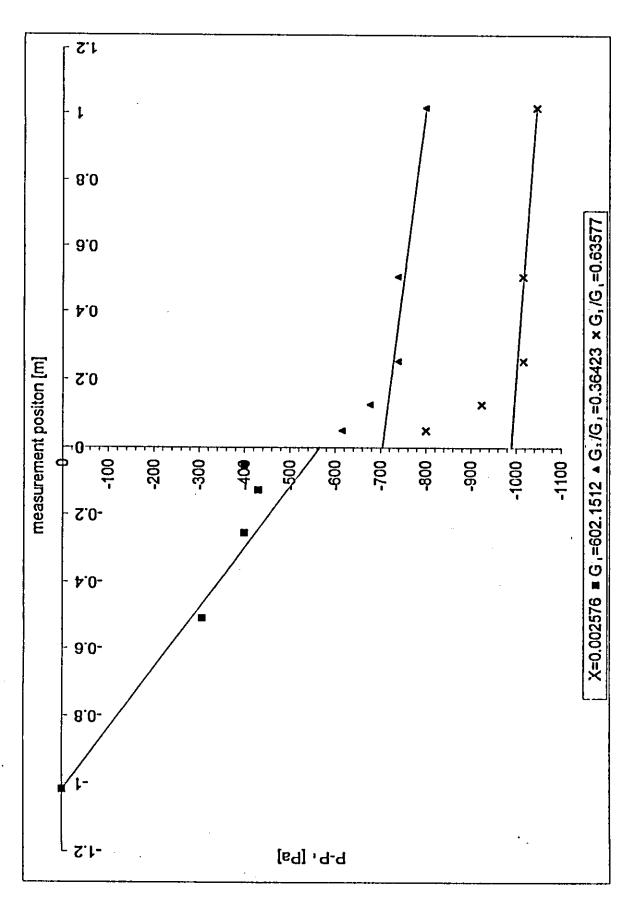


Figure (4.16) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

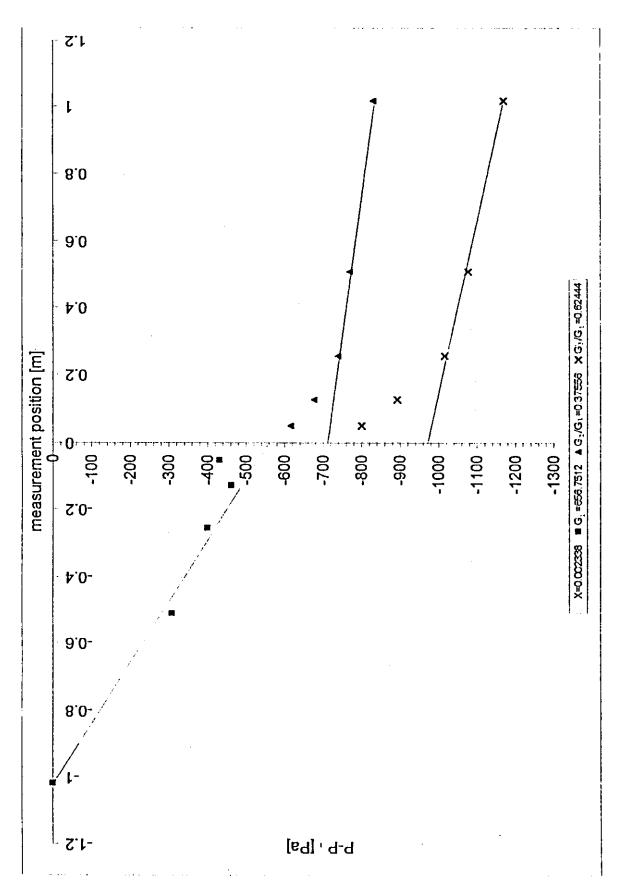


Figure (4.17) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

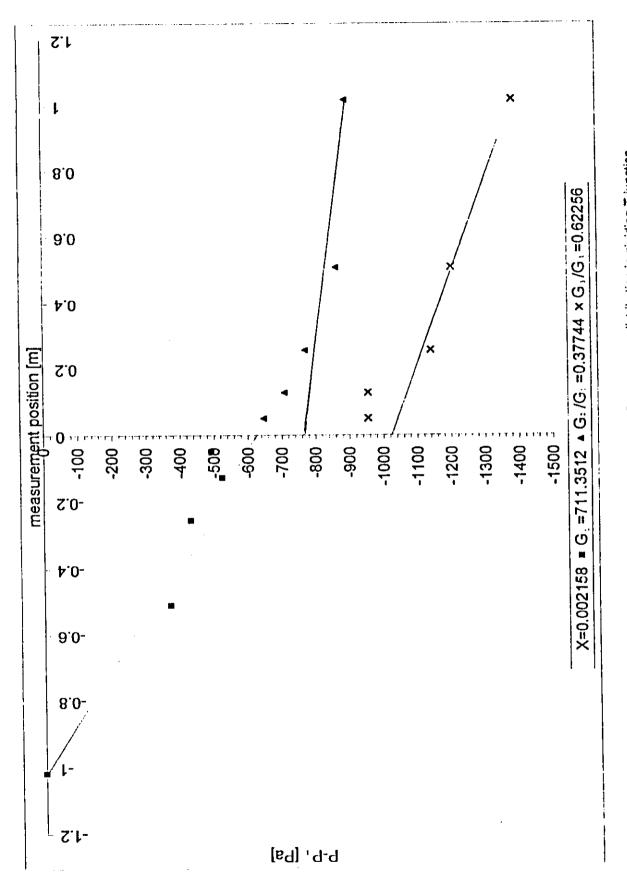


Figure (4.18) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

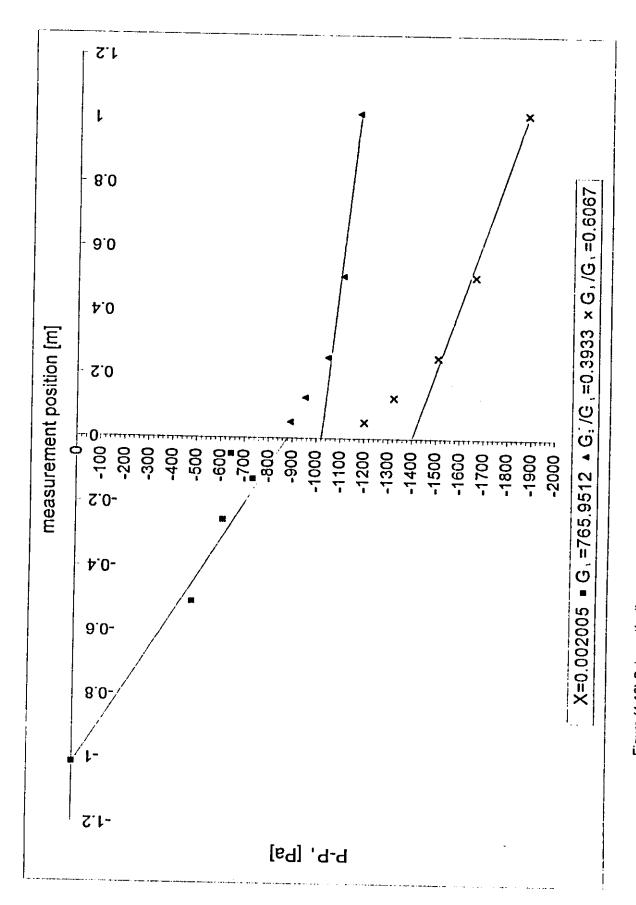


Figure (4.19) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

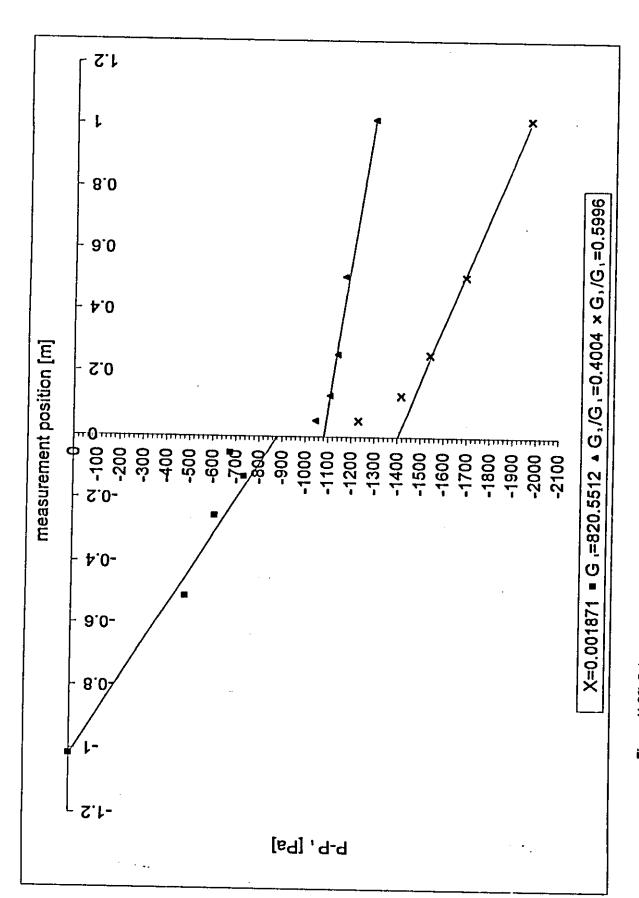


Figure (4.20) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

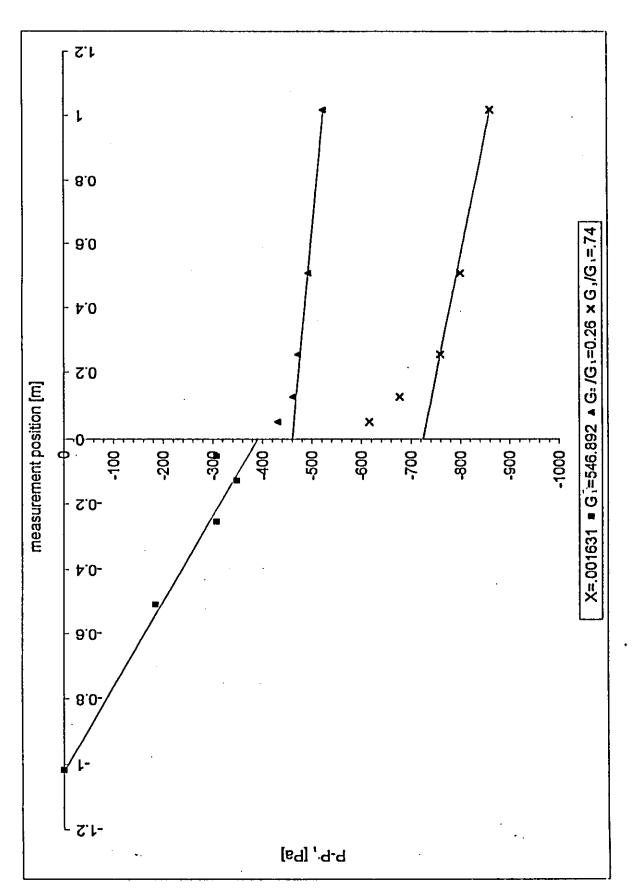


Figure (4.21) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

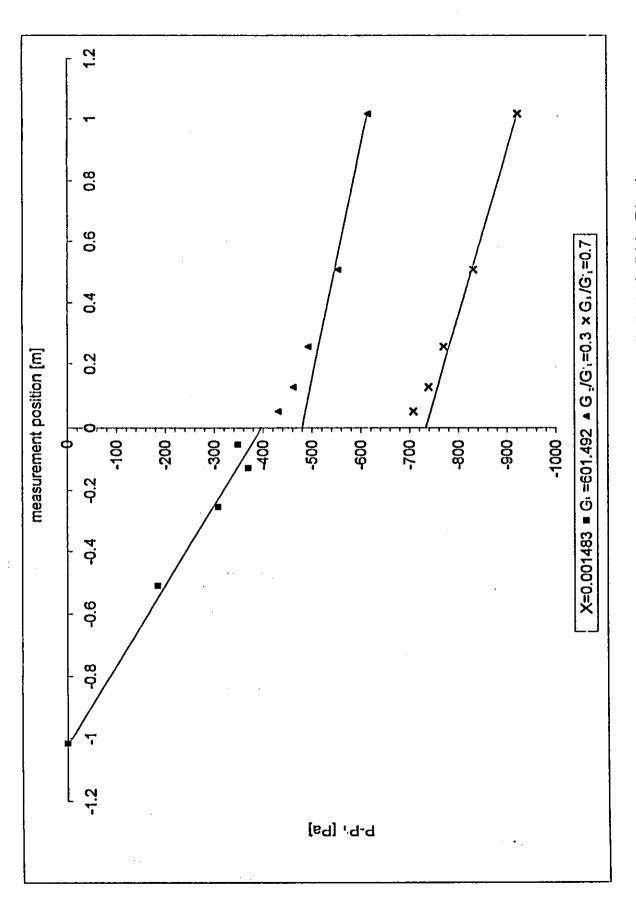


Figure (4.22) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

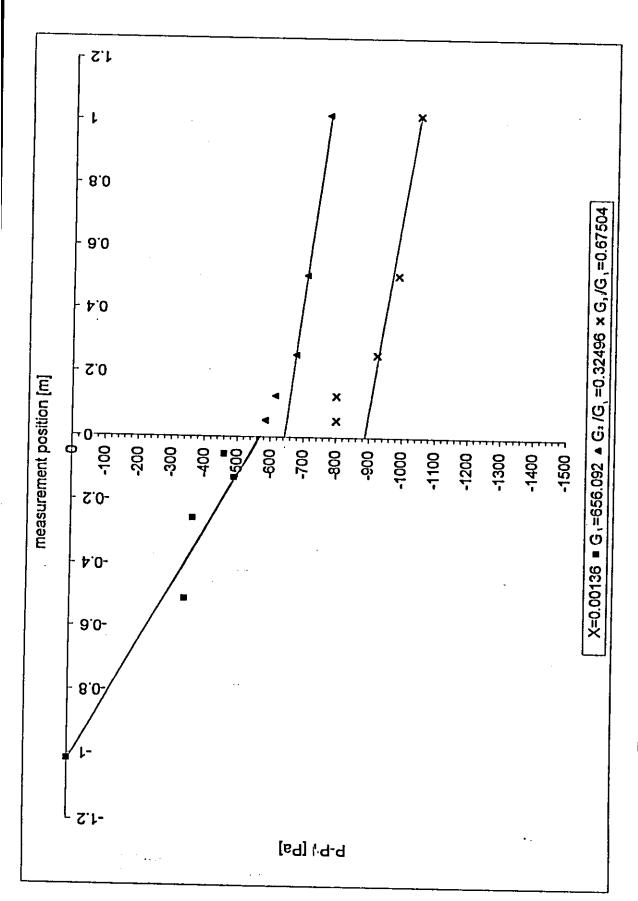


Figure (4.23) Schematic diagram of a typical two-phase flo pressure distribution in dividing T-junction

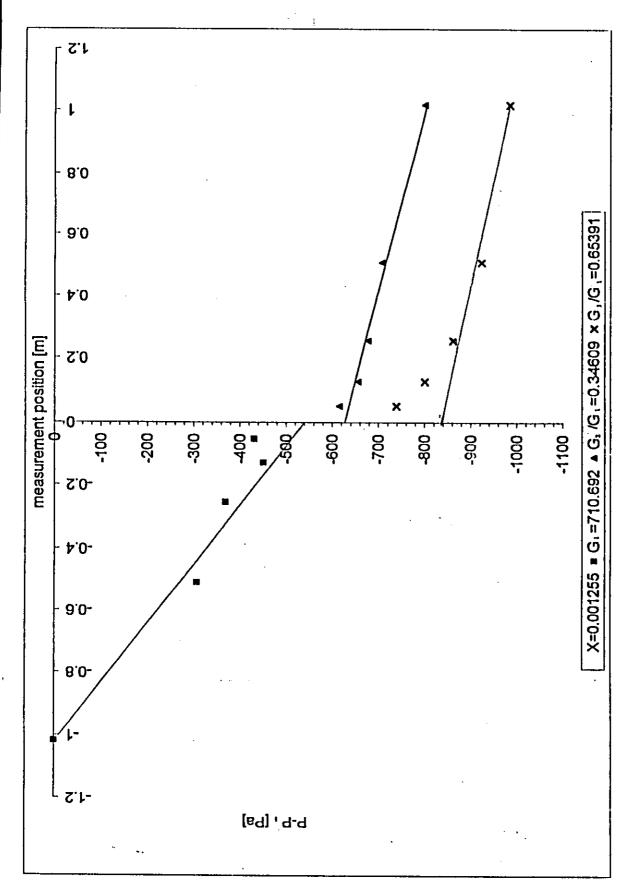


Figure (4.24) Schematic diagram of a typical two-phasa flow pressure distribution in dividing T-junction

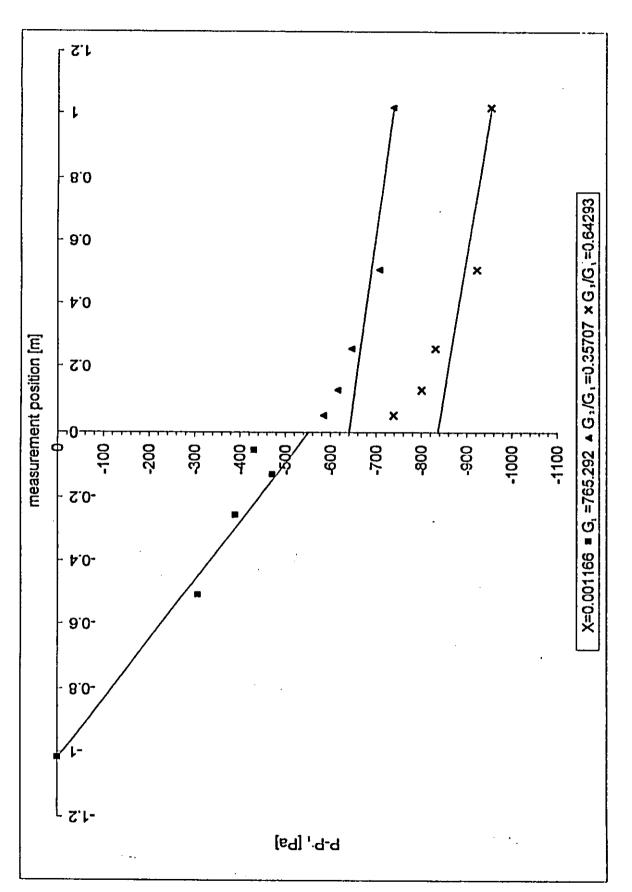


Figure (4.25) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

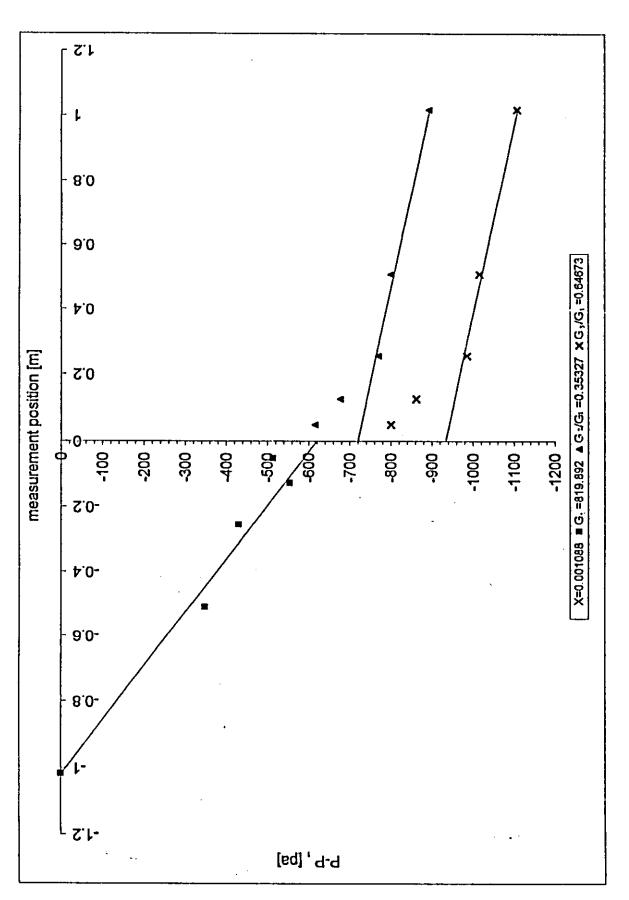


Figure (4.28) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

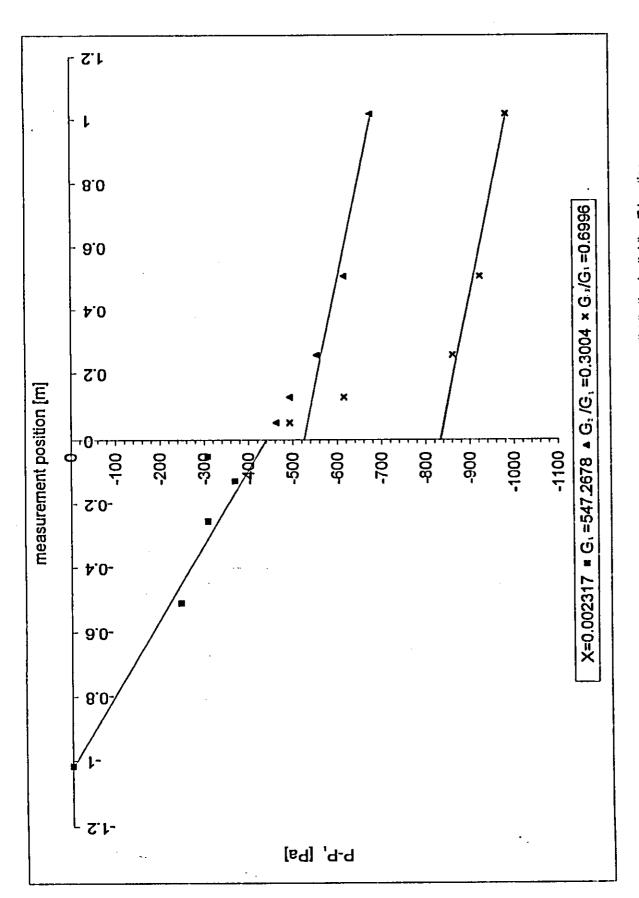


Figure (4.27) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-Junction

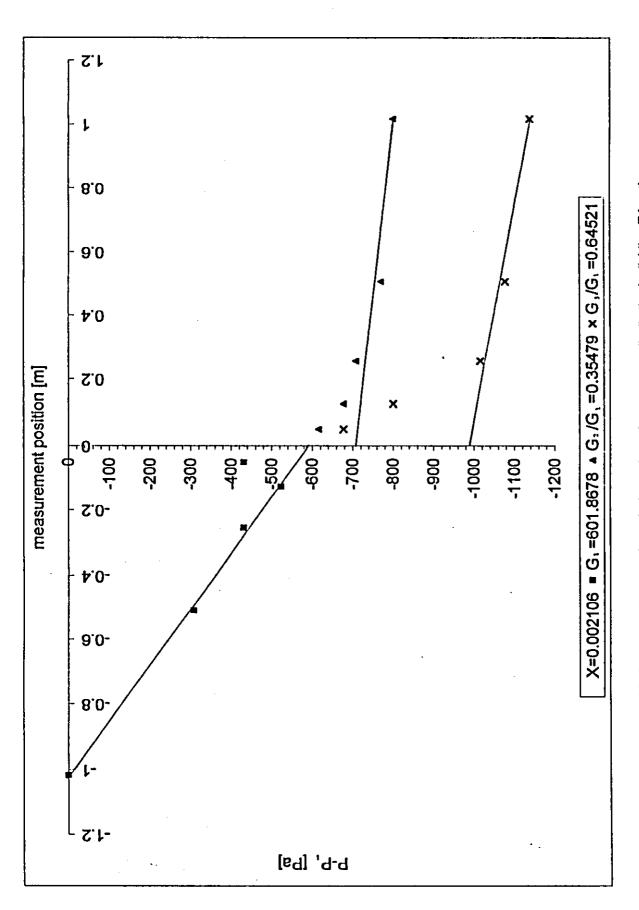


Figure (4.28) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

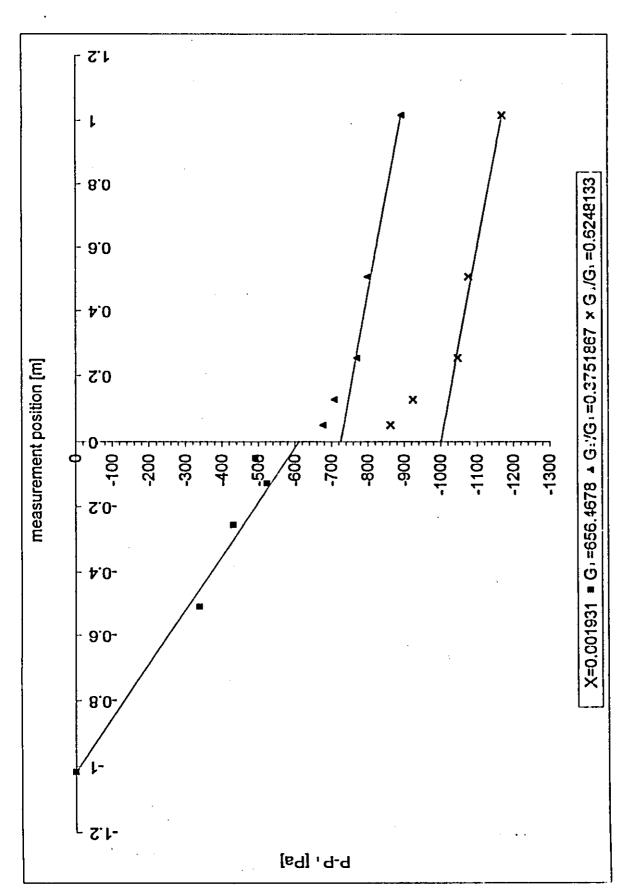


Figure (4.29) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

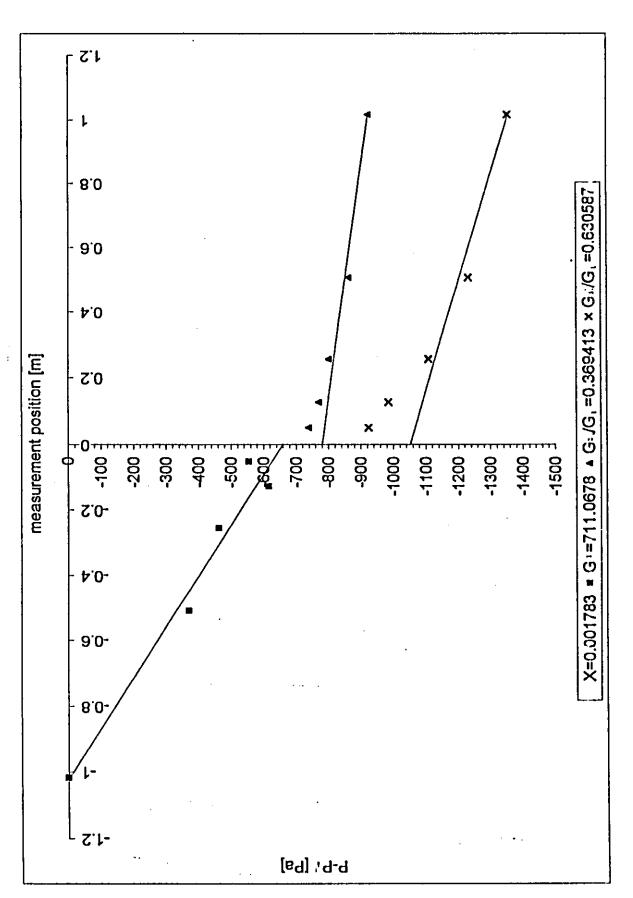


Figure (4.30) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

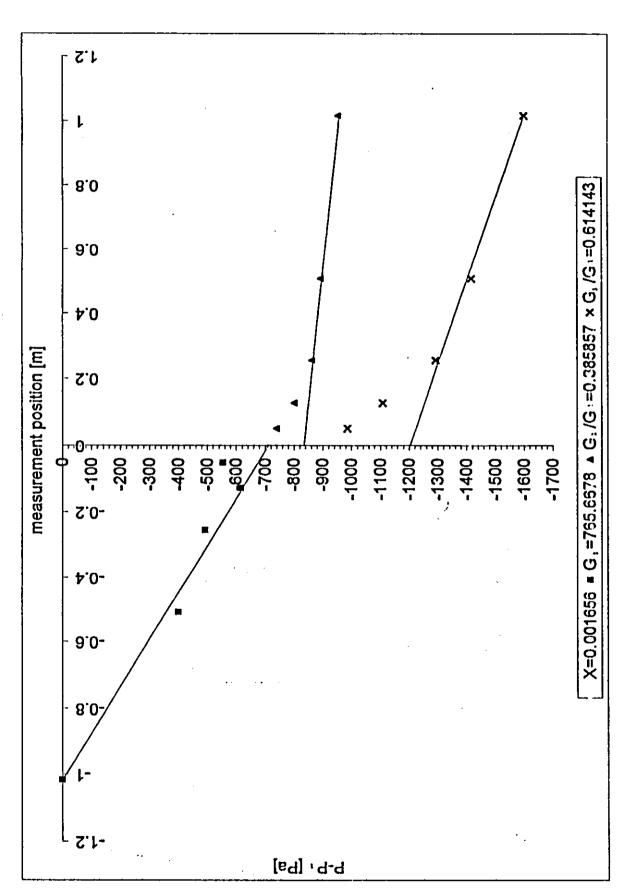


Figure (4.31) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

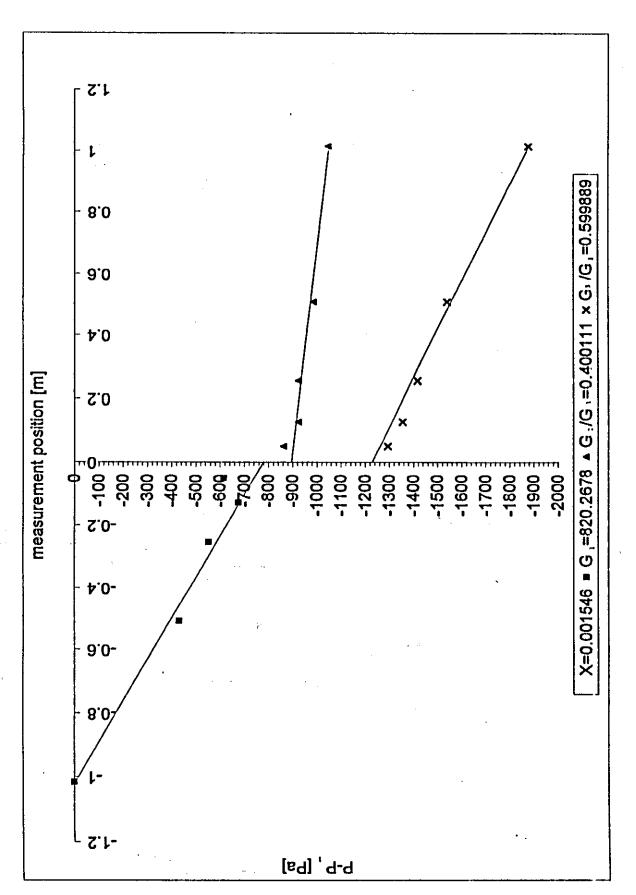


Figure (4.32) Schematic diagram of a typical two-phase flow pressure distribution in dividing T-junction

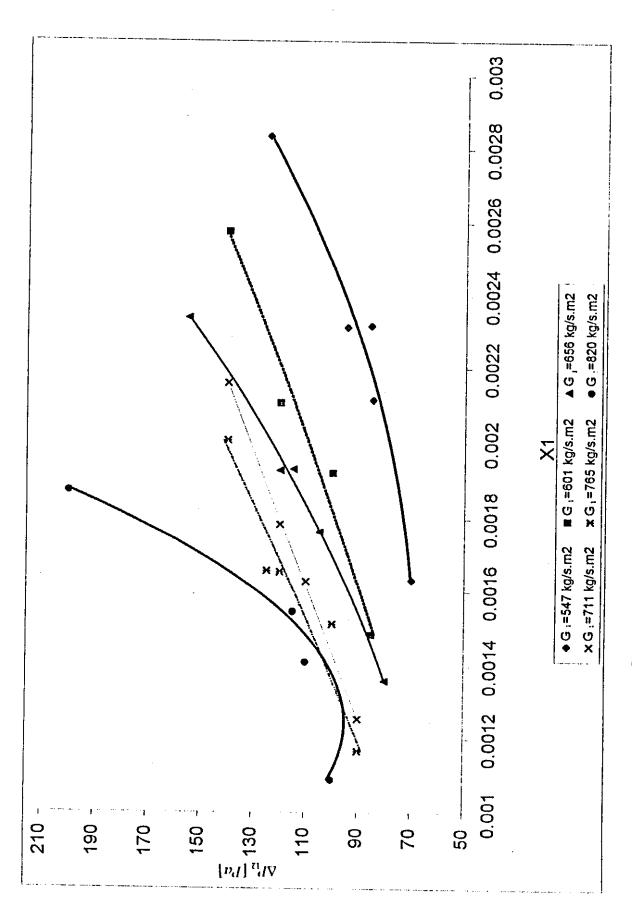


Figure (4.33) Two-phase flow pressure difference against inlet quality

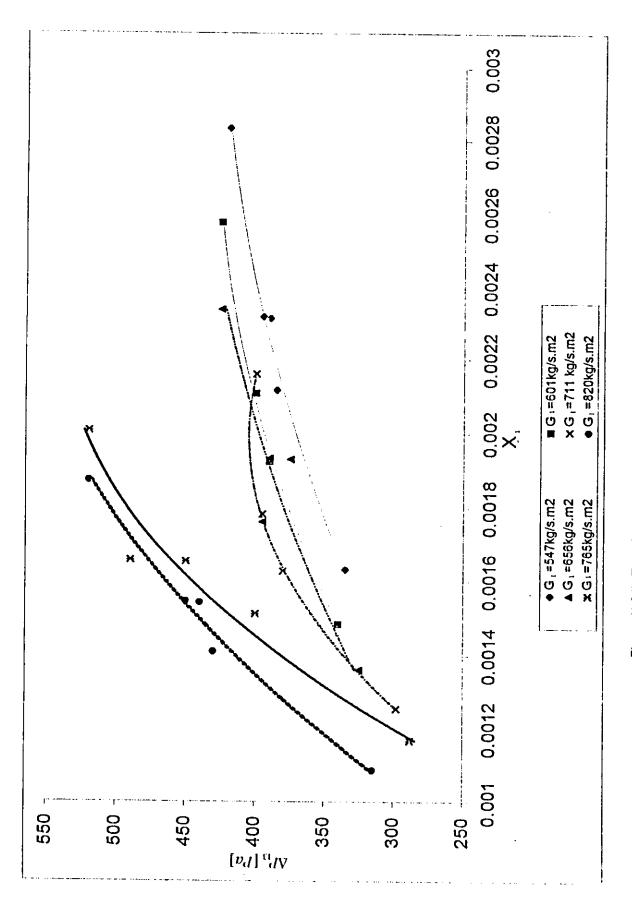


Figure (4.34) Two-phase flow pressure difference against inlet quality

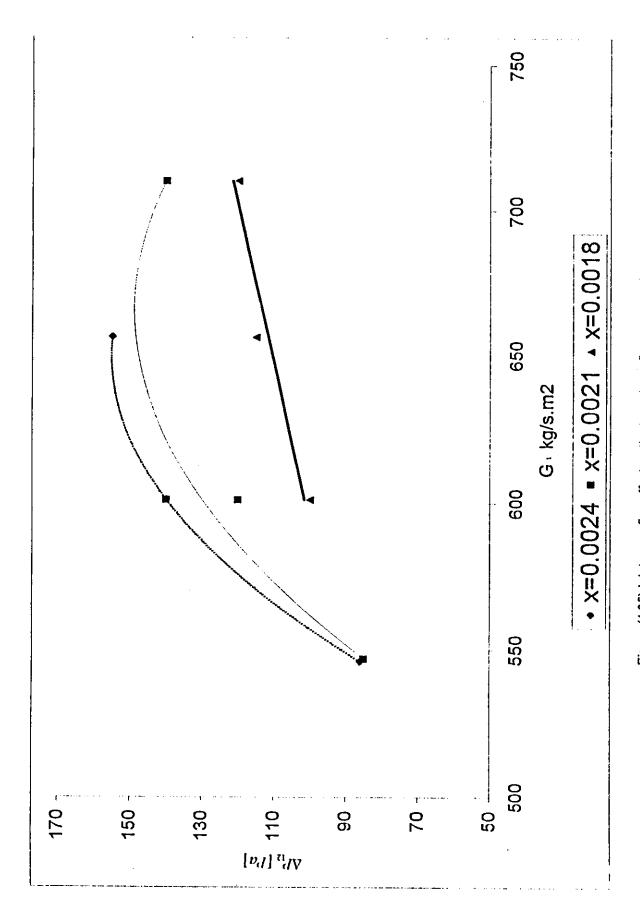


Figure (4.35) Inlet mass flux effect on the two-phase flow pressure drop

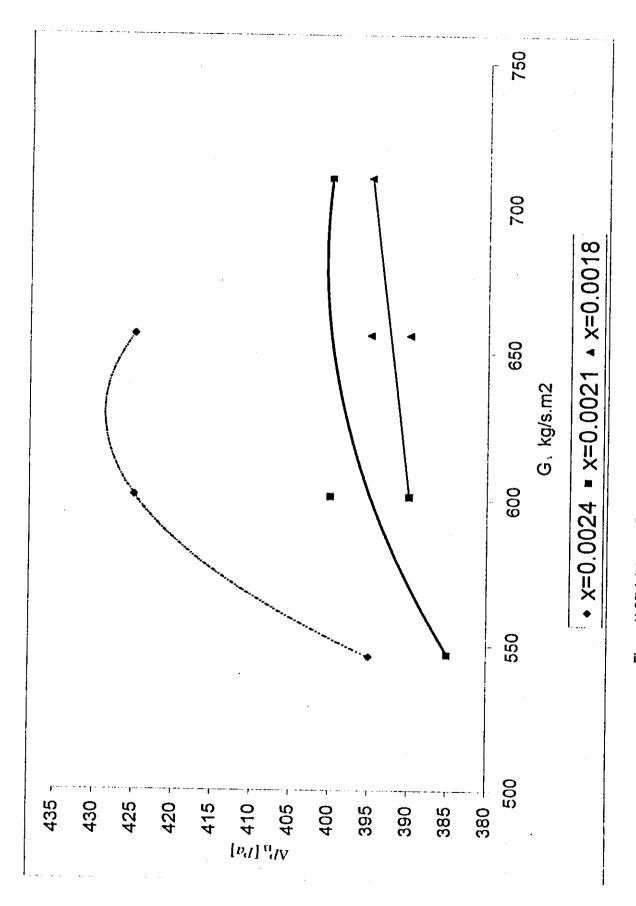


Figure (4.36) Inlet mass flux effect on the two-phase flow pressure drop

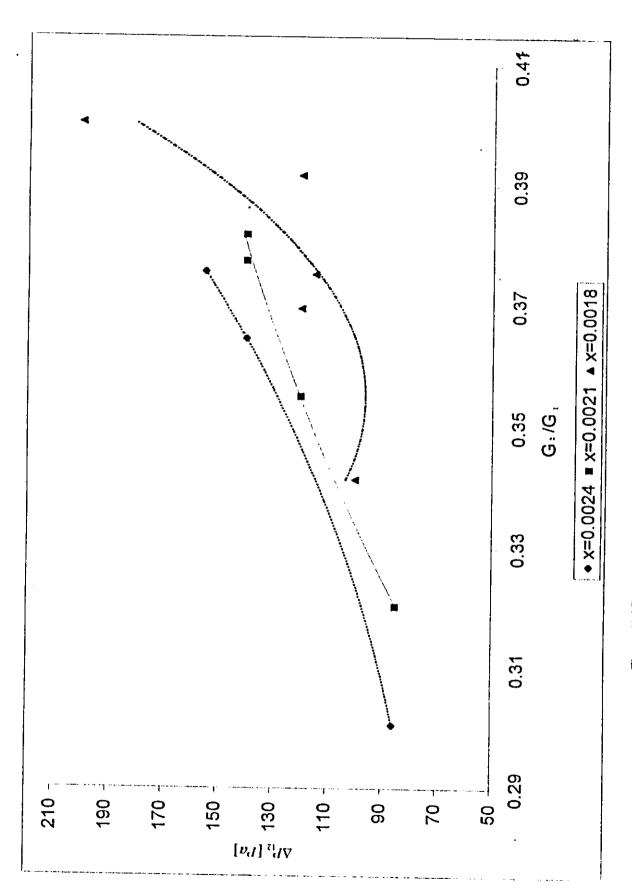


Figure (4.37) Effect of extraction rate on the two-phase flow pressure drop

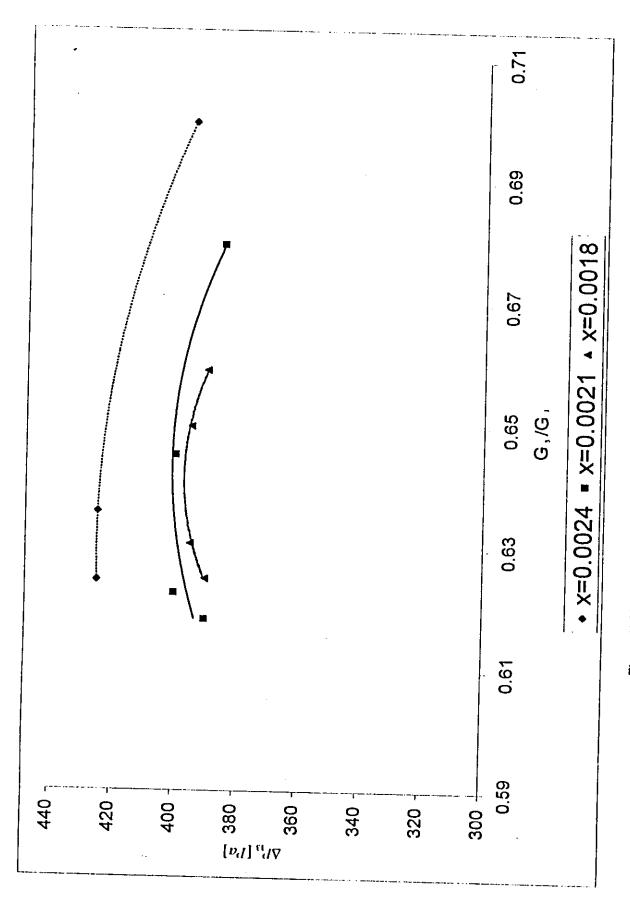


Figure (4.38) Effect of extraction rate on the two-phase flow pressure drop

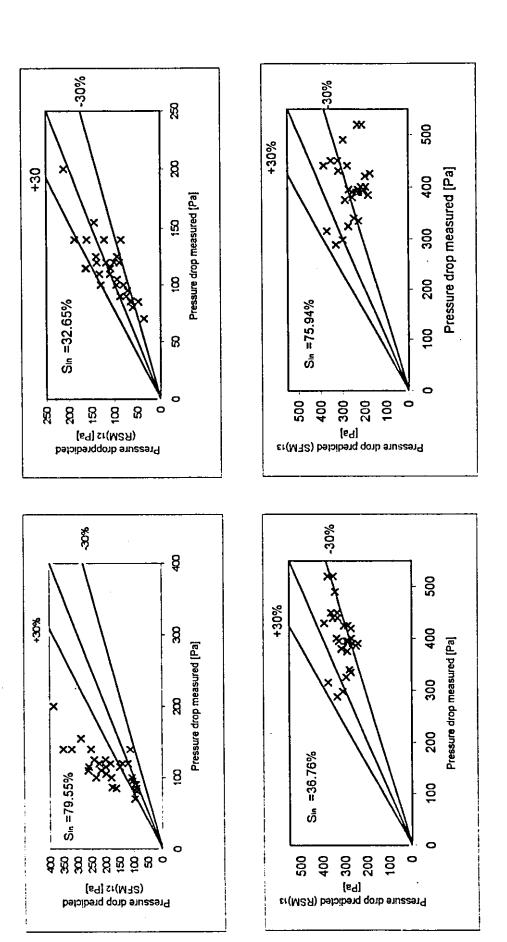


Figure (4.39) Reproductive accuracy of the pressure drop values with (SFM) and (RSM)

4.3 Discussion of the results

4.3.1 Single phase flow:

The average pressures P_1,P_2 at the T-junction inlet and branch-2 at different extraction rates $0.0 \le G_2/G_1 \le 1$ and constant inlet mass flux G_1 =710 kg/s.m² are shown in figure (4.1). These average values were obtained by extrapolating the fully developed pressure gradients at the inlet and branch up to the centerline of the tee, since a constant pressure gradient due to wall friction will be assumed on each leg, where the pressure drop along a straight pipe can be calculated from the following equation:

$$\Delta P = \frac{fl\rho v^2}{2D}$$

Referring to figure (4.2), it is clear that the pressure drop (ΔP) will be increased with increasing G_2/G_1 which is consistence with Bernoulli principle. Here the loss coefficient (K) is assumed to be the same for the two branches due to many reasons such as equal branch diameters, equal branch lengths and similar orientation (horizontal).

The variation of the single phase flow loss coefficient (K) against G_2/G_1 is shown in figure (4.3). It is clear that the trend of the present data agrees well with similar reported results. However, the values of K become much larger

for $G_2/G_1 > 0.6$, this difference can be related to the difference in the nominal diameter (50, 37.6 mm) used in Saba, Buell and Seeger experiments while 25.4mm inner diameter were used in the present study, the difference in the material type since PVC were used, but Saba and Seeger used carbon steel. Buell used commercial Type k-copper and Ballyk used stainless steel. On other hand, there is a difference in the flow direction through the T-junction, where in the present study the flow is divided into two branches while in their study the flow divides into run and branch.

4.3.2 Two-Phase flow:

Two-phase flow pressure difference depends on many parameters such as: the inlet flow regime, inlet quality and extraction rates.

Generally, the pressure drop values associated with two-phase flow are significantly higher than those for single-phase flow. This is cleared from comparing the results obtained for single-phase flow experiments at inlet mass flux G_1 =709.8 kg/s.m², shown in figure (4.1) and those for two-phase flow experiments at G_1 = 710 kg/s.m² shown in figures (4.12, 4.18, 4.24, 4.30), at the same extraction rate, where the two-phase flow pressure difference is about twice of that for single-phase flow. This result is consistent with that reported by Powers, 1992, who states that the actual pressure drop in horizontal construction dewatering pipe containing both water and air can be from 150 to 200 percent of the drop due to water flowing in the pipe alone. The difference between the two flow types referred to the introduction of a second phase which causes a sharp rise in pressure drop due to the following:

1) the cross-sectional area available for the flow of one phase is reduced by the introduction of the second; 2) the flow boundary is no longer a relatively smooth pipe, but is rough, irregular, mobile interface between two-phases; 3)

the introduction of a second phase can cause a very unsteady flow accompanied by an intense turbulence.

Figures (4.4-4.32) obtained at different inlet flow rates, extraction rates and inlet qualities but at the same system pressure ($P_s = 1$ bar) were used to estimate for the average pressures P_1 , P_2 and P_3 at the inlet and branches by extrapolating the fully developed pressure gradients to the centerline of the T-junction, as explained previously in the single phase flow discussion.

One can see from figures (4.4-4.32) that the pressure drop along the inlet test leg is increased with increasing the inlet mass flux, taking in consideration the effect of inlet quality, which is consistent with fundamentals of single-phase flow. The pressure differences ΔP_{12} , ΔP_{13} were taken from these figures and plotted against the inlet mass flux(G₁) figures (4.35, 4.36)), the inlet quality x₁ figures (4.33, 4.34) and the extraction rates (G₂/G₁)& (G₃/G₁), figures (4.37, 4.38).

The variations of ΔP_{12} , ΔP_{13} with the inlet quality x_1 , at constant inlet mass flux are shown in figures (4.33, 4.34), from which the pressure difference is noticed to be increased with increasing the inlet mass flux (G₁). The difference in the trend shape between figure (4.33) and figure (4.34) were related to the influence of the branch qualities on the pressure drop along the branch legs.

The effect of the inlet mass flux (G_1) on the branch pressure differences $\Delta P_{12}, \Delta P_{13}$ at constant inlet qualities is shown in figures (4.35, 4.36). From these figures, the pressure differences $\Delta P_{12}, \Delta P_{13}$ increases for a given x_1 with increasing G_1 , and also increases at a given G_1 with increasing x_1 .

From the above discussion it is clear that the inlet mass flux (G_1) and the inlet quality x_1 are not enough to describe the two-phase flow pressure differences at a given flow pattern. The extraction rate (G_2/G_1) or (G_3/G_1) are the other significant parameter. Figures (4.37, 4.38) shows that ΔP_{12} , ΔP_{13} increase with increasing x_1 at a given extraction rate, while Figure (4.37) shows that ΔP_{12} increases with increasing (G_2/G_1) which is an expected result. An opposite trend for that can be noticed in figure (4.38). But, actually, they are the same, at which for high values of (G_3/G_1) , the trends fall down due to low inlet mass flux (G_1) which consequently reduces the pressure difference value.

From obtained results it is clear that the pressure difference values at branch-3 (ΔP_{13}) is larger than that at branch-2 (ΔP_{12}) . These differences can be related to the high extraction rate at branch-3 with respect to branch-2. Low extraction rates at branch two related to the existence of the air-water separator, which cause a back pressure and then preference of the flow to branch-3.

The statistical mean predictive accuracy of the measured pressure difference values with the separated flow model and Riemann & Seeger model are plotted in the parity plot as shown in figure (4.39), from obtained results of the logarithmic standard deviation and the scatter of the results, Reimann & Seeger model shows better predictions of the present data than that of the separated flow model.

CHAPTER 5

CONCLUSIONS AND RECOMMENDATIONS

5.1 Conclusions

In this work, a two-phase flow test loop was designed, constructed. Experimental investigations for slug flow regime with variable inlet mass fluxes, extraction rates and qualities were carried out. The following conclusions can be stated:

- 1. The experimental set up designed in this work was sufficient to study the slug flow characteristics and two-phase flow in general.
- 2. The obtained results for the pressure differences associated with two-phase flow in a horizontal orientation was found to be twice that of single-phase flow.
- 3. Experimental results showed that the pressure difference is a function of inlet mass flux (G_1) , the inlet quality (x_1) , and the extraction rates (G_2/G_1) and G_3/G_1 .
- 4. There is a strong interdependence between the separation phenomena and the pressure distribution.
- Results obtained were better fitted with Reimann and Seeger model than that of separated flow model, using Reimann and Seeger model.

test loop that constructed through this study is sed for further studies as:

Tecomilow pattern on the pressure difference phenomenon.

distribution phenomena on the T-junction with horizontal and branches orientations.

studying the effect of diameter ratio (D_{branch}/D_{inlet}) on the phase distribution and pressure difference phenomena.

- 4. Studying the phase distribution and pressure difference phenomena at different branches orientation(horizontal, vertical upward and down ward), different types of fittings (elbows, bends, values, ...), different types of fluids as a working fluid, using another material types such as stainless steel, copper,
- 5. Studying the phase distribution and pressure drop phenomena using two inlets and one outlet.

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Table A.1 Measured data for single-phase flow test at T_1 =25.2 °C Inlet water volume flow rate ($V = 1300\ell/hr$)

300	Hg)		Γ	2			
$\begin{vmatrix} \dot{V}_2 = 1300 \\ \ell/hr \end{vmatrix}$	H ₂ (mm Hg)		7	7.25	7.5	∞	0
$\begin{vmatrix} \dot{V_2} = 1100 \\ \ell/hr \end{vmatrix}$	_ <u>=</u>		6.5	٦	7.5	7.75	× ×
$\begin{vmatrix} \dot{V}_2 = 970 \\ \ell/hr \end{vmatrix}$	H ₂ (mm Hg)		6.5	7	7.5	7.75	8.5
$\dot{V_2} = 770$ ℓ/hr	H ₂ (mm Hg)		5.5	5.75	9	6.75	7.25
$\dot{V}_2 = 500$ ℓ/hr	H ₂ (mm Hg)		4	4.25	4.5	4.75	5
$\dot{V_2} = 340$ ℓ/hr	H ₂ (mm Hg)		3	3.25	3.5	3.75	7
$\dot{V}_2 = 0.0$ ℓ/hr	$ m H_2$ (mm Hg)		3.5	4	4.25	4.5	5
	Distance at	orancnz (mm)	0.0508	0.127	0.254	0.508	1.016
•	H ₁ (mmHg)		1.5	2.75	3.5	3.75	3.5
:	Distance at inlet	(mm)	-1.016	-0.508	-0.254	-0.127	-0.0508

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Table A.2.1 Measured data for average pressure heads of two-phase flow test at $P_s = 1$ bar

$\dot{V}_{1w} = 1500\ell/hr$	$\dot{V}_{2w} = 640l/hr$ T ₁ = 28.8 °C	Manometer head (mm Hg)	4	8.33	10	11	10.67	12	12.67	13.33	13.67	14	14.33	15.67	17	18	20
$\dot{V}_{1w} = 1400\ell/hr$	$\dot{V}_{2*} = 580\ell/hr$ $T_1 = 28.5 ^{\circ}C$	Manometer head (mm Hg)	4.33	7.67	9.33	10	6.67	11	11.33	12.33	12.33	13	14	15	15.33	16	17.67
$\dot{V}_{1\omega} = 1200\ell/\hbar r$	$\dot{V}_{2,w} = 470\ell/hr$ T ₁ =28.3 °C	Manometer head (mm Hg)	4	7	7.67	8.67	8.33	29.6	10	10.5	10.67	11	10.33	11.67	12.67	13	13.67
$\dot{V}_{1w} = 1100\ell/hr$	$\dot{V}_{2w} = 420\ell/hr$ T ₁ = 28.1 °C	Manometer head	4	9	29'9	79.7	7.33	8.67	6	9.33	9.33	01	10	11	11.33	11.67	12.33
$\dot{V}_{1w} = 1000\ell/hr$	$\dot{V}_{2,*} = 370\ell/hr$ T, =26 6 °C	Manometer head	3.83	9	6.67	7	6.67	7.33	8	8 67	6	9.33	6	10.67	11.33	11.67	12.33
Distance	(m)		-1 016	-0.508	-0.254	-0.127	-0.0508	0.0508	0.127	0.254	0 508	1.016	0.0508	0 127	0.254	0.508	1 016

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-	$V_{1w} = 1500\ell/hr$	$V_{2w} = 540\ell/hr$ $T_1 = 24.8 ^{\circ}C$	Manometer head	(mm Hg)	4	8	9.25	10.25	9.75		11.5	11.75	12.5	12.75	13.25		13.5	14	16.25	18	21	
bar	$\dot{V}_{1,w} = 1400 \ell / hr$	$\dot{V}_{2w} = 530\ell/hr$ T ₁ = 24.8 °C	Manometer head	(mm Hg)	4	8	8.5	10	9.75		11	11.25	11.75	12	12.25		13	14	15.75	17.25	20	
$\frac{1}{3}$ $\frac{1}$	$\dot{V}_{1w} = 1300\ell/hr$	$V_{2v} = 470\ell/hr$	Manometer head	(mm Hg)	3.5	6.5	€ ×	2 8	8.25	27.0	10	01	10.5	11	11 5	11:3	12	13	14	16	10	
neads of two-phase	$\dot{V}_{1,*} = 1200\ell/hr$	$\dot{V}_{2w} = 420\ell/hr$	11=24.7	Manometer nead	2.5	2.5	67.0	10.00	27.7	6.7		600	9.53	9.67	01	10.75		11 25	11.23	3 (1	20.71	15.23
r average pressure l	V = 1100ℓ/hr	$\dot{V}_{2w} = 375\ell/\hbar r$	$T_1 = 24.7 ^{\circ}C$	Manometer head	(mm Hg)	3.5	5.5	9	6.5	9		8	8	8.25	8.75	9.5		6	10	10.75	11.5	12.75
) Moonwood data fo	INTERSMENT DAMES IN	$V_{1w} = 100001 m$ $\dot{V}_{2w} = 3200 l hr$	$T_1 = 24.7 ^{\circ}C$	Manometer head	(mm Hg)	3.5	5	5.75	6.25	5.5		7	7	7.5	7.5	8		6	10	10.25	11	12.25
c -	Table A.2	(II)				-1.016	-0.508	-0.254	-0.127	-0.0508		0.0508	0.127	0.254	0.508	1.016		0.0508	0.127	0.254	0.508	1.016

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	$\dot{V}_{1w} = 1500\ell/\hbar r$	$\dot{V}_{2*} = 600\ell/hr$ $T_1 = 27.7 ^{\circ}\text{C}$	Manometer head	(mm Hg)	4	8	6	10	9.5		12.5	12:3	CI	13.25	13.5	14.5		14	15.5	17.5	10:5	C/./I	70	
	$\dot{V}_{1\nu} = 1400\ell/hr$	$\dot{V}_{2v} = 550\ell/hr$ T ₁ =27.6 °C	Manometer head	(mm Hg)	3.75	7.75	8.75	9.75	6		11	11,	11.5	12.25	12.75	12.05	72.01	12.5	12.5	C.4.1	Ιρ	17.25	19	
w test at $P_s = 1$ bar	$\dot{V}_1 = 1300\ell/hr$	$V_{2w} = 490\ell/hr$	Manameter head	(mm Hg)	4.25	725	7775	8 \$	20.0	0.25		9.5	10	10.5	11.05	11.20	2.11		71	12	10.5	14	15.5	10.0
re heads of two-phase flow test at $P_s = 1$ bar	$\dot{V}_{1w} = 1200\ell/hr$	$\dot{V}_{2w} = 450\ell/hr$	11-21 C	Manometer neau	4	2,7	20.0	27.7	(1.7)	7.5		6	9.5	10	01	10.25	10.75	1	10.5	11.25	12.25	12.75	12.6	13.3
erage pressure head	$\dot{V}_{i} = 1100\ell/hr$	$\dot{V}_{2w} = 400\ell/hr$	11 = 20.8°C	Manometer head	(gu mm)	4	6.5	27./	7.5	7.25		6	0 \$		ΙO	10	10.5		10.5	11.5	12.25	12.25	12.27	12.5
Commend data for asy	Table A.2.3 Infeasured data for avoiding $V_{i} = 1100\ell$	$\dot{V}_{2w} = 350\ell/hr$	$T_1 = 26.3 ^{\circ}\text{C}$	Manometer head	(mm Hg)	4	5.75	6.75	7	6.75		×	2 0	6.0	8.75	8.75	6		8.5	10.5	115	11.2	71	13
, c c c c c c c c c c c c c c c c c c c	Table A.2.5 IVI	(m)				-1.016	-0.508	-0.254	-0.127	-0.0508		00500	0.0000	0.127	0.254	0.508	1.016		0.0508	0.000	0.127	0.234	0.508	1.016

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Table A.2.4 Measured data for average pressure heads of two-phase flow test at $P_s = 1$ bar

Distance (m)	$\dot{V}_{1w} = 1000\ell/hr$	$\dot{V}_{1w} = 1100\ell/hr$	$\dot{V}_{1w} = 1200\ell/hr$	$\dot{V}_1 = 1300\ell/hr$	$\dot{V}_{1*} = 1400\ell/hr$	$\dot{V}_{1*} = 1500\ell/hr$
,	$\dot{V}_{2w} = 260\ell/hr$ T ₁ =25 °C	$\dot{V}_{2w} = 330\ell/hr$ T ₁ = 25.1 °C	$V_{2*} = 390\ell/hr T_1$ =25.1 °C	$\dot{V}_{2w} = 450\ell/hr$ $T_1 = 25.1 {}^{\circ}C$	$\dot{V}_{2w} = 500\ell/hr$ T ₁ = 25.2 °C	$\dot{V}_{2*} = 530\ell/hr$ T, =25.2 °C
	Manometer head	Manometer head	Manometer head	Manometer head	Manometer head	Manometer head
	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)
-1.016	3.5	3.5	3.5	4	4.5	4.5
-0.508	5	5	6.33	6.5	7	7.33
-0.254	9	9	6.5	7	7.67	8
-0.127	6.33	6.5	7.5	7.67	8.33	6
-0.0508	9	6.33	7.25	7.5	8	8.67
0.0508	7	7	8.25	6	9.25	9.5
0.127	7.25	7.25	8.5	9.33	9.5	10
0.254	7.33	7.5	6	9.5	9.75	10.75
0.508	7.5	8	9.25	9.75	10.25	11
1.016	7.75	8.5	9.75	10.5	10.5	11.75
0.0508	8.5	9.25	10	10	10.5	
0.127	6	9.5	10	10.5	11	11.5
0.254	29.6	9.75	11	11	11.25	12.5
0.508	10	10.25	11.5	11.5	12	12.75
1.016	10.5	11	12	12	12.25	13.5

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Distance (m)	$V_{1\nu} = 1000\ell/hr$	$\dot{V}_{1\nu} = 1100\ell/hr$	$\dot{V}_{1w} = 1200\ell/hr$	$\dot{V}_1 = 1300\ell/hr$	$\dot{V}_{1w} = 1400\ell/hr$	$\dot{V}_{1w} = 1500\ell/hr$
	$\dot{V}_{zw} = 300\ell/hr$	$\dot{V}_{2w} = 390\ell/hr$	$\dot{V}_{2\nu} = 450\ell/hr$	$\dot{V}_{2w} = 480\ell/hr$	$\dot{V}_{2,\nu} = 540\ell/hr$	$\dot{V}_{2w} = 600\ell/hr$
	T ₁ =24.3 °C	T ₁ =24.4 °C	$T_1 = 24.4 ^{\circ}C$	$T_1 = 24.5 ^{\circ}C$	$T_1 = 24.5 ^{\circ}C$	T ₁ =24.5 °C
	Manometer head	Manometer head	Manometer head	Manometer head	Manometer head	Manometer head
	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)	(mm Hg)
-1.016	3.5	3.5	4	4	4.5	4.5
-0.508	5.5	9	6.75	7	7.75	∞
-0.254	9	7	7.5	7.75	8.5	6
-0.127	6.5	7.75	8.25	6	9.5	10
-0.0508	9	7	8	8.5	6	9.5
1						
0.0508	7.25	8.5	9.5	10	10.5	11.5
0.127	7.5	6	9.75	10.25	11	12
0.254	8	9.25	10.25	10.5	11.5	12
0.508	8.5	9.75	10.5	11	11.75	12.5
1.016	6	10	11.25	11.5	12.25	13
9030	i t					
0.0508	C./	ý	11	11.5	12.5	15
0.127	8.5	10	11.5	12	13.5	15.5
0.254	10.5	11.75	12.5	13	15	16
0.508	11	12.25	12.75	14	16	17
1 016	115	12.75	12 5	31	176	10.10

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Table A.3 Measured data for air mass flow rate through an orifice and pitot tube

Manometer liquid density	Kg/m′	787	970	787						970	787
Pipe diameter	(mm)	8	9	9	-					8	8
Pitot tube manometer Pipe diameter	dynamic head h _d (mm)		4.5	11	15	16	16	16	16		2
T_i	(Ç)	42	42	42	42	48	48	48	48	44	41
P ₃	(bar abs.)									_	
Orifice differential head	∆P (mm wg)	r	25	4.5						15	

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Table B.1 Results obtained for single-phase flow system pressure difference, T_1 =25 °C Inlet mass flux $(G_1) = 709.8 \text{ kg/s}$

G ₂ =709.8 Kg/s.m ²		$(\Delta P_{\omega}),$	Pa			861 781	10/100	072.3389	923.3368	984 8076	1108 004	1100.001
G ₂ =600.6 Kg/s.m ²		$(\Delta P_{\pi})_2$	Pa		_	800.2257	861 781	10/1100	923.3368	954,1147	1046 448	21.01.01
G ₂ =529.6 Kg/s.m ²		$(\Delta P_{_{\mathcal{A}}})_{_{2}}$	Pa			800.2252	861 781	2720 000	973.3368	954.1147	1046.448	
G ₂ =420.42 Kg/s.m ²		$(\Delta P_{_{H}})_2$	Pa			677.1137	707.8916	730 6605	730,0093	831.0031	892.5589	
G ₂ =273 Kg/s.m ²		$(\Delta_{\pi}^{O})_{2}$	ਮੂ ਫ਼		2777 007	492.4463	523.2242	554 0001	1700.100	384./8	615.5579	
G ₂ =185.64 Kg/s.m ²	(40)	(\(\alpha''_n\)_2 D2	4		260 2247	702.2347	400.1126	430.8905	161 6691	+	492.4463	
G ₂ =0.0 Kg/s.m ²	┥_	D ₃	3		430 8005	407 4462	492.4403	223.2242	554 0021	615 5570	013.3379	
	Distance	at	branch2	(mm)	0.0508	0.127	0.754	U.234	0.508	1 016	1.010	
	$(\Delta P_{\mu})_{i}$	Pa			184.6674	338 5568	130 8005	730.070	461.6684	430 8905	22.2	
	Distance	at inlet	(mm)		-1.016	-0.508	-0.254	- 67:0	-0.127	-0.0508		

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Table B.2 Results obtained for single-phase flow pressure difference with tap#1, T_1 =25 °C Inlet mass flux (G₁) = 709.8 kg/s

			$G_2=0.0$ Kg/s.m ²	$G_2=185.64$ Kg/s.m ²	$G_2=273$ Kg/s.m ²	$G_2=420.42$ Kg/s.m ²	$G_2=529.6$ Kg/s.m ²	$G_2=600.6$ Kg/s.m ²	$G_2 = 709.8$ Kg/s.m ²
Distance at inlet	$(\Delta P_{ii})_1$ Pa	Distance at	$(\Delta P_u)_2$ Pa	$(\Delta P_{l_1})_2$ Pa	$(\Delta P_{ii})_2$ Pa	$(\Delta P_{11})_2$ Pa	$(\Delta_n^p)_2$ Pa	$(\Delta P_{ii})_2$ Pa	$(\Delta P_{1r})_2$ Pa
(mm)		branch2 (mm)							
-1.016	0.0	0.0508	-246.221	-184.665	-307.776	-492.444	-615.555	-615.555	-677.111
-0.508	-153.887	0.127	-307.776	-215.443	-338.554	-523.222	-677.111	-677.111	-707.889
-0.254	-246.221	0.254	-338.554	-246.221	-369.332	-553,999	-738.667	-738.667	-738.667
-0.127	-273.998	0.508	-369.332	-276.998	-400.11	-646.333	-769.445	-769.445	-800.223
-0.0508	-246.221	1.016	-430.888	-307.776	-430.888	-707.889	-861.778	-861.778	-923.334

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Table B.3 Results obtained for single-phase loss coefficient (k)

G_2/G_1	P ₁₂	K (Present)	K (Ballyk)	K (Buell)	K (Seeger)	G,/G	K (Saba)
	(Pa)	-		,	,	· !	·
0.0	25	1.09885	1.081	1.0	1.0369	0.0	1.18
0.2615	35	1.070008	0.91379	0.856414	0.870172	0.2	1.17
0.3846	55	1.069553	0.884789	0.854145	0.849081	0.4	1.04
0.5923	125	1.143431	0.907998	0.915936	0.896789	9.0	1.17
0.7462	185	1.174676	0.983628	0.995531	0.999604	0.8	1.18
0.8462	245	1.252676	1.05943	1.054272	1.09719	1.0	1.38
1.0	410	1.62114	1.217	1.144	1.2946		

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Table B.4.1 Results obtained for air and water mass flow rates and inlet and outlet qualities

		_	,	_	
X	0.002446	0.002267	0.002112	0.00188	0.001793
X ₂	0.002313 0.00208617	0.00183827	0.00164303	0.00133184	0.001543 0.00120713
X	0.002313	0.002103	0.001928	0.001653	0.001543
m "3 Kg/s	0.1743	0.188133	0.201966	0.226866	0.237933
m _{w2} Kg/s	0.102366	0.1162	0.130033	0.160466	0.177066
m »1 Kg/s	0.276666	0.304333	0.331999	0.387332	0.414999
m a3 Kg/s	0.000427				
<i>m</i> a1 Kg/s					

E Table B.4.2 Results obtained for air and water mass flow rates and inlet and outlet qualities

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Table B.4.3 Results obtained for air and water mass flow rates and inlet and outlet qualities

Mal	Ma2	m a3	M w1	Mw2	M w3	Þ	>	>
Kg/s	Kg/s	Kg/s	Kg/s	Kg/s	Kg/s	ľv	Λ2	
0.000786	0.0004	0.000386	0.276666	0.096833	0.179833	0.002833	0.002833 0.00411383	0.002142
0.000786	0.000467	0.000319	0.304333	0.110666	0.193666	0.002576	0.00420216	0.001644
0.000778	0.000482	0.000296	0.331999	0.1245	0.2075	0.002338	0.00385657	0.001424
0.000778	0.000482	0.000296	0.359666	0.135566	0.224099	0.002158	0.002158 0.00354286	0.001319
0.000778	0.000482	0.000296	0.387332	0.152166	0.235166	0.002005	0.00315759	0.001257
0.000778	0.000482	0.000296	0.414999	0.166	0.248999	0.001871	0.00289522	0.001187

Table B.4.4 Results obtained for air and water mass flow rates and inlet and outlet qualities

	51	91	17	35	75	84
X	0.001551	0.001491	0.001417	0.00135	0.001275	0.001184
X_2	0.00185938	0.00146554	0.00124035	0.00107515	0.00096774	091301
	0.00	00.0	0.00	0.00	_	00.0
X_1	0.001631	0.001483	0.00136	0.001255	0.001166	0.001088 0.00091301
mw3 Kg/s	0.204733	0.213033	0.224099	0.235166	0.248999	0.268366
mw2 Kg/s	0.071933	0.0913	0.1079	0.1245	0.138333	0.146633
m w1 Kg/s	0.276666	0.304333	0.331999	0.359666	0.387332	0.414999
m a3 Kg/s	0.000318					
m a z Kg/s	0.000134					
m _{a1} Kg/s	0.000452					

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Table B.4.5 Results obtained for air and water mass flow rates and inlet and outlet qualities

	•	•	•					
m _{a1}	ma2	M a 3	/// w1	171 22	111 w3	>	>	×
Kg/s	Kg/s	Kg/s	Kg/s	Kg/s	Kg/s	Ī	747	7.7
0.000642	0.000303	0.000339	0.276666	0.083	0.193666	0.002317	0.00363733	0.001749
			0.304333	0.1079	0.196433	0.002106	0.0028003	0.001725
			0.331999	0.1245	0.2075	0.001931	0.00242783	0.001633
			0.359666	0.1328	0.226866	0.001783	0.00227644	0.001494
			0.387332	0.1494	0.237933	0.001656	0.00202401	0.001424
4.78.7 / 3.			0.414999	0.166	0.248999	0.001546	0.001546 0.00182198 0.001361	0.001361

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Table B.5.1 Results obtained for air and water mass fluxes, total mass fluxes and superficial gas and liquid velocities

J _{(I}	0.548193	0.603012	0.657831	0.76747	0.822289
J_{GI}	344.8235 1.075448 0.548193				
G_3 V_{α}/c_{13}	344.8235	372.1235	399,4235	448.5635	470,4035
G ₂	547.2658 202.4423	601.8658 229.7423	656.4658 257.0423	765.6658 317.1023	820.2658 349.8623 470.4035
G_1	547.2658	601.8658	656.4658	765.6658	820.2658
t	343.98	371.28	398.58	447.72	469.56
G_{2w}	202.02	229.32	256.62	316.68	349.44
G _{Iw}	546	9.009	655.2	746.4	819
G _{3a}	0.843485				
G_{2a}	265819 0.422334				
G_{1a}	1.265819				

Table B.5.2 Results obtained for air and water mass fluxes, total mass fluxes and superficial gas and liquid velocities

							,
Ja	in/s	0.548193	0.603012	0.657831	0.712651	0.76747	0.822289
J_{G1}	s/m	547.1565 175.2805 371.876 0.982558 0.548193					
Č [‡] D	Kg/s.m²	371.876	396.446	426.476	453.776	475.616	524.756
G ₂	Kg/s.m²	175.2805	601.7565 205.3105	656.3565 229.8805	257.1805	289.9405	295.4005
G ₁	Kg/s.m ² Kg/s.m ²	547.1565	601.7565	656.3565	453.18 710.9565 257.1805	765.5565 289.9405	524.16 820.1565 295.4005 524.756
G ₃ ,,	Kg/s.m ²	371.28	395.85	425.88	453.18	475.02	524.16
G _{2w}	Kg/s.m²v	174.72	204.75	229.32	256.62	289.38	294.84
G _{I"}	Kg/s.m ²	546	9.009	655.2	709.8	764.4	819
G33	Kg/s.m ²	\sim					
G_{2a}	Kg/s.m ²	0.560481 (
Gla	Kg/s.m ²	1.156486					

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Table B.5.3 Results obtained for air and water mass fluxes, total mass fluxes and superficial gas and liquid velocities

Ğ	G_{2a}	ű	G ₁ *	G _{2w}	G_{3w}	Gı	G ₂	පි	JGI	Ja
Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	Kg/s.m ^{2v}	Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	m/s	s/m
1.551191	0.78941	0.761781	546	191.1		547.5512	547.5512 191.8894		355.6618 1.317901 0.548193	0.548193
1.551191	551191 0.921636 0.629554	0.629554	9.009	218.4	382.2	602.1512	219.3216	602.1512 219.3216 382.8296 1.317901	.317901	0.603012
1.535402	.535402 0.951239 0.584163	0.584163	655.2	245.7	409.5	656.7354	246.6512	656.7354 246.6512 410.0842 1	1.304488 0.657831	0.657831
1.535402	.535402 0.951239 0.584163	0.584163	709.8	267.54	442.26	711.3354	711.3354 268.4912	442.8442	1.304488	0.712651
1.535402	.535402 0.951239 0.584163	0.584163	764.4	300.3	464.1	765.9354	765.9354 301.2512	464.6842 1.304488 0.76747	1.304488	0.76747
1.535402	1.535402 0.951239 0.584163	0.584163	819	327.6	491.4	820.5354	328.5512	820.5354 328.5512 491.9842 1.304488 0.822289	1.304488	0.822289

Table B.5.4 Results obtained for air and water mass fluxes, total mass fluxes and superficial gas and liquid velocities

G	Ğ2,	3,	G _I	Gzw	G _{3w}	ڻ ت	Ğ	පි	J_{G1}	
Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	Kg/s.m ^{2v}	Kg/s.m ²	Kg/s.m²	Kg/s.m ²	Kg/s.m ²	s/m	m/s
0.892033	33 0.264452 0.	0.627581		141.96	404.04	546.892	142.2245	142.2245 404.6676 0.757877 0.548193	0.757877	0.548193
				180.18	420.42	601.492	601.492 180.4445 421.0476	421.0476		0.603012
			655.2	212.94	442.26	656.092	213.2045 442.8876	442.8876		0.657831
			709.8	245.7	464.1	710.692	245.9645 464.7276	464.7276		0.712651
			764.4	273	491.4	765.292	273.2645 492.0276	492.0276		0.76747
			819	289.38	529.62	819.892	819.892 289.6445 530.2476	530.2476		0.822289

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Ja	m/s 0.548193 0.603012 0.657831 0.712651 0.76747 0.822289
Table B 5 5 Results obtained for air and water mass fluxes, total mass fluxes and superficial gas and liquid velocities	Kg/s.m² m/s m/s 382.8698 1.077125 0.548193 388.3298 0.603012 410.1698 0.657831 448.3898 0.712651 470.2298 0.76747 492.0698 0.822289
cial gas and li	G1 G2 G3 Kg/s.m² Kg/s.m² Kg/s.m² 547.2678 164.398 382.8698 601.8678 213.538 388.3298 656.4678 246.298 410.1698 711.0678 262.678 448.3898 765.6678 295.438 470.2298 820.2678 328.198 492.0698
s and superfi	G1 U2 Kg/s.m² Kg/s.m² 547.2678 164.398 601.8678 213.538 656.4678 246.298 711.0678 262.678 765.6678 295.438 820.2678 328.198
al mass fluxe	G _{3w} G ₁ Kg/s.m ² Kg/s.m ² 382.2 547.2678 387.66 601.8678 409.5 656.4678 447.72 711.0678 469.56 765.6678
ss fluxes, tot	G _{2w} Kg/Sm ^{2v} Kg/163.8 38 212.94 38 245.7 40 225.08 44 294.84 46
and water ma	Kg/s.m ² Kg 546 1 546 2 600.6 2 709.8 2 764.4 2
ained for air 8	G _{3a} Kg/s.m ² K
Results obta	G _{1a} G _{2a} G _{3a} G _{3a} Kg/s.m ² Kg/s.m ² 1.267792 0.597978 0.669814
ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡ ተ ፡	G _{1a} Kg/s.m ² 1.267792

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Table B.6.2 Results obtained for pressure differences values at different measurement positions

					Т	٦,	2	8	S	74		-	1	8	45	23	3/6	7	_	59	12	2	7	.57	5.9		
60176	39824	141	Š	170	(ra)	00	.492.448	-646.338	-769.45	1707.894		1,50	-923.34	-954.118	-1046.45	1077 23	7	-1158.79		-1169.59	1,1231 12	+	+	1-1723.57	-2092.9	-	
G,/G1=0.360176	G3/G1=0.639824	X1=0.00141			(Pa)	492.448	984.896	1138.79	1261.9	120034	120021	***	1415.79	1446.57	15389		4	1631.23		1662.01	1772 57	+	-	1 2216.02	258535	-	
レン		01511	VP.	= <i>′</i>	(Pa)	0.0	-492.448	-554.004	-738 672	707 804	-/0/.034		-861.784	292 562	05.4 118	-934.110	-984.896	-1015.67		1-110801	+	-1231.12	-1446.57	1-163123	+	-	
$G_2/G_1=0$	$G_3/G_1=0.621268$	$X_1 = 0.001511$	Q.A.	۳ 3	(Pa)	430.448	984.896	1046.45	1231 12	1000	1200.34		1354 23	1395 01	10.0001	1440.37	1477.34	1508.12	┨	1600 46	4	_	1939.01	2122 68		1 2462.24	
361739	638261	01627	Q.V	71	(Pa)	0.0	-369 336	554 004	215 56	-013.30	-584.782		800 008	22000-	-800.228	-861.784	-923.34	-084 896	272.7	1046.45	-1040.43	-1169.56	-1292 68	╫		1-1908.24	
G,/G,=0.361739	$G_2'/G_1=0.638261$	$X_1 = 0.001627$		3	(Pa)	430 892	900.024	300.220	704.070	1046.45	1015.67		137113	1231.12	1231.12	1292.68	125173	21.5.70	1413.72		-923.34 1477.34	1600.46	- -	[2339.13	
150237	549763	C9210	70/10	Δ <i>P</i>	(P ₃)		0.0	-358.558	-430.892	-523.226	-492.448 1015.67		111	-677.116	-717.743	759 601	800 000	-300.223	-892.50		-923.34	054 118	1046 45	-1040.45	-1108.01	1-1200.34	
750737	G2/G1-0.330237 G-7G0.549763	C3/10/01/2/50	71-0.0	<i>P.</i>	(p ₃)	(n 1)	430.892	769.45	861.784	954.118	923 34			1108.01	1148.64	1100 40	1150.77	1231.12	1323.45		1354 23	130501	10.0001	1477.34	1538.9	1631 23	100 1:20
0011	11185	58815	776	ΛΡ.	= (e	(ra)	0.0	-246.224	-307.78	369 336	307.78	-201.10	İ	-554,004	554 004	100.100	-284.782	-646.338	-738.672		711 22		-800.228	-892.562	984 896	1120 70	-1130.17
	$G_2/G_1=0.341185$	$G_3/G_1 = 0.058815$	$X_1 = 0.001922$	dv	 5 ((Fa)	430.892	677.116	738.672	800 000	900.220	138.017		984 896	200 400	704.070	1015.67	1077.23	1169.56		1000	1108.01	1231.12	1323 45	1415 70	1415.79	1569.68
}		19652	114		77	(Pa)	0.0	-184.668	277 002	033 000	-338.558	-246.224	İ	600 067	430.024	-430.892	-492.448	492 448	-554 004			-677.116	-800,228	921 006	-021.000	-923.34	-1077.23
	$G_2/G_1=0.320348$	$G_3/G_1=0.679652$	$X_1 = 0.002114$		 β β	(Pa)	430 892	┼-	+	+	-	677.116 -		\mid	+	861.784	923 34	-	-	-		1108.00	1231 12	127.1	1261.9	1354.23	1508.12
ľ		_	Distance		E)	_	1 016	╀	+	-}	-0.127	0500	-	-	0.0508	0.127	0.0054	0.020		1.010		0.0508	2010	0.127	0.0254	0.508	1.016
	<u></u>		_	_																							

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Table B.6.4 Results obtained for pressure differences values at different measurement positions

G ₂ /G ₁ =0.357072 G ₂ /G ₁ =0.353271	$G_3/G_1=0.6429278 \mid G_3/G_1=0.6467285$	$X_1=0.001166$ $X_1=0.001088$	$\Delta P_{st} = \Delta P_{tt} = \Delta P_{st}$	(Pa) (Pa) (Pa)	554.004 0.0 554.004	861.784 -307.78 902.411 -348.407	944.269 -390.265 984.896	1025.52 -471.519 1108.01 -554.004	984.896 -430.892 1067.38 -513.377	1138.8 -584.782 1169.56	1169.56 -615.56 1231.12	1200.34 -646.338 1323.45	1261.9 -707.894 1354.23	1292.68 -738.672 1446.57 -892.562	1292.68 -738.672 1354.23	1354.23 -800.228 1415.79	1385.01 -831.006 1538.9	1477.34 -923.34 1569.68 -1015.67	
$G_2/G_1=0.346091$	$G_3/G_1=0.6539085$	A1=0.001233	∆P _{1ι}	(Pa)	0.0	-307.78	-369.336	-451.821	-430.892	-615.56	-656.187	-677.116	-707.894	-800.228	-738.672	-800.228	-861.784	-923.34	
$G_2/G_1=($	$G_3/G_1=0$	ハーレー	ΔP_{st}	(Pa)	492.448	800.228	861.784	944.269	923.34	1108.01	1148.64	1169.56	1200.34	1292.68	1231.12	1292.68	1354.23	1415.79	
324961	5750388	00100	ΔP_{1t}	(Pa)	0.0	-348.41	-369.336	-492.448	-461.67	-584.782	-615.56	-677.116	-707.894	-769.45	-800.228	-800.228	-923.34	-984.896	
$G_2/G_1=0.324961$	$G_3/G_1=0.6750388$	∧ı=0.00130	ΔP_{st}	(Pa)	430.892	779.299	800.228	923.34	892.562	1015.67	1046.45	1108.01	1138.34	1200.34	1231.12	1231.12	1354.23	1415.79	
.299995	0000525	1403	ΔP_{1r}	(Pa)	0.0	-184.666	-307.778	-369.335	-348.406	-430.891	-461.668	-492.446	-554.002	-615.558	-707.892	-738.669	-769.447	-831.003	
$G_2/G_1=0.0.299995$	$G_3/G_1=0.70000525$	A1=0.001405	ΔP_{st}	(Pa)	430.892	615.558	738.67	800.225	779.296	861.78	892.559	923.337	984.893	1046.45	1138.78	1169.56	1200.34	1261.89	
		01031	ΔP_{1r}	(Pa)	0.0	-184.666	-307.778	-348.405	-307.778	-430.888	-461.667	471.517	-492.446	-523.224	-615.558	-677.114	-759.598	-800.225	
$G_2/G_1=0.260059$	G ₃ /G ₁ =0.7399405	A1=0.001031	ΔP_{st}	(Pa)	430.892	615.558	738.67	779.296	738.67	861.78	892.559	902.408	923.337	954.115	1046.45	1108.00	1190.49	1231.12	
	000000	(m)	1		-1.016	-0.508	-0.254	-0.127	-0.0508	0.0508	0.127	0.0254	0.508	1.016	0.0508	0.127	0.0254	0.508	

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Table B.6.5 Results obtained for pressure differences values at different measurement positions

	$G_2/G_1=0$	$G_2/G_1=0.300398$ $G_2/G_1=0.6996023$	$G_2/G_1=0.354792$ $G_3/G_1=0.64520783$	354792 4520783	$G_2/G_1 = 0$	$G_2/G_1=0.375187$ $G_3/G_1=0.6248133$	$G_2/G_1=0.369413$ $G_2/G_1=0.6305866$.369413 6305866	$G_2/G_1=0.385857$ $G_3/G_1=0.6141434$.385857 6141434	$G_2/G_1=0.400111$ $G_3/G_1=0.5998892$	400111
Distance (m)	$X_1 = 0.0$	X ₁ =0.002317	$X_1 = 0.002106$	02106	$X_1 = 0.00193$	001931	X ₁ =0.001783	01783	X ₁ =0.001656	01656	$\frac{3}{X_1=0.001546}$	01546
,	∆P _{st}	$\Delta P_{ m tr}$	ΔP_{x}	ΔP_{1r}	Δν,	ΔP_{11}	W.	ΔP_{1i}	ΔP_{π}	ΔP_{1r}	∆P _{st}	ΔP_{11}
	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)	(Pa)
-1.016	430.892	0.0	430.892	0.0	492.448	0.0	492.448	0.0	554.004	0.0	554.004	0.0
-0.508	677.114	-246.223	738.67	-307.779	831.006	-338.558	861.784	-369.336	954.118	-400.114	984.896	-430.892
-0.254	738.67	-307.779	861.78	-430.891	923.34	-430.892	954.118	-461.67	1046.45	-492.448	1108.01	-554.004
-0.127	800.225	-369.335	954.115	-523.224	1015.57	-523.226	1108.01	-615.56	1169.56	-615.56	1231.12	-677.116
-0.0508	738.67	-307.779	861.78	-430.891	984.896	-492.448	1046.45	-554.004	1108.01	-554.004	1169.56	-615.56
0.0508	892.559	-461.668	1046.45	-615.558	1169.56	-677.116	1231.12	-738.672	1292.68	-738.672	1415.79	-861.784
0.127	923.337	-492.446	1108.00	-677.114	1200.34	-707.894	1261.9	-769.45	1354.23	-800.228	1477.34	-923.34
0.0254	984.893	-554.002	1138.78	-707.892	1261.9	-769.45	1292.68	-800.228	1415.79	-861.784	1477.34	-923.34
0.508	1046.45	-615.558	1200.34	-769,447	1292.68	-800,228	1354.23	-861.784	1446.57	-892.562	1538.9	-984.896
1.016	1108.00	-677.114	1231.12	-800.225	1385.01	-892.562	1415.79	-923.34	1508.12	-954.118	1600.46	-1046.45
0.0508	923.337	-492.446	1108.00	-677.114	1354.23	-861.784	1415.79	-923.34	1538.9	-984.896	1846.68	-1292.68
0.127	1046.45	-615.558	1231.12	-800.225	1415.79	-923.34	1477.34	-984.896	1662.01	-1108.01	1908.24	-1354.23
0.0254	1292.67	-861.781	1446.56	-1015.67	1538.9	-1046.45	1600.46	-1108.01	1846.68	-1292.68	1969.79	-1415.79
0.508	1354.23	-923.337	1508.12	-1077.23	1569.69	-1077.23	1723.57	-1231.12	1969.79	-1415.79	2092.90	-1538.9
1 016	1415 78	-984.893	1569.67	-1138.78	1662.01	-1169.56	1846.68	1-1354.23	2154.46	-1600.46	2431.46	-1877.46

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 ΔP_{11} (bar) 390 390 375 450 440 420 425 425 520 400 520 390 385 395 440 380 430 ΔP_{12} (bar) Measured values 140 120 120 100 105 110 100 110 95 140 155 125 140 140 200 P₃ (bar) 1035 1255 1460 878 850 755 1010 1120 1275 1330 1025 1400 1400 825 860 985 066 P₂ (bar) 1135 1010 009 780 583 925 455 1020 1080 535 720 715 850 935 565 705 765 P_1 (bar) 1020 488 460 099 805 370 435 615 740 835 900 440 565 560 625 880 880 0.001928 0.002114 0.002313 0.002103 0.001653 0.001543 0.001922 0.002576 0.001762 0.001627 0.002833 0.002338 0.002158 0.002005 0.00141 0.00187 0.00151 × 0.585848 0.658815 0.679652 0.630084 0.608445 0.618283 0.573477 0.649763 0.621268 0.639824 0.606689 0.624428 0.622553 0.638261 0.64955 0.63577 0.59959 G_3/G_1 0.369916 0.320348 0.414152 0.341185 0.361739 0.426523 0.378732 0.360176 0.381717 0.350237 0.375572 0.377447 0.393312 0.39155 0.35045 G_2/G_1 0.36423 0.400411 (kg/s.m²) 601.8658 547.2658 656.4658 765.6658 820.2658 547.1565 601.7565 656.3565 710,9565 765.5565 602.1512 820.1565 547.5512 656.7354 711.3354 765,9354 820.5354

Table B.7 Results obtained for pressure difference values at different inlet mass fluxes, inlet qualities and extraction rates

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Table B.7 Results obtained for pressure difference values at different inlet mass fluxes, inlet qualities and extraction rates

ß	G ₂ /G ₁	G_3/G_1	Xı			Measured values	alues	
(kg/s.m ²)				P ₁ (bar)	P ₂ (bar)	P ₃ (bar)	ΔP_{12} (bar)	ΔP_{13} (bar)
546.892	0.260056	0.739941	0.001631	390	460	725	70	335
601.492	0.299995	0.700005	0.001483	395	480	735	85	340
656.092	0.324961	0.675039	0.00136	565	645	068	80	325
710.692	0.346091	0.653909	0.001255	540	630	838	06	298
765.292	0.357072	0.642928	0.001166	550	640	838	06	288
819.892	0.353271	0.6467285	0.001088	620	720	935	100	315
547.2678	0.300398	0.699602	0.002317	440	426	835	98	395
601.8678	0.354792	0.645208	0.002106	290	710	066	120	400
656.4678	0.375187	0.624813	0.001931	610	725	1000	115	390
711.0678	0.369413	0.630587	0.001783	099	780	5501	120	395
765.6678	0.385857	0.614143	0.001656	710	835	1200	125	490
820.2678	0.400111	0.599889	0.001546	780	895	1230	115	450

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Table B.8 Results obtained for homogeneous densities, velocity ratios and void fractions

0	ρ,,,	ρ	S.	S,	S,	5	2	Į
(kg/m³)	(kg/m^3)	(kg/m³)	•	3	•	3 	2	ર્દે
337.0609	360.4426	324.6952	1.705633	2.280345	1.925303	0.534926	0.436867	0.518713
358.5822	390.0159	341.5854	1.652439	2.142255	1.866131	0.519067	0.421122	0.507432
378.7421	416.9594	357.6466	1.608103	2.033537	1.81506	0.504122	0.406473	0.496632
415.4639	468.5527	384.654	1.53842	1.860322	1.738817	0.476676	0.37758	0.478309
432,2337	492.999	395.9373	1.510543	1.790932	1.71005	0.464045	0.363487	0.470592
357.4433	268.996	422.9996	1.685413	2.541315	1.798787	0.515376	0.516479	0.430255
379.5207	301.1475	438.6391	1.63405	2.332985	1.75667	0.49929	0.498217	0.420384
400.1225	325.4119	456.6322	1.591241	2.203095	1.711792	0.48418	0.484211	0.408918
419.3918	350.459	472.0101	1.555014	2.087929	1.676153	0.469957	0.469547	0.399024
437.454	378.1518	483.6952	1.523959	1.978401	1.65059	0.456547	0.453083	0.391448
454.4189	382.5379	508.1721	1.497043	1.962511	1.600861	0.443881	0.45045	0.375409
		,						
293.421	222.4662	354.4075	1.758372	2.540218	1.877298	0.577564	0.579141	0.491751
313.4713	218.8172	416.7494	1.700405	2.423806	1.788275	0.562421	0.595679	0.438027
334.6731	233.8224	451.9065	1.649657	2.291243	1.732531	0.545883	0.588452	0.410638
352.6458	249.3434	470.9379	1.608953	2.195744	1.687172	0.532203	0.578102	0.39849
369.6675	271.4746	482.9027	1.574059	2.078514	1.66049	0.519193	0.563245	0.390783
385.8117	288.9393	497.1132	1.543814	1.998715	1.630471	0.506804	0.551437	0.381559

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Table B.8 Results obtained for homogeneous densities, velocity ratios and void fractions

72215 0.428689 47469 0.422432
0.372215
0.457938 0 0.441553 0
1.727071 0.4
111111111111111111111111111111111111111
7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7

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Table B.9 Results obtained for momentum densities, energy weighted densities, and two-phase flow loss multiplier (ϕ)

								-		_			_	- 1	-			_	1	Т		_	1	\neg	
$(\phi_{Rsm})_3$	2 846547	2 645948	7 403703	2.403203	2.21954	2.110807		3 297501	0,00000	3.033103	2.840795	2 672826	2.07/2020	2.517482	2.45108		4.099963	4 224 146	2010	4.018522	3.771773	3 519675	222710	3.320310	
$(\phi_{Rsm})_2$	2 159939	201007	3.021024	2.895115	2.703655	2 628259		2 006063	2.020202	2.082417	2.024449	1 004575	1.704020	1.968161	1.845105		2.5736	2 2 1 7 0 1 8	2.2112.2	2.079237	1 997007	1 079536	1.37,0030	1.933371	
$^{nylS}(\phi)$	7 1/8/721	17/041.7	50///0.7	2.015098	1 909403	1 864417	1.00.1	200000	0/07007	1.995756	1 937759		7/7089.1	1,838761	1 796901	7,7,7,1,1	235575	200000	7700777	2.200265	2 13587	720000	0508/07	2.025858	
ρ _{E3}	(Mg Oct	4/8.934/	490.3351	501.2032	519 5961	10/2:010	7075.175	0,0	567.3612	577.2621	8677 003	300.7470	598.6479	606 2239	7777 777	022.2477	505 0271	303.5371	559.6364	587 0057	500 1650	259.1027	606.8737	616.0931	
PE1 3	(Kg/III)	559.6064	575.7282	590,6065	507013	019.7423	633.8889		478.97	498 1323	270.17	217.2780	527.5574	12C 112	244.604	240.9142	0000000	410.2038	400.3233	108 1574	400.1014	418.8/8/	434.1166	446 1361	140.1001
PB.	(kg/m²)	463.2663	479.1479	404 0035	27.07.02	521.5040	534.1081		482.7701	109 9671	470.0071	513.9707	528 1739	541 5572	341.3373	554.1915		420.6968	435 8937	7037 637	452.4507	466.1349	479.1366	401 5082	491.3002
PM3	(kg/m²)	479.4525	490 7492	501 5577	301.3327	519.8614	527.5655		59 295	2005	0/1/2090	588.9555	508 8771	270.027	606.3842	622.3749		506.3514	550 971	177://	587.2292	599.3519	9050 209	71,000	616.2381
PM1,	(kg/ m²)	560.5554	576 4141	1001	291.122	620.0468	634.1275		2677 001	400.702	499.3269	513.4748	500 027	770.070	544.7544	547.3923		418.2565	700 101	401.3270	409.4496	419.9257	434 0052	434,3032	446.7742
PMI	(kg/ m³)	463 5315	470.2613	4/9.5015	494.2688	521.6291	534.2151		3000 001	483.0083	499.059	514 1283	200000	528.3057	541.6692	554.2877		421 0369	120.121	436.1074	452.6727	3065 330	20.000	4/9.2942	491,6436

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Table B.9 Results obtained for momentum densities, energy weighted densities, and two-phase flow loss multiplier (ϕ)

$(\phi_{{\scriptscriptstyle R}{\scriptscriptstyle S}{\scriptscriptstyle m}})_3$	2.448425	2.247161	2.101711	1.983381	1.896457	1.842059		3.530036	3.145195	2.91104	2.777716	2.612392	2.473458
$(\phi_{Rsm})_2$	2.200129	2.268338	2.254906	2.225133	2.1678	2.079802		2.147522	2.296042	2.270267	2.14933	2.123992	2.094197
$^{nys}(\phi)$	1.843709	1.789588	1.742007	1.699844	1.662222	1.628443		2.150266	2.079228	2.016488	1.960663	1.910666	1.865625
$ ho_{E3}$ (kg/m ³)	568.9975	575.2669	583.3405	591.1058	600.4061	612.7246		545.8283	548.0296	556.6185	570.8673	578.5957	586.0446
$ ho_{\mathcal{E}_{1}}$ (kg/m ³)	622.9336	648.767	667.9159	684.9449	697.7945	704.9942		457.3921	488.7549	507.3047	516.0193	532.4758	547.7802
$ \rho_{\rm a} $ (kg/m ³)	540.0375	556.4097	571.6369	585.8379	599.1146	611.5555		462 9327	478 8101	493.7524	507.8444	521.1595	533.762
ρ_{M3} (kg/m ³)	569.2458	575.4927	583,5407	591.2843	600.5622	612.8556		546 1315	548 3227	556 8755	571 0752	578.7815	586.2116
ρ_{M2} (kg/m ³)	624.5123	649.6059	668.4573	685.3186	9620.869	705 2399		450 6406	780 0057	508 1068	516 7018	532 9851	548.1732
ρ_{M1} (kg/m ³)	540 2153	556.5526	571 7543	585 936	599 1979	611 6771	1,70,110	163 1084	470 0730	103 0781	507 0914	521 2843	533.8693

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Table B.10 Results obtained for single-phase flow loss coefficient, Separated flow model and Reimann and Seeger model pressure difference values

$(\Delta P_{13})_{Rsm}$	(Pa)	242.8258	265.9848	290.2363	334.0798	353.462	250.9461	267.5594	293.5189	316.6579	332.7418	392.1107	273.1428	286.188	304.1111	336.9806	352.0549	377.7437
$(\Delta P_{12})_{Rsm}$	(Pa)	72.31312	86.47338	101.6965	139.4678	162.656	66.16411	81.22607	94.50628	110.2528	130.7129	133.6063	94.53968	122.7673	145.091	161.5931	188.2333	211.3177
$(\Delta P_{13})_{Slm}$	(Pa)	225.7379	256.4242	288.3739	353.0966	385.7355	183.8038	205.2876	230.7787	255.8637	278.806	319.8243	198.0584	175.1701	175.2486	195.7392	212.4056	232.2905
$(\Delta P_{12})_{Sim}$	(Pa)	105.6608	114.1369	123.0527	140.987	150.7645	164.0643	180.0554	199.0846	217.6174	235.0905	262.7069	202.0796	252.9373	288.6434	321.6566	352.6433	386.134
K ₁₃		1.139553	1.131534	1.125079	1.111043	1.103826	1.176519	1.160333	1.153593	1.145285	1.133534	1.146398	1.153437	1.143523	1.135672	1.1344	1.123948	1.119446
K ₁ ,	1	1.032545	1.034234	1.035872	1.040425	1.043384	1.028738	1.029692	1 030399	1 031552	1.033778	1.031378	1.030417	1.031839	1 033317	1 033588	1 036186	1.037524

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Table B.10 Results obtained for single-phase flow loss coefficient, Separated flow model and Reimann and Seeger model pressure difference values

$(\Delta P_{13})_{Rsm}$	(Pa)	274.0137	279.9808	294.9068	0100010	310.951/	335.4192	376.4396		1100000	784./811	272.0142	786 6367	000000000000000000000000000000000000000	327.0013	343.6617	2612504	501.5504
$(\Delta P_{12})_{Ram}$	(Pa)	36.43924	48.42115	60,80902	200000	74.9255	87.93858	96.14501			64.54689	89,93911	100 2814	102.201	119.0054	140 5906	000000	103.8402
$(\Delta P_{i,})_{cc}$	(Pa)	1901 800	248 361	272 6164	4010.677	300.025	331.2593	371 6966			203.7295	216 5136	220 2160	239.3100	271.5725	296 2594	1,07.0/7	321.9073
(AP.).	(Pa)	06 07176	01 61758	71.012.0	91.8093	93.63858	98.92481	108 6181	1010:011		177.5487	184 2830	(CO7: FO)	200./850	224 0727	241.054	77.1.7	261.1067
N.:-	NI3	1 770627	1.226034	1.195255	1.172855	1.156658	1 148622	1 151374	1.1212/7		1 192896	1 150360	1.130203	1.135934	1 139901	1100001	1.128/92	1.119635
	N12	1 021050	1.031239	1.028/11	1.028868	1 030053	1 031049	1.000.1	1.0500/5		1 028703	1.020.1	1.050821	1.033262	1 023/18	1.05248	I.034898	1.037466

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Table B.11 Results of measured pressure difference values at constant inlet mass flux and different inlet analities

			9 0 4	<u>r</u>	bar		1 1 0	000	275	430	215		450
3	$G_1 = 820$	(kg/s.m ²)	_		bar	_				110	1_	_ •	
\ 	5	(kg			× 10°	_		101	1.8/1	14.	000	1.000	1.546
-		-	4	\ <u>\</u> 2	bar		054 	3	470	100 400	000	7.00	490
	$G_1 = 765$	$(kg/s.m^2)$	4	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	bar	3	07.1		140	100	3	3	125
	ٿ ^ι	(kg/		- K			0.001653	0000	0.002005	0 001511		0.001100	0.001656
	-		1		bar		400		380	298		595	
	$G_1 = 711$	$(kg/s.m^2)$		\	bar		140		100	06		120	
	Ġ.	(kg		×		_	0.002158	-+	0.001627		-1-	0.001783	
		-		\	bar		375		425	305		390	325
	$G_1 = 656$	$k_{\varphi}/s.m^2$	1	₽	bar		120	1	155	105	3	115	80
	G	(kg		×	-		0.001928		390 0.002338 155	0.001762	340 0.001102	400 0.001931	0.00136
2711				\ P_1	72	ı	425		390	2.40	2	400	
יר הוממו	G1=602	ko/s m ²)	7	D. D.	har		140		100		ı	120	
alla different mice quanticas	ئ	(kg)	17	×			0.002576		420 0 00192	0.001		0.002106	
מות חווש				∆P.		100	390		420		282	335	١
	G.=547	(1-a/c m ²)	3.111	\ VP.	1 4	- 5	95				85	70	2
	<u>ئ</u> ا	<u> </u>	H V	Ϋ́			0.002313		0.000833 105	0.00200	0.002114	0.001631	1.000.0

Table B.12 Results of measured pressure difference values at constant inlet qualities and different inlet mass fluxes and extraction rates

	מום סוום	ally ullicities that there are contaction and	111000	77777	חות כעונים	2-121								
1000 V	101				$X_1 = 0.0021$	21				$X_1 = 0.0018$	18	•		
ろうし て	+70										Ć		ر ان	- 0
ن	G-/G1	G ₁ /G ₁	V V	ΔP.	<u>.</u>	5/3	<u>G</u> ,/G	۵,	4	- 5		3	֖֖֖֝֞֝֞֝֞֝֝֞֝֝֓֞֝֝֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֓֡֝֓֡֓֡֓֡֓֡֓֡֓	, T
$K_0/\kappa m^2$	Kg/s.m ² Kg/s.m ²	Kg/s.m ²	har	har har	Kg/s.m ²	Kg/s.m ²	Kg/s.m ²	bar	bar bar	Kg/s.m²	Kg/s.m ²	bar	bar Kg/s.m	bar
,)))		Oar							2000		2110070	17.0
601	0.36423 0.63577	0.63577	140	425	601	0.381719	0.381719 0.618281 140		390	959	0.391555 120		0.608445	5/5
									Ì				2100270	000
737	0 375563	7575562 0 624437 155		202	710	0 377438	0377438 0.622562 140 400	140	400	109	0.341185	3	0.341185 100 0.658815 390	060
000	0.077.00	0.024420.0			2					1	FCC074 C	100	205 570750	205
177	020020	A8 70A0A0 805005 0	86	395	547	0.320348	0.320348 0.679652 85		385	650	0.350237	102	0.04%/0.0	22
247	0.5005.0	700000	3		· ·					656	0.375187 115	115	0.624813	390
									3		00000	130	620620	205
					601	0.354792	0.354792 0.645208 120 400	120	400	710	0.309413 120	120	П	37.3
					•					820	0.400411 200	200		
	_													

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	X _{13,i=} In		$X_{13,i} = A_{13,i} = A_{13,i}$	$(X_{1n,12,})^{\frac{2}{5m}}$	$(X_{\ln,13,i})^{\frac{2}{35m}}$	$(X_{ln,12,i})_{Rsm}^{\dagger}$	$(X_{\mathrm{ln},13,i})^{\tilde{i}}_{\mathrm{ism}}$
) Sim	$(\frac{\Delta P_{13.i.meas.}}{\Delta P_{13.i.pred}})_{Slm}$	$\operatorname{Ln}(\overline{\Delta P_{12,t,pred}})_{Rsm}$	$(\Delta P_{13.i,pred})_{Rsm}$				
-	0.546772	0.272871	0.473803	0.011312	0.29896	0.074459	0.224489
-	0.419314	0.481806	0.382707	0.041715	0.175824	0.232137	0.146465
	0.262668	0.165499	0.256231	0.000631	0.068995	0.02739	0.065654
-0.16118	0.242506	-0.15034	0.297868	0.025978	0.058809	0.022603	0.088725
-0.27079	0.131623	-0.34671	0.218999	0.073326	0.017325	0.120204	0.04796
-0.48035	0.751693	0.279294	0.43026	0.230734	0.565042	0.078005	0.185124
ļ. <u></u> .	0.886332	0.131352	0.39544	0.349871	0.785584	0.017253	0.156373
-0.62177	0.885884	0.066064	0.334696	0.386594	0.78479	0.004364	0.112021
-0.83184	0.714682	-0.14344	0.171439	0.691962	0.51077	0.020575	0.029391
-0.92381	0.895331	-0.29604	0.390042	0.853433	0.801618	0.087639	0.152132
-0.65787	0.80584	-0.05505	0.319613	0.432789	0.649378	0.00303	0.102152
-0.65761	0.739375	0.250513	0.428005	0.432447	0.546675	0.062757	0.183188
-0.58809	0.641735	0.207934	0.376805	0.345855	0.411824	0.043237	0.141982
-0.63977	0.537427	0.105294	0.296944	0.409305	0.288827	0.011087	0.088176
-0.68226	0.395526	-0.0023	0.182349	0.465476	0.156441	5.27E-06	0.033251
-0.8548	0.456259	-0.26783	0.279408	0.730684	0.208172	0.071735	0.078069
-0.87056	0.296013	-0.19442	0.092241	0.757872	0.087624	0.037798	0.008508
-0.32592	0.38432	0.652849	0.200952	0.106223	0.147702	0.426212	0.040382
-0.07492	0.314062	0.562714	0.194225	0.005613	0.098635	0.316648	0.037723
-0.13769	0.172098	0.274289	0.097166	0.018958	0.029618	0.075234	0.009441
-0.03963	-0.00677	0.183315	-0.04254	0.001571	4.59E-05	0.033605	0.00181

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Table B.13 Results of the standard deviation of logarithmic ratios between measured and predicted pressure difference values

36.76%	32.65 %	75.95 %	79.55 %	$\sum_{i=1} X_{\text{ln},i}^2 =$ $S_{\text{ln},i} =$		
7 548227	2 075352	010000	72700 0			
0.048136	0.125313	0.112214	0.672395	0.2194	-0.354	
0.125846	0.013815	0.25318	0.436173	0.354748	-0.11754	1
0.035691	6.93E-05	0.140368	0.389974	0.188922	0.008323	
0.094822	0.002602	0.238494	0.31059	0.307931	0.051006	
0.148695	0.083151	0.376764	0.184029	0.38561	0.288359	
0.107037	0.082343	0.438367	0.525477	0.327165	0.286955	i
0.03175	0.001545	0.027392	0.006834	-0.17818	0.039313	1
0.023232	0.000537	0.019583	0.00894	-0.15242	0.023171	

ملخص " فقدان الضغط الناتج عن جريان ثناني الطور في وصلة ثنانية التفرع (T)"

إعداد علاء الدين توفيق سليم الحلحولي

> المشرف أ.د. محمود حماد

المشرف المشارك أ.د. محمود الحسين

خلال هذه الدر اسة تم تصميم وبناء جهاز لإجراء تجارب في مجال جريان ثنائي الطور خلال وصلة ثنائية التفرغ (T). بني هذا الجهاز باستخدام أنابيب مصنوعة من مادة (PVC) بأقطار داخلية متساوية (70,10) موصلة باتجاه أفقي .

يتضمن جهاز التجربة: مضخة ماء طاردة عن المركز ، ضاغطة هواء، أنابيب ، فاصل للماء عن الهواء ،خلاط ، صمامات ، وصلة ثنائية التفرغ (٦) وساعة ضغط.

تم إجراء التجارب على جريان الماء و جريان ثنائي الطور في مختبرات الهندسة الميكانيكية في جامعة مؤتة، من خلال تغيير معدل تدفق الماء و الهواء خلال المدخل و استخدام ضغط نظام ثابت ابار . وتم قياس معدل جريان الماء و الهواء ، بالإضافة إلى فروق الضغط على طول المدخل و النفر عات باستخدام أدوات القياس: ساعة ضغط، أدوات لقياس معدل التدفق (Rotometers, Pitot tube, Orifice).

تم الحصول على معامل فقدان الطاقة (K) في جريان الماء(single-phase)بالإضافة إلى فقدان الضغط الناتج عن جريان الماء و جريان ثنائي الطور (ماء+هواء).

اظهرت النتائج أن فقدان الضغط الناتج عن جريان ثنائي الطور يعتمد على تدفق ثنائي الطور من المدخل بالإضافة إلى الكمية الموزعة على التفرعات. و من الواضح أن فرق الضغط يزداد بزيادة أي من هده العوامل كما أن مقارنة النتائج هذه الرسالة مع در اسات سابقة أظهرت التوافق.